

GOVERNING CODES & STANDARDS		
BUILDING CODE:	OBC 2024 IBC 2021	OHIO BUILDING CODE INTERNATIONAL BUILDING CODE
STANDARDS:	ASCE 7-16 ADM ACI 318 TMS 402 ASC 360 AISC 341 AWS D1.1 AWS D1.3 AWS D1.4 ANSI 8100 AWC NDS AWC SPDWS ASTM	AMERICAN SOCIETY OF CIVIL ENGINEERS: MINIMUM DESIGN LOADS FOR BUILDINGS AND OTHER STRUCTURES THE ALUMINUM ASSOCIATION: SPECIFICATION FOR ALUMINUM STRUCTURES AMERICAN CONCRETE INSTITUTE: BUILDING CODE REQUIREMENTS FOR STRUCTURAL CONCRETE THE MASONRY SOCIETY: BUILDING CODE REQUIREMENTS FOR MASONRY STRUCTURES AMERICAN INSTITUTE OF STEEL CONSTRUCTION: SPECIFICATION FOR STRUCTURAL STEEL BUILDINGS AMERICAN INSTITUTE OF STEEL CONSTRUCTION: SEISMIC PROVISIONS FOR STRUCTURAL STEEL BUILDINGS AMERICAN WELDING SOCIETY: STRUCTURAL WELDING CODE - STEEL AMERICAN WELDING SOCIETY: STRUCTURAL WELDING CODE - SHEET STEEL AMERICAN WELDING SOCIETY: STRUCTURAL WELDING CODE - REINFORCING STEEL AMERICAN IRON AND STEEL INSTITUTE: NORTH AMERICAN SPECIFICATION FOR THE DESIGN OF COLD-FORMED STEEL STRUCTURAL MEMBERS AMERICAN WOOD COUNCIL: NATIONAL DESIGN SPECIFICATION FOR WOOD CONSTRUCTION AMERICAN WOOD COUNCIL: SPECIAL DESIGN PROVISIONS FOR WIND AND SEISMIC AMERICAN SOCIETY FOR TESTING AND MATERIALS

DESIGN CRITERIA		
1. DEAD LOADING		
A. SUPERIMPOSED DEAD LOAD	5 PSF (IN ADDITION TO THE STRUCTURE SELF WEIGHT)	
2. LIVE LOADING		
A. ROOF LIVE LOAD	20 PSF (REDUCIBLE)	
3. RAIN LOADING		
A. DESIGN RAINFALL: 4.5"/HOUR (100-YEAR, 1-HOUR RAINFALL)		
a. RAINWATER AT LOWEST POINT OF ROOF SHALL NOT POND DURING DESIGN RAINFALL		
B. DESIGN RAIN LOAD, R:	20 PSF	
4. SNOW LOADING		
A. GROUND SNOW LOAD	20 PSF	
B. MIN ROOF LOAD	25 PSF	
C. SNOW EXPOSURE FACTOR	1.0	
D. SNOW LOAD IMPORTANCE FACTOR	1.0	
E. THERMAL FACTOR	1.2	
5. WIND LOADING INPUTS		
A. RISK CATEGORY	II	
B. ULTIMATE WIND SPEED	110 MPH (ASD=SQRT(0.6)*Vult)	
C. WIND EXPOSURE FACTOR	C	
D. DIRECTIONALITY/OTHER FACTORS	Kd=0.85, G=0.85, Kz=0.85, Kzt=1.0	
E. METHODOLOGY	OPEN HOST ATTACHED CANOPY	
F. MEAN ROOF HEIGHT	10' - 0"	
G. SYSTEM MOUNTING HEIGHT	0' - 0"	
6. SEISMIC LOADS		
A. RISK CATEGORY	II	
B. SITE CLASS	D	
C. S _s	0.136	
D. S ₀₅	0.145	
E. S ₁	0.049	
F. S _{D1}	0.078	
G. SEISMIC DESIGN CATEGORY (SDC)	B	
H. LONG TRANSITION PERIOD (T _L)	12	
I. LATERAL RESISTING SYSTEM	NONSTRUCTURAL COMPONENTS (APPENDAGES & ORNAMENTATIONS)	
J. REDUNDANCY FACTOR, ρ _r	1.0	
K. OVERSTRENGTH FACTOR, Ω _o	2.00	
L. RESPONSE MODIFICATION FACTOR, R:	2.50	
M. AMPLIFICATION FACTOR:	2.50	
7. SERVICEABILITY:		
a. TOTAL LOAD DEFLECTION:	L/120	
b. LIVE LOAD DEFLECTION:	L/180	

EXISTING CONDITIONS & DEMOLITION		
1. EXISTING CONDITIONS		
A. THE CONTRACTOR SHALL VERIFY ALL DIMENSIONS AND CONDITIONS OF EXISTING BUILDINGS AT THE CONSTRUCTION SITE AND REPORT ANY DISCREPANCIES FROM THE ASSUMED CONDITIONS SHOWN ON THE DRAWINGS TO THE ARCHITECT/STRUCTURAL ENGINEER OF RECORD PRIOR TO FABRICATION OR ERECTION OF ANY STRUCTURAL MEMBER.		
B. EXISTING CONDITIONS SHALL BE SHOWN HALFTONE ON THE CONTRACT DRAWINGS UNLESS NOTED OTHERWISE.		
C. EXISTING CONDITIONS SHOWN ON THE CONTRACT DRAWINGS WERE OBTAINED FROM EXISTING CONSTRUCTION DOCUMENTS AND LIMITED SITE OBSERVATION. DRAWINGS OF EXISTING CONSTRUCTION ARE AVAILABLE TO THE CONTRACTOR UPON REQUEST TO THE OWNER. HOWEVER, THE AVAILABLE DRAWINGS OF THE EXISTING CONDITIONS ARE NOT NECESSARILY COMPLETE. THE CONTRACTOR SHALL FIELD VERIFY ALL APPLICABLE INFORMATION.		
D. THE CONTRACTOR SHALL VERIFY THE LOCATION OF EXISTING UTILITIES PRIOR TO START OF CONSTRUCTION AND ENSURE PROTECTION OF EXISTING UTILITIES THAT REMAIN IN SERVICE.		

CONCRETE		
1. CONCRETE MIXTURES SHALL BE DESIGNED TO REACH A COMPRESSIVE STRENGTH OF 3,000 PSI IN 28 DAYS.		
2. NORMAL WEIGHT CONCRETE SHALL BE USED FOR ALL CONCRETE MEMBERS UNLESS NOTED OTHERWISE. NORMAL WEIGHT CONCRETE SHALL HAVE A CURED DENSITY OF 145 PCF ±5 PCF, WHERE LIGHTWEIGHT CONCRETE IS SPECIFIED THE CURED DENSITY SHALL BE 112 PCF ±3 PCF.		
3. ALL REINFORCEMENT SHALL BE DEFORMED BARS OF INTERMEDIATE GRADE NEW BILLET STEEL CONFORMING TO CURRENT REQUIREMENTS OF ASTM A615, GRADE 60 (U.O.N.).		
4. CLEAR COVER FOR REINFORCEMENT SHALL BE:		
A. FOOTINGS PERMANENTLY EXPOSED TO EARTH: 3"		
B. UNFORMED FACES EXPOSED TO EARTH: 3"		
C. FORMED FACES IN CONTACT WITH EARTH: 2"		
5. THE USE OF RECYCLED CONCRETE IS PROHIBITED WITHOUT WRITTEN APPROVAL FROM THE STRUCTURAL ENGINEER OF RECORD.		
6. EACH MIX SHALL BE UNIQUELY IDENTIFIED BY MIX NUMBER AND THE INTENDED LOCATION OF PLACEMENT ON THE SPECIFIC PROJECT SHALL BE CLEARLY STATED.		
7. WHERE WELDS ARE INDICATED FOR REINFORCING STEEL ON THE DRAWINGS, REINFORCING STEEL SHALL BE A706, GRADE 60 UNLESS OTHERWISE NOTED.		
8. WELDED WIRE REINFORCEMENT SHALL CONFORM TO THE MATERIAL REQUIREMENTS OF ASTM A1064.		
9. ALL 90°, 135°, AND 180° HOOKED REINFORCEMENT SPECIFIED AND GRAPHICALLY DEPICTED IN THE CONTRACT DOCUMENTS SHALL BE DETAILED IN ACCORDANCE WITH ACI 318 STANDARD HOOK GEOMETRY FOR DEFORMED BARS IN TENSION AND FOR STIRRUPS, TIES, AND HOOPS.		
10. FOR EVERY VERTICAL OR HORIZONTAL BAR DISCONTINUED BY AN OPENING, ONE BAR (MINIMUM OF 2 BARS) SHALL BE ADDED AT SIDE OF OPENING (HALF TO EACH SIDE, TYPICAL).		
11. ALL LAP SPLICES SHALL BE CLASS B TENSION LAP SPLICES IN ACCORDANCE WITH ACI 318 UNLESS NOTED OTHERWISE. SEE LAP SPLICE SCHEDULE ON SHEET S002 FOR LAP SPLICE LENGTHS. UNLESS NOTED AS CONTINUOUS, REINFORCEMENT SHALL ONLY BE SPLICED AT LOCATIONS SHOWN ON THE CONTRACT DOCUMENTS. SPLICES AT NON-SPECIFIED LOCATIONS SHALL BE SUBMITTED BY THE CONTRACTOR TO THE STRUCTURAL ENGINEER OF RECORD FOR APPROVAL.		
12. A MINIMUM LAP SPLICE OF 8" SHALL BE PROVIDED AT ALL END AND SIDE LAP CONDITIONS FOR WELDED WIRE REINFORCEMENT UNLESS NOTED OTHERWISE.		

STRUCTURAL ALUMINUM & ALUMINUM WELDING:		
1. ALL COMPONENTS SHALL BE STRUCTURAL ALUMINUM (U.N.O.) AND SHALL BE FABRICATED AND ERECTED ACCORDING TO THE GOVERNING BUILDING CODE AND MATERIAL STANDARDS REFERENCED ON THIS SHEET.		
2. ALL STRUCTURAL ALUMINUM SHALL BE MIN 1/8" THICK U.N.O. AND BE OF THE FOLLOWING ALLOY AND TEMPER:		
A. BEAMS, PURLINS, COLUMNS	6063-T6	
B. ALL OTHER EXTRUSIONS	6063-T6	
C. FASTENERS	SS 316	
3. STRUCTURAL ALUMINUM SHALL BE FRAMED PLUMB AND TRUE AND ADEQUATELY BRACED DURING CONSTRUCTION.		
4. WHERE ALUMINUM IS IN CONTACT WITH OTHER METALS EXCEPT 300 SERIES STAINLESS STEEL, ZINC OR CADMIUM AND THE FAYING SURFACES ARE EXPOSED TO MOISTURE, THE OTHER METALS SHALL BE PAINTED OR COATED WITH ZINC, CADMIUM, OR ALUMINUM.		
5. UNCOATED ALUMINUM SHALL NOT BE EXPOSED TO MOISTURE OR RUNOFF THAT HAS COME IN CONTACT WITH OTHER UNCOATED METALS EXCEPT 300 SERIES STAINLESS STEEL, ZINC, OR CADMIUM. ALUMINUM SURFACES TO BE PLACED IN CONTACT WITH MASONRY, CONCRETE, WOOD, FIBERBOARD, OR OTHER POROUS MATERIAL THAT ABSORBS WATER SHALL BE PAINTED.		
6. FOR ALUMINUM IN CONTACT WITH CONCRETE: ACCEPTABLE PAINTS: PRIMING PAINT (ONE COAT), SUCH AS ZINC MOLYBDATE PRIMER IN ACCORDANCE WITH FEDERAL SPECIFICATION TT-P-845B ("GOOD QUALITY", NO LEAD CONTENT), ALT: HEAVY COATING OF ALKALI-RESISTANT BITUMINOUS PAINT. ALT: WRAP ALUMINUM WITH A SUITABLE PLASTIC TAPE APPLIED IN SUCH A MANNER AS TO PROVIDE ADEQUATE PROTECTION AT THE OVERLAPS.		
7. ALUMINUM SHALL NOT BE EMBEDDED IN CONCRETE TO WHICH CORROSIVE COMPONENTS SUCH AS CHLORIDES HAVE BEEN ADDED IF THE ALUMINUM WILL BE ELECTRICALLY CONNECTED TO STEEL. EMBEDDED ALUMINUM ELEMENTS WILL BE COVERED WITH PLASTIC TAPE OR OTHERWISE PROTECTED AS PER 2020 ADM M.7.3.		
8. BOLT HOLES SHALL BE DRILLED IN THE SAME NOMINAL DIAMETER AS THE BOLT + 1/16".		
9. ALUMINUM WELDING SHALL BE PERFORMED IN ACCORDANCE WITH WELD FILLER ALLOYS MEETING ANSI/AWS A5.10 STANDARDS TO ACHIEVE ULTIMATE DESIGN STRENGTH IN ACCORDANCE WITH THE ALUMINUM DESIGN MANUAL PART I-A, TABLE 7.3.1. ALL ALUMINUM CONSTRUCTION SHALL BE IN CONFORMANCE WITH THE TOLERANCES, QUALITY, AND METHODS OF CONSTRUCTION AS SET FORTH IN THE AMERICAN WELDING SOCIETY'S STRUCTURAL WELDING CODE ALUMINUM (D1.2). MINIMUM WELD IS 1/8" THROAT FULL PERIMETER FILLET WELD UNLESS OTHERWISE NOTED.		
10. STAINLESS STEEL FASTENERS SHALL BE ASTM F593 316 SS COLD WORKED CONDITION, PROVIDE (5) PITCHES MINIMUM PAST THE THREAD PLANE FOR ALL SCREW CONNECTIONS. ALL FASTENER CONNECTIONS TO METAL SHALL PROVIDE 2xDIAMETER EDGE DISTANCE AND 3xDIAMETER SPACING.		
11. SELF-DRILLING SCREWS SHALL BE TEK BRAND / ALL POINTS FASTENERS OF SIZE #14, STAINLESS STEEL 300 SERIES, WITH MINIMUM 1/2" THREAD ENGAGEMENT BEYOND THE CONNECTED PART, UNLESS OTHERWISE NOTED.		
12. THE CONTRACTOR IS RESPONSIBLE TO INSULATE ALL MEMBERS FROM DISSIMILAR MATERIALS TO PREVENT ELECTROLYSIS.		

STRUCTURAL WOOD		
1. ALL DIMENSION LUMBER SHALL BE STRUCTURAL GRADE #2 SOUTHERN YELLOW PINE OR BETTER MEETING APPLICABLE REQUIREMENTS OF THE SOUTHERN PINE INSPECTION BUREAU (SPIB) AND PRESSURE-IMPREGNATED (PT) BY AN APPROVED PROCESS (ACQ 0.4 PRESSURE TREATED) PRESERVATIVE IN ACCORDANCE WITH THE APPLICABLE PROVISIONS OF THE BUILDING CODE AND AMERICAN WOOD PRESERVERS ASSN (AWPA) "BOOK OF STANDARDS" OR 0.55 SPECIFIC GRAVITY MIN.		
2. MEMBER SIZES SHOWN ARE NOMINAL UNLESS NOTED OTHERWISE.		
3. ALL METAL CONNECTORS IN CONTACT WITH WOOD USED IN LOCATIONS EXPOSED TO WEATHER SHALL BE GALVANIZED.		
4. NAILS SHALL PENETRATE THE SECOND MEMBER A DISTANCE EQUAL TO THE THICKNESS OF THE MEMBER BEING NAILED THERETO. THERE SHALL BE NOT LESS THAN 2 NAILS IN ANY CONNECTION.		
5. MEMBERS SHALL BE FREE OF CRACKS AND KNOTS. MOISTURE CONTENT SHALL BE 19% OR LESS. DRY WOOD MAY SPLIT MORE EASILY, IF WOOD TENDS TO SPLIT, PRE-BORING HOLES SHALL BE USED WITH DIAMETERS NOT EXCEEDING 3/4" OF THE NAIL DIAMETER OR USE A 5/32" BIT FOR SDS SCREWS. A FASTENER THAT SPLITS THE WOOD SHALL BE REEVALUATED PRIOR LOADING THE CONNECTION.		
6. WOOD THAT IS IN CONTACT WITH CONCRETE OR MASONRY, AND AT OTHER LOCATIONS AS SHOWN ON STRUCTURAL DRAWINGS, SHALL BE PROTECTED WITH 30 # FELT OR PRESSURE TREATED IN ACCORDANCE WITH AITC-109. MEMBER SIZE SHOWN ARE NOMINAL.		

LOUVERED PERGOLA DESIGN & OPERATION:		
1. THE LOUVERED ROOFING SYSTEM SHALL BE INSTALLED WITH THE AZENCO R-BLADE PRODUCT.		
2. LOUVERS SHALL BE SPACED SUCH THAT THEY ALLOW 50% SYSTEM POROSITY WHEN FULLY OPENED.		
3. LOUVERS SHALL BE ROTATED TO A 90 DEGREE 'OPEN' POSITION DURING ANY NAMED STORM OR WEATHER DESIGN EVENT TO PREVENT EXCESSIVE FORCE TO THIS STRUCTURE. THE OWNER SHALL BE NOTIFIED IN WRITING OF THIS AND THAT CARE SHALL BE TAKEN TO AVOID BUILDUP OF SNOW, DEBRIS, CONSTRUCTION LOADS, & ANY OTHER FORCES THAT CAN AFFECT THE INTEGRITY OF THIS DESIGN.		
4. THE STRUCTURE SHALL BE POSTED WITH A LEGIBLE AND READILY VISIBLE DECAL OR PAINTED INSTRUCTIONS TO THE OWNER OR TENANT STATING TO REPOSITION THE LOUVERS AND WINDSCREENS DURING WIND OR SNOW ADVISORIES. THE CANOPY OWNER SHALL BE NOTIFIED OF THESE CONDITIONS BY THE PERMIT HOLDER AT THE TIME OF SALE.		
5. SYSTEM SHALL BE EQUIPPED AND TELECO CERTIFIED TO OPERATE PROPERLY WITH SENSORS WHICH WILL PROMPT THE LOUVER BLADES TO ROTATE TO THE OPEN (VERTICAL) POSITION WHENEVER THE WIND SPEED REACHES 45 MPH MINIMUM, OR TEMPERATURE DROPS TO 32° FAHRENHEIT OR LOWER, AND/OR INCLUDE THE ABILITY FOR MANUAL OPENING AND LOCKING BY THE USER.		
6. LOUVERED ROOF SYSTEM SHALL BE PER MANUFACTURER MAXIMUM SPANS OR BY OTHERS.		
7. NO CERTIFICATION IS OFFERED FOR WATERPROOFING, SIZING, OR OPERATION OF GUTTERS. SYSTEM NOT DESIGNED FOR WATERSHED OF RAINFALL FROM ADJACENT ROOFS UNLESS SPECIFICALLY SHOWN HEREIN, TYP.		

GENERAL NOTES		
1. ALLOWABLE DESIGN PRESSURES UTILIZED IN THIS DOCUMENT HAVE BEEN CALCULATED PER THE REQUIREMENTS OF THE CODES AND STANDARDS STATED HEREIN USING ASCE 7-16 ALLOWABLE STRESS DESIGN METHODOLOGY WITH THE CRITERIA AS OUTLINED HEREIN. THE CONTRACTOR SHALL CONTACT THE AUTHORITY HAVING JURISDICTION TO ENSURE APPROPRIATE CRITERIA TO BE USED BEFORE CONSTRUCTION BEGINS.		
2. DIMENSIONS ARE SHOWN TO ILLUSTRATE DESIGN FORCES AND OTHER DESIGN CRITERIA. IN SOME CASES, DETAILS MAY BE INTENTIONALLY ALTERED FOR PRESENTATION PURPOSES. DO NOT SCALE DIMENSIONS ELECTRONICALLY OR OTHERWISE. FIELD INSTALLATION MAY VARY SLIGHTLY BUT MUST REMAIN WITHIN 5% OF THE INTENDED DESIGN. THE CONTRACTOR IS TO VERIFY ALL DIMENSIONS PRIOR TO INSTALLATION.		
3. THIS STRUCTURE HAS BEEN DESIGNED AND SHALL BE FABRICATED IN ACCORDANCE WITH THE STRUCTURAL PROVISIONS OF THE ABOVE-REFERENCED BUILDING CODE. STRUCTURE SHALL BE FABRICATED IN ACCORDANCE WITH ALL GOVERNING CODES. THE CONTRACTOR SHALL INVESTIGATE AND CONFORM TO ALL LOCAL BUILDING CODE AMENDMENTS WHICH MAY APPLY AND GOVERN. DESIGN CRITERIA OR SPANS BEYOND STATED HEREIN MAY REQUIRE ADDITIONAL SITE-SPECIFIC SEALED ENGINEERING.		
4. THE CONTRACTOR SHALL CAREFULLY CONSIDER POSSIBLE IMPOSING LOADS ON ROOF, INCLUDING BUT NOT LIMITED TO ANY CONCENTRATED LOADS WHICH MAY JUSTIFY GREATER DESIGN CRITERIA. ALL STRUCTURAL MEMBERS AS SHOWN HAVE BEEN DESIGNED TO CARRY IN PLACE DESIGN LOADS ONLY; THE CONTRACTOR SHALL BE SOLELY RESPONSIBLE FOR THE SUPPORT OF ANY ADDITIONAL LOADS AND FORCES IMPOSED DURING MANUFACTURING, TRUCKING, ERECTING, AND HANDLING. SYSTEM NOT DESIGNED TO HANDLE CONCENTRATED LOADS FROM HUMAN ACTIVITY.		
5. SPECIAL INSPECTIONS MAY BE REQUESTED OR REQUIRED AT THE DISCRETION OF THE AUTHORITY HAVING JURISDICTION.		
6. GENERAL STRUCTURAL NOTES ARE APPLICABLE TO THE DESIGN AND CONSTRUCTION OF THE ENTIRE PROJECT AND THUS ARE APPLICABLE TO EVERY SHEET WITHIN THIS SET.		
7. SHOULD THE CONTRACTOR ENCOUNTER A CONFLICT BETWEEN THESE DRAWINGS AND ANY OTHER CONTRACT DOCUMENT OR APPLICABLE CODE OR STANDARD OF PRACTICE DURING BIDDING, THE PROVISION RESULTING IN THE GREATER COST APPLIES. SHOULD THE CONTRACTOR ENCOUNTER A CONFLICT DURING CONSTRUCTION, THE CONTRACTOR SHALL SUBMIT A WRITTEN REQUEST FOR CLARIFICATION TO THE DESIGN TEAM, WHO WILL PROVIDE A WRITTEN RESPONSE IN RETURN.		
8. THE CONTRACTOR SHALL SUPERVISE AND DIRECT ALL WORK AND SHALL BE RESPONSIBLE FOR CONSTRUCTION MEANS, METHODS, PROCEDURES, TECHNIQUES, AND SEQUENCE. THE CONTRACTOR HAS SOLE RESPONSIBILITY FOR THE QUALITY AND CORRECTNESS OF THE WORK.		
9. THE CONTRACTOR SHALL BE SOLELY RESPONSIBLE FOR COORDINATION OF THE STRUCTURAL WORK WITH OTHER TRADES INCLUDING, BUT NOT LIMITED TO: ARCHITECTURAL, CIVIL, AND MEP FOR FLOOR SLAB STEPS, SLOPES AND CURBS, FLOOR SLAB FINISH, OPENINGS IN STRUCTURAL FLOORS, ROOFS AND WALLS, ETC.		
10. THE BUILDING HAS BEEN DESIGNED BY THE STRUCTURAL ENGINEER OF RECORD TO RESIST THE CODE REQUIRED VERTICAL AND LATERAL FORCES IN ITS FULLY COMPLETED CONDITION. THE CONTRACTOR SHALL PROVIDE ALL REQUIRED BRACING, SHORING, AND OTHER CONSTRUCTION SUPPORTS NECESSARY TO ENSURE THE BUILDING'S STABILITY AND SAFETY THROUGHOUT THE DURATION OF CONSTRUCTION. FURTHER, THE CONTRACTOR SHALL NOT OVERLOAD THE STRUCTURE DURING CONSTRUCTION. THE CONTRACTOR SHALL RETAIN A LICENSED PROFESSIONAL ENGINEER TO PROVIDE THE ANALYSIS AND DESIGN NECESSARY TO DETERMINE POTENTIALLY OVERLOADED, UNSTABLE, OR HAZARDOUS CONDITIONS THAT MAY OCCUR AT ANY STAGE DURING CONSTRUCTION.		
11. THE CONTRACTOR SHALL VERIFY ALL EXISTING DIMENSIONS AND CONDITIONS AND COORDINATE WITH THE CONTRACT DOCUMENTS AND SHOP DRAWINGS.		
12. THE CONTRACTOR SHALL NOT EMPLOY CONSTRUCTION MEANS OR METHODS THAT MAY DAMAGE UTILITIES, ADJACENT BUILDINGS, OR PROPERTY. DOCUMENTATION OF ADJACENT CONDITIONS PRIOR TO CONSTRUCTION IS RECOMMENDED. FURTHER, THE CONTRACTOR SHALL EITHER ADEQUATELY CONFINE THE SITE OR PROTECT ADJACENT PROPERTY FROM DAMAGE.		
13. THE CONTRACTOR SHALL BE SOLELY RESPONSIBLE FOR PROJECT SAFETY AND OSHA REQUIREMENTS. SHOULD THE STRUCTURAL ENGINEER OF RECORD NOTIFY THE CONTRACTOR OF A POTENTIALLY UNSAFE CONDITION, IT IS SOLELY AS A COURTESY FROM ONE PROFESSIONAL TO ANOTHER. IT SHOULD NOT BE INTERPRETED AS THE STRUCTURAL ENGINEER OF RECORD ASSUMING ANY RESPONSIBILITY FOR PROJECT SAFETY.		
14. ALL STRUCTURES REQUIRE PERIODIC MAINTENANCE TO EXTEND LIFE SPAN AND ENSURE STRUCTURAL INTEGRITY FROM EXPOSURE TO THE ENVIRONMENT. A PLANNED PROGRAM OF MAINTENANCE SHALL BE ESTABLISHED BY THE BUILDING OWNER. THIS PROGRAM SHALL INCLUDE, BUT NOT BE LIMITED TO: PAINTING OF STRUCTURAL STEEL, PROTECTIVE COATINGS FOR CONCRETE, SEALANTS, CALLED JOINTS, EXPANSION JOINTS, CONTROL JOINTS, SPALLS AND CRACKS IN CONCRETE, AND PRESSURE WASHING OF EXPOSED STRUCTURAL ELEMENTS EXPOSED TO A SALINE OR OTHER HARSH CHEMICAL ENVIRONMENT.		
15. THE BUILDING OWNER SHALL NOT ALTER OR MODIFY ANY STRUCTURAL ELEMENT WITHOUT CONSULTING A LICENSED PROFESSIONAL ENGINEER. FURTHER, BUILDING OWNER SHALL NOT RENOVATE, REPURPOSE, ADD-ON TO, OR OTHERWISE MODIFY THE EXISTING STRUCTURAL SYSTEMS WITHOUT CONSULTING A LICENSED PROFESSIONAL ENGINEER.		

ELECTRONIC DATA/REPRODUCTION		
1. ALL INFORMATION CONTAINED IN THE ELECTRONIC FILES OF THE CONTRACT DOCUMENTS ARE INSTRUMENTS OF SERVICE OF THE ARCHITECT/STRUCTURAL ENGINEER OF RECORD AND SHALL NOT BE USED FOR OTHER PROJECTS, ADDITIONS TO THE PROJECT, OR THE COMPLETION OF THE PROJECT BY OTHERS. ELECTRONIC FILES OF THE STRUCTURAL DOCUMENTS REMAIN THE PROPERTY OF AM STRUCTURES AND IN NO CASE SHALL THEIR TRANSFER BE CONSIDERED A SALE.		
2. THE USE OF ELECTRONIC FILES OR REPRODUCTIONS OF THESE CONTRACT DOCUMENTS BY ANY CONTRACTOR, SUBCONTRACTOR, ERECTOR, FABRICATOR, OR MATERIAL SUPPLIER IN LIEU OF PREPARATION OF SHOP DRAWINGS SIGNIFIES THEIR ACCEPTANCE OF ALL INFORMATION SHOWN HEREIN AS CORRECT AND OBLIGATES THEMSELVES TO ANY JOB EXPENSE, REAL OR IMPLIED, ARISING DUE TO ANY ERRORS OR OMISSIONS THAT MAY OCCUR HEREIN. THE USE OF ELECTRONIC FILES DOES NOT RELIEVE THE CONTRACTOR'S RESPONSIBILITY FOR PROPER CHECKING AND COORDINATION OF DIMENSIONS, DETAILS, SIZE, AND QUANTITIES.		
3. DIMENSIONS AND ELEMENT SIZES AND LOCATIONS IN THE ELECTRONIC FILES MAY NOT BE PRECISE AND, IN SOME CASES, HAVE BEEN INTENTIONALLY ALTERED FOR PRESENTATION PURPOSES. DO NOT SCALE DIMENSIONS ELECTRONICALLY OR OTHERWISE.		
4. WHEN USED FOR THE PREPARATION OF SHOP DRAWINGS, ALL INFORMATION NOT APPLICABLE TO THE SUBCONTRACT SHALL BE REMOVED FROM THE DRAWINGS, INCLUDING, BUT NOT LIMITED TO: SHEET NUMBERS, SECTION MARKS, TITLE BLOCKS, AND REFERENCES TO THE CONTRACT DOCUMENTS.		



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ENGINEER'S SEAL



STATE OF OHIO
ANDREW MCCANN
REGISTERED PROFESSIONAL ENGINEER
E-87255
Exp. 12/31/26

THIS ITEM HAS BEEN DIGITALLY SIGNED AND SEALED BY ANDREW MCCANN, P.E. ON 07/19/2024. PRINTED COPIES OF THIS DOCUMENT ARE NOT CONSIDERED SIGNED AND SEALED AND THE SIGNATURE MUST BE VERIFIED ON ANY ELECTRONIC COPIES.

TIMAN WINDOW TREATMENTS, INC.

RAYMOND TATKO

1751 E Hines Hill Rd, Hudson, OH 44236

ISSUANCE SCHEDULE		
NO.	DESCRIPTION	DATE
0.	INITIAL ISSUANCE	07/19/2024

GENERAL NOTES

PROJECT # 24577
DATE 07/19/2024
DWN BZ
CHK AEM

S1

SCALE: AS INDICATED

STRUCTURAL DRAWING LIST	
SHEET NUMBER	SHEET NAME
S1	GENERAL NOTES
S2	GENERAL NOTES
S3	FOUNDATION PLAN
S4	ROOF FRAMING PLAN
S5	REFLECTED CEILING PLAN
S6	ELEVATIONS
S7.0	DETAILS
S7.1	DETAILS CONT
S7.2	DETAILS CONT

ANCHORS & FASTENERS

- ALL BOLTS SHALL BE HOT DIPPED GALVANIZED, OR STAINLESS STEEL & MEET THE REQUIREMENTS OF ASTM A307 GRADE A. WASHERS SHALL BE USED BETWEEN WOOD & BOLT HEAD & BETWEEN WOOD & NUT CONFORMING TO FEDERAL SPECIFICATION FF-W-92 FOR WASHERS. NUTS SHALL BE INSTALLED SUCH THAT THE END OF THE THREADED ROD OR BOLT IS AT LEAST FLUSH WITH THE TOP OF NUT.
- BOLT HOLES SHALL BE AT LEAST A MINIMUM OF 1/32" AND NO MORE THAN A MAXIMUM OF 1/16" LARGER THAN THE BOLT DIAMETER.
- NAILS SHALL PENETRATE THE SECOND MEMBER A DISTANCE EQUAL TO THE THICKNESS OF THE MEMBER BEING NAILED THERETO. THERE SHALL BE NOT LESS THAN 2 NAILS IN ANY CONNECTION.
- DRY WOOD MAY SPLIT MORE EASILY. IF WOOD TENDS TO SPLIT, PRE-BORING HOLES SHALL BE USED WITH DIAMETERS NOT EXCEEDING 3/4" OF THE NAIL DIAMETER OR USE A 5/32" BIT FOR SDS SCREWS. A FASTENER THAT SPLITS THE WOOD SHALL BE REEVALUATED PRIOR LOADING THE CONNECTION.
- ANCHORS SHALL BE INSTALLED IN ACCORDANCE WITH MANUFACTURERS' RECOMMENDATIONS. MINIMUM EMBEDMENT SHALL BE AS NOTED HEREIN. MINIMUM EMBEDMENT AND EDGE DISTANCE ARE DEPTHS INTO SOLID SUBSTRATE AND DO NOT INCLUDE THICKNESS OF STUCCO, FOAM, BRICK, AND OTHER WALL FINISHES.
- ANCHOR QUANTITIES INDICATED IN DETAILS ARE FOR GRAPHICAL PURPOSES ONLY. DO NOT SCALE DIAMETER, LENGTH, OR PENETRATION(S). HEAD STYLE(S) ARE FREELY INTERCHANGEABLE.
- MECHANICAL ANCHORS (EXPANSION ANCHORS/EXPANSION BOLTS) INTO EXISTING CONCRETE AS SHOWN ON THE STRUCTURAL DRAWINGS SHALL BE ONE OF THE FOLLOWING PRODUCTS:
 - KWIK BOLT TZ ANCHORS MANUFACTURED BY HILTI FASTENING SYSTEMS
 - STRONG-BOLT 2 ANCHORS MANUFACTURED BY SIMPSON STRONGTIE COMPANY
 - POWER-STUD+ SD2 ANCHORS MANUFACTURED BY DEWALT
- MECHANICAL ANCHORS (EXPANSION ANCHORS/EXPANSION BOLTS) INTO EXISTING CONCRETE MASONRY AS SHOWN ON THE STRUCTURAL DRAWINGS SHALL BE ONE OF THE FOLLOWING PRODUCTS:
 - KWIK BOLT 3 ANCHORS MANUFACTURED BY HILTI FASTENING SYSTEMS
 - WEDGE-ALL ANCHORS MANUFACTURED BY SIMPSON STRONGTIE COMPANY
 - POWER-STUD+ SD1 ANCHORS MANUFACTURED BY DEWALT
- SCREW ANCHORS INTO EXISTING CONCRETE AND CONCRETE MASONRY AS SHOWN ON THE STRUCTURAL DRAWINGS SHALL BE ONE OF THE FOLLOWING PRODUCTS:
 - KWIK HUS EZ ANCHORS MANUFACTURED BY HILTI FASTENING SYSTEMS
 - TITEN HD ANCHORS MANUFACTURED BY SIMPSON STRONGTIE COMPANY
 - SCREW-BOLT+ ANCHORS MANUFACTURED BY DEWALT
- ADHESIVE ANCHORS (EPOXY ANCHORS/DRILL & EPOXY) INTO EXISTING CONCRETE AS SHOWN ON THE STRUCTURAL DRAWINGS SHALL BE ONE OF THE FOLLOWING ADHESIVE PRODUCTS:
 - HIT-HY200 EPOXY ADHESIVE WITH HAS ROD MANUFACTURED BY HILTI FASTENING SYSTEMS
 - AT-XP ADHESIVE MANUFACTURED BY SIMPSON STRONGTIE COMPANY WITH AN ALL-THREAD F1554 GRADE 36 STEEL ROD
 - PURE110+ EPOXY ADHESIVE MANUFACTURED BY DEWALT WITH AN ALL-THREAD F1554 GRADE 36 STEEL ROD
- ADHESIVE ANCHORS (EPOXY ANCHORS/DRILL & EPOXY) INTO EXISTING CONCRETE MASONRY AS SHOWN ON THE STRUCTURAL DRAWINGS SHALL BE ONE OF THE FOLLOWING ADHESIVE PRODUCTS:
 - HIT-HY70 INJECTION ADHESIVE WITH HAS ROD MANUFACTURED BY HILTI FASTENING SYSTEMS
 - AT-XP ADHESIVE MANUFACTURED BY SIMPSON STRONGTIE COMPANY WITH AN ALL-THREAD F1554 GRADE 36 STEEL ROD
 - AC1010+ GOLD MANUFACTURED BY DEWALT WITH AN ALL-THREAD F1554 GRADE 36 STEEL ROD
- ADHESIVE FOR ANCHORING REINFORCING BARS (REBAR) INTO EXISTING CONCRETE AS SHOWN ON THE STRUCTURAL DRAWINGS SHALL BE ONE OF THE FOLLOWING ADHESIVE PRODUCTS:
 - HIT-HY200 EPOXY ADHESIVE MANUFACTURED BY HILTI FASTENING SYSTEMS
 - AT-XP ADHESIVE MANUFACTURED BY SIMPSON STRONGTIE COMPANY
 - PURE110+ EPOXY ADHESIVE MANUFACTURED BY DEWALT
- THE GENERAL CONTRACTOR SHALL OBTAIN APPROVAL FROM THE STRUCTURAL ENGINEER OF RECORD PRIOR TO USING POST-INSTALLED ANCHORS FOR MISSING OR MISPLACED CAST-IN-PLACE ANCHORS.
- SUBSTITUTION REQUESTS FOR ALTERNATIVE PRODUCTS SHALL BE SUBMITTED TO THE STRUCTURAL ENGINEER OF RECORD WITH CALCULATIONS THAT ARE PREPARED AND SEALED BY A LICENSED PROFESSIONAL ENGINEER. CALCULATIONS SHALL SHOW THAT THE SUBSTITUTED PRODUCT WILL ACHIEVE AN EQUIVALENT CAPACITY USING THE APPROPRIATE DESIGN PROCEDURE REQUIRED BY THE REFERENCED BUILDING CODE.
- LOCATE, BY NON-DESTRUCTIVE MEANS, ALL EXISTING REINFORCEMENT, AND AVOID DURING INSTALLATION OF ANCHORS. IF EXISTING REINFORCEMENT LAYOUT PROHIBITS THE INSTALLATION OF ANCHORS AS INDICATED ON THE STRUCTURAL DRAWINGS, THE CONTRACTOR SHALL NOTIFY THE STRUCTURAL ENGINEER OF RECORD IMMEDIATELY.
- HILES SHALL BE DRILLED AND CLEANED, AND ANCHORS SHALL BE INSTALLED PER THE MANUFACTURER'S PUBLISHED INSTALLATION INSTRUCTIONS. DEFECTIVE OR ABANDONED HOLES SHALL BE FILLED WITH NON-SHRINK GROUT OR AN INJECTABLE ADHESIVE MATCHING THE ADJACENT CONCRETE COMPRESSIVE STRENGTH.
- MASONRY ANCHORS SHALL NOT BE INSTALLED IN HOLLOW CORE MASONRY. IF INSTALLATION INTO HOLLOW CORE MASONRY IS DESIRED, SUBMIT ALTERNATIVE PRODUCT FOR REVIEW AND APPROVAL BY THE STRUCTURAL ENGINEER OF RECORD.
- MASONRY ANCHORS SHALL NOT BE INSTALLED IN HEAD JOINTS.
- IN ADDITION TO THE MANUFACTURER'S PRINTED INSTALLATION INSTRUCTIONS, THE FOLLOWING GUIDELINES SHALL BE FOLLOWED FOR INSTALLATION OF ADHESIVE ANCHORS:
 - ADHESIVE ANCHORS SHALL BE INSTALLED IN CONCRETE HAVING A MINIMUM AGE OF 21 DAYS AT TIME OF ANCHOR INSTALLATION. CONCRETE SHALL HAVE A MINIMUM COMPRESSIVE STRENGTH OF 2,500 PSI AT THE TIME OF INSTALLATION UNLESS HIGHER STRENGTH IS REQUIRED PER THE MANUFACTURER'S PRINTED INSTALLATION INSTRUCTIONS.
 - ADHESIVE ANCHORS SHALL BE INSTALLED IN DRY CONCRETE, AND DURING DRY CONDITIONS.
 - ADHESIVE ANCHORS SHALL BE INSTALLED IN HOLES PREDRILLED WITH A CARBIDE TIPPED DRILL BIT.
 - ADHESIVE ANCHORS SHALL BE INSTALLED WITHIN THE TEMPERATURE RANGE SPECIFIED IN THE MANUFACTURER'S PRINTED INSTALLATION INSTRUCTIONS, BUT NOT OUTSIDE OF THE DESIGN TEMPERATURE RANGE. LOADS SHALL NOT BE APPLIED TO ADHESIVE ANCHORS UNTIL THE FULL CURING TIME ASSOCIATED WITH THE INSTALLATION TEMPERATURE HAS ELAPSED.
- INSTALLATION OF ADHESIVE ANCHORS SHALL BE PERFORMED BY PERSONNEL CERTIFIED BY AN APPLICABLE CERTIFICATION PROGRAM. CERTIFICATION SHALL INCLUDE WRITTEN AND PERFORMANCE TESTS IN ACCORDANCE WITH THE ACI/CRSI ADHESIVE ANCHOR INSTALLER CERTIFICATION PROGRAM, OR EQUIVALENT.
- SPECIAL INSPECTIONS SHALL BE PROVIDED FOR POST-INSTALLED ANCHORS IN ACCORDANCE WITH THE ANCHOR MANUFACTURER'S PRINTED INSTALLATION INSTRUCTIONS AND/OR EVALUATION REPORTS, UNLESS MORE SPECIFIC REQUIREMENTS ARE SPECIFIED IN THE CONSTRUCTION DOCUMENTS.
- CONTINUOUS INSPECTION SHALL BE PROVIDED FOR ADHESIVE ANCHORS INSTALLED IN HORIZONTAL OR UPWARDLY INCLINED ORIENTATIONS TO RESIST SUSTAINED TENSILE LOADS.
- HOLE DRILLING AND INSTALLATION OF ADHESIVE ANCHORS SHALL BE IN ACCORDANCE WITH MANUFACTURER'S PRINTED INSTALLATION INSTRUCTIONS. ANCHORS SHALL BE INSTALLED IN CONCRETE DRY CONDITION.

ABBREVIATIONS

ADDL	ADDITIONAL
ADJ	ADJACENT
AFF	ABOVE FINISHED FLOOR
ALT	ALTERNATE
APPROX	APPROXIMATE
ARCH	ARCHITECT OR ARCHITECTURAL
ASD	ALLOWABLE STRESS DESIGN
B/	BOTTOM OF
BLDG	BUILDING
BLKG	BLOCKING
BP	BASE PLATE
BOT	BOTTOM
BTWN	BETWEEN

C	COMPRESSION
CFS	COLD-FORMED STEEL
CIP	CAST-IN-PLACE
CL	CENTER LINE
CLR	CLEAR OR CLEARANCE
CMU	CONCRETE MASONRY UNIT
COL	COLUMN
CONC	CONCRETE
CONN(S)	CONNECTION(S)
CONST	CONSTRUCTION
CONT	CONTINUOUS
COORD	COORDINATE

D&E	DRILL & EPOXY
db	REINFORCING BAR DIAMETER
°	DEGREE(S)
Ø	DIAMETER
DIAG	DIAGONAL
DIM(S)	DIMENSION(S)
DL	DEAD LOAD
DWG(S)	DRAWING(S)

ABBREVIATIONS

EA	EACH
EF	EACH FACE
EJ	EXPANSION JOINT
EL	ELEVATION
EMB	EMBED
EOS	EDGE-OF-SLAB
EQ	EQUAL
EW	EACH WAY
EXIST	EXISTING
EXP	EXPANSION
EXT	EXTERIOR
F/F	FACE-TO-FACE
FF	FINISH FLOOR
FND	FOUNDATION
FS	FAR SIDE
FT	FEET
FTG	FOOTING
GA	GAGE, GAUGE
GALV	GALVANIZED
GB	GRADE BEAM
GC	GENERAL CONTRACTOR
HCA	HEADED CONCRETE ANCHORS
HORIZ	HORIZONTAL
HSS	HOLLOW STRUCTURAL SECTION

ID	INSIDE DIAMETER
IF	INSIDE FACE
IN	INCH
INFO	INFORMATION
INT	INTERIOR
JST(S)	JOIST(S)

ABBREVIATIONS

K	KIPS (1,000 POUNDS)
KLF	KIP PER LINEAR FOOT
KSF	KIP PER SQUARE FOOT
KSI	KIPS PER SQUARE INCH
L	LENGTH
LB(S)	POUND(S)
LL	LIVE LOAD
LLH	LONG LEG HORIZONTAL
LLV	LONG LEG VERTICAL
LRFD	LOAD RESISTANCE FACTORED DESIGN
LSH	LONG SIDE HORIZONTAL
LSV	LONG SIDE VERTICAL
LTS	LAP TENSION SPLICE
LWC	LIGHT WEIGHT CONCRETE

M	MOMENT
MAX	MAXIMUM
MC	MOMENT CONNECTION(S)
MEP	MECHANICAL, ELECTRICAL, PLUMBING, FIRE PROTECTION
MFR	MANUFACTURER
MID	MIDDLE
MIN	MINIMUM
NIC	NOT IN CONTRACT
NS	NEAR SIDE
NTS	NOT TO SCALE
NWC	NORMAL WEIGHT CONCRETE

OC	ON CENTER
OD	OUTSIDE DIAMETER
OF	OUTSIDE FACE
OPNG(S)	OPENING(S)
OPP	OPPOSITE

ABBREVIATIONS

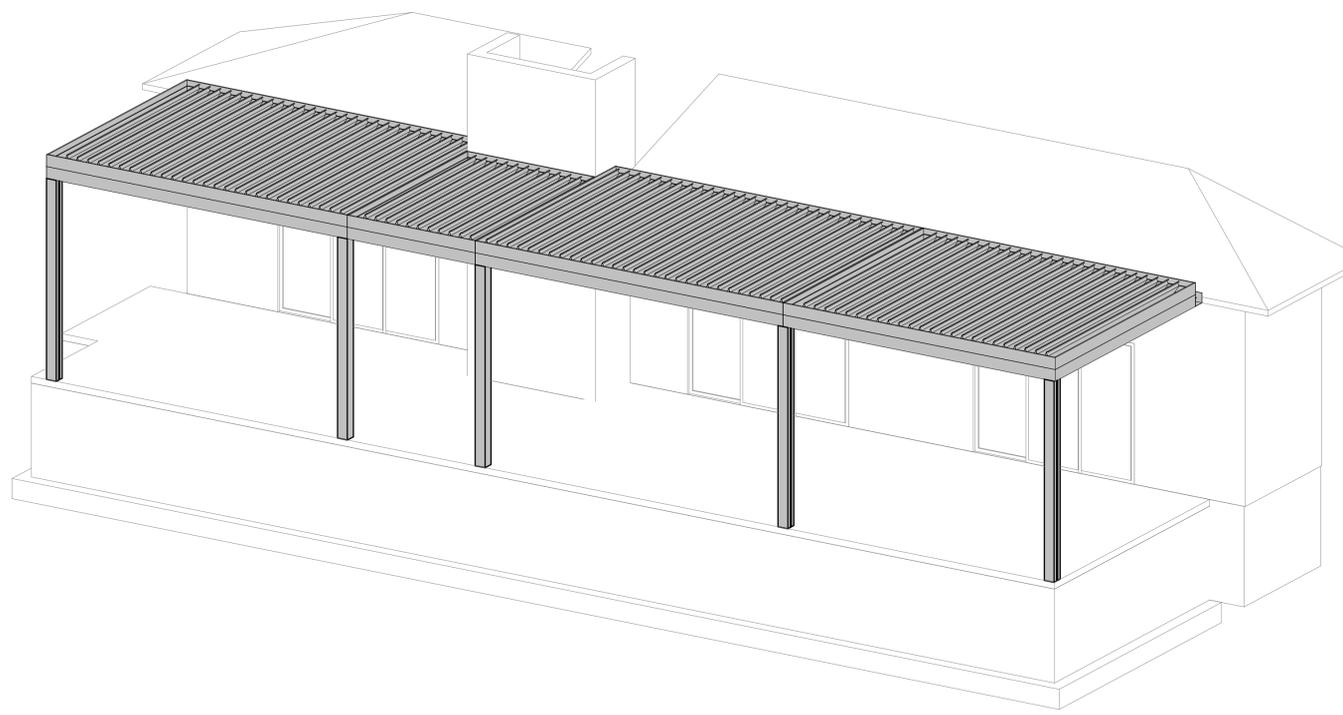
PAF	POWDER ACTUATED FASTENER
PERP	PERPENDICULAR
PL	PLATE
PLF	POUNDS PER LINEAL FOOT
PREFAB	PRE-FABRICATED
PSF	POUNDS PER SQUARE FOOT
PSI	POUNDS PER SQUARE INCH
PT	POST-TENSIONED

REF	REFERENCE
REINF	REINFORCED(D) (ING) OR (MENT)
REQ(D)	REQUIRE(D)
REV	REVISION
RTU	ROOF TOP UNIT
SCHED	SCHEDULE(D)
SF	SQUARE FOOT (FEET)
SIM	SIMILAR
SMS	SHEET METAL SCREW
SOG	SLAB-ON-GROUND
SPEC(S)	SPECIFICATION(S)
SS	STAINLESS STEEL
STD	STANDARD
STIFF	STIFFENER
SYM	SYMMETRICAL

T	TENSION
TSB	TOP AND BOTTOM
T/	TOP OF
TYP	TYPICAL

UNO	UNLESS NOTED OTHERWISE
V	SHEAR
VERT	VERTICAL
VIF	VERIFY IN FIELD

W/	WITH
W/O	WITHOUT
WWR	WELDED WIRE REINFORCEMENT



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TIMAN WINDOW TREATMENTS, INC.

RAYMOND TATKO

1751 E Hines Hill Rd, Hudson, OH 44236

ISSUANCE SCHEDULE

NO.	DESCRIPTION	DATE
0.	INITIAL ISSUANCE	07/19/2024

GENERAL NOTES

PROJECT #	24577
DATE	07/19/2024
DWN	BZ
CHK	AEM

S2

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ISSUANCE SCHEDULE

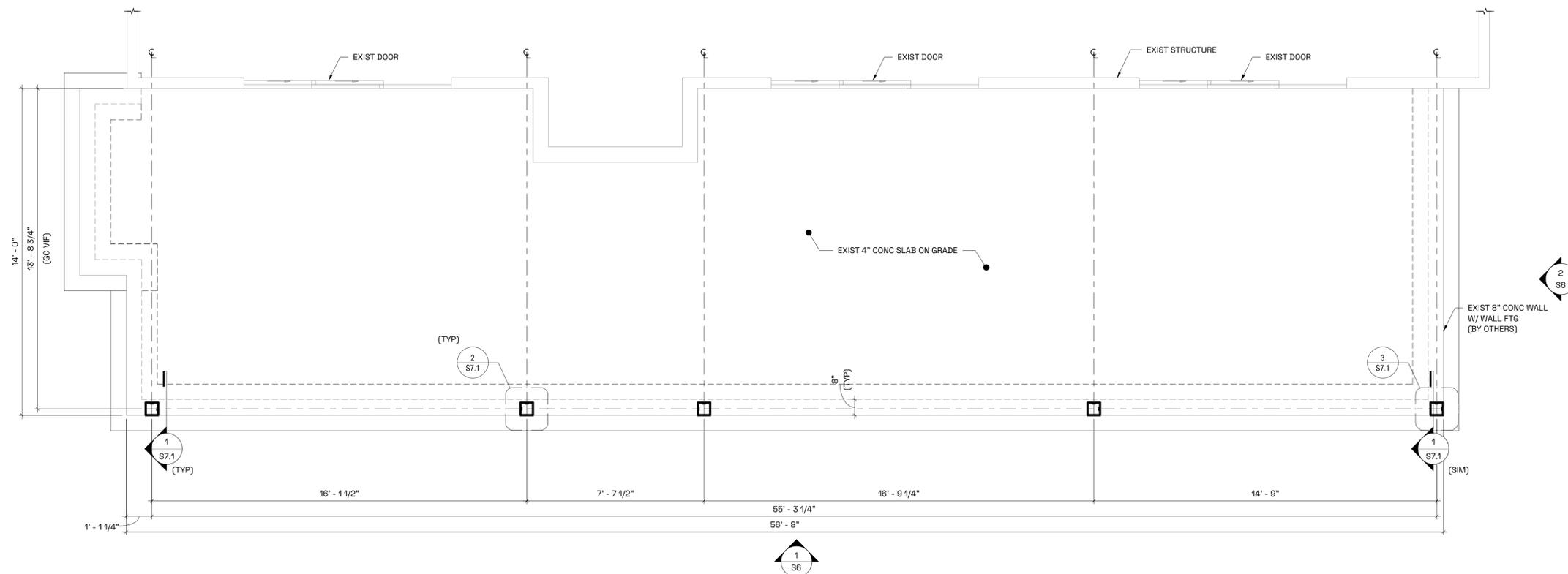
NO.	DESCRIPTION	DATE
0.	INITIAL ISSUANCE	07/19/2024

FOUNDATION PLAN

PROJECT #	24577
DATE	07/19/2024
DWN	BZ
CHK	AEM

S3

SCALE: AS INDICATED



1

FOUNDATION PLAN

3/8" = 1'-0"

NOTES:

- CONTRACTOR SHALL FIELD VERIFY ALL DIMENSIONS.

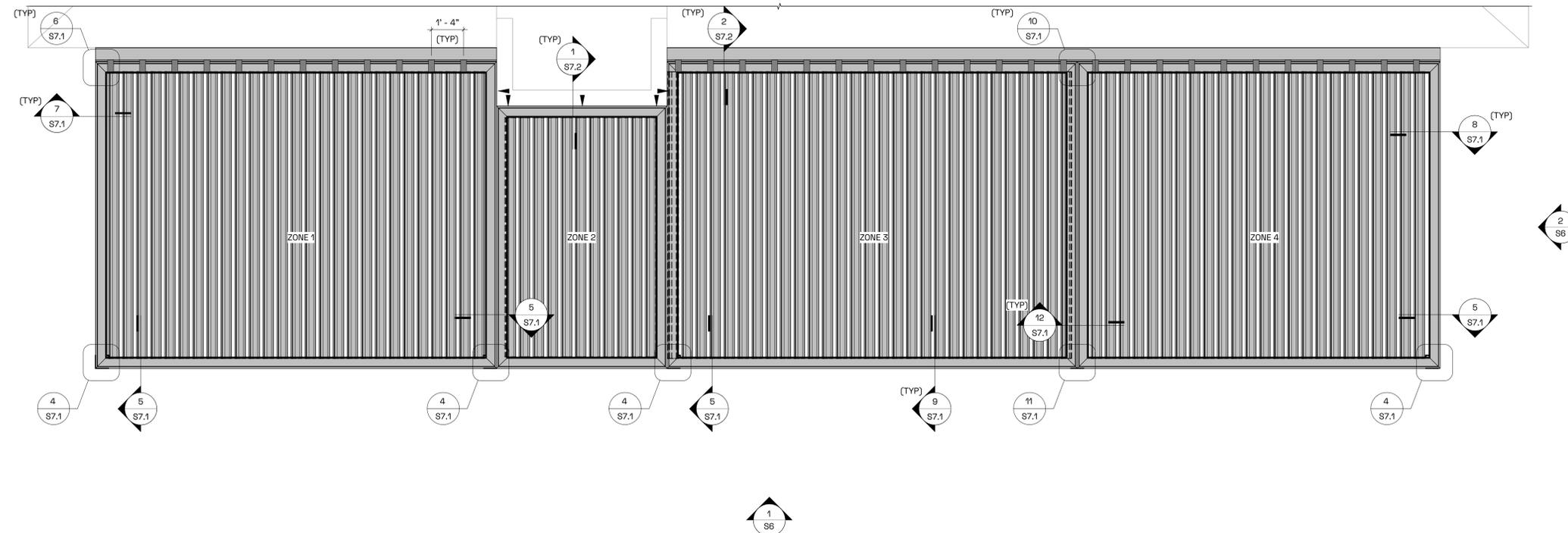


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1 ROOF FRAMING PLAN

3/8" = 1'-0"

NOTES:

1. ■ DENOTES SUPER GUTTER STRAP.
2. ▼ DENOTES (2) (K) BOLTS
3. CONTRACTOR SHALL FIELD VERIFY ALL DIMENSIONS
4. ZONES 1-4 DENOTE R-BLADE AZENCO PRODUCT.

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ROOF FRAMING PLAN

PROJECT #	24577
DATE	07/19/2024
DWN	BZ
CHK	AEM

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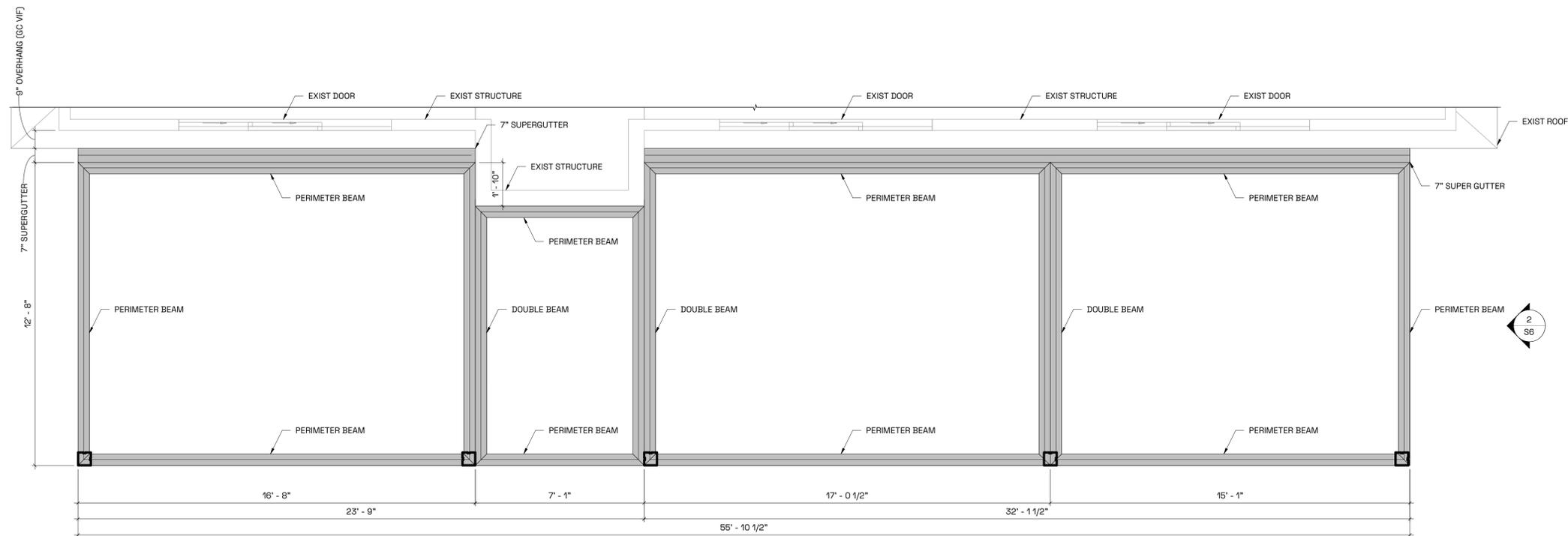
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0.	INITIAL ISSUANCE	07/19/2024

REFLECTED CEILING PLAN

PROJECT #	24577
DATE	07/19/2024
DWN	BZ
CHK	AEM

S5

SCALE: AS INDICATED



1

REFLECTED CEILING PLAN

3/8" = 1'-0"



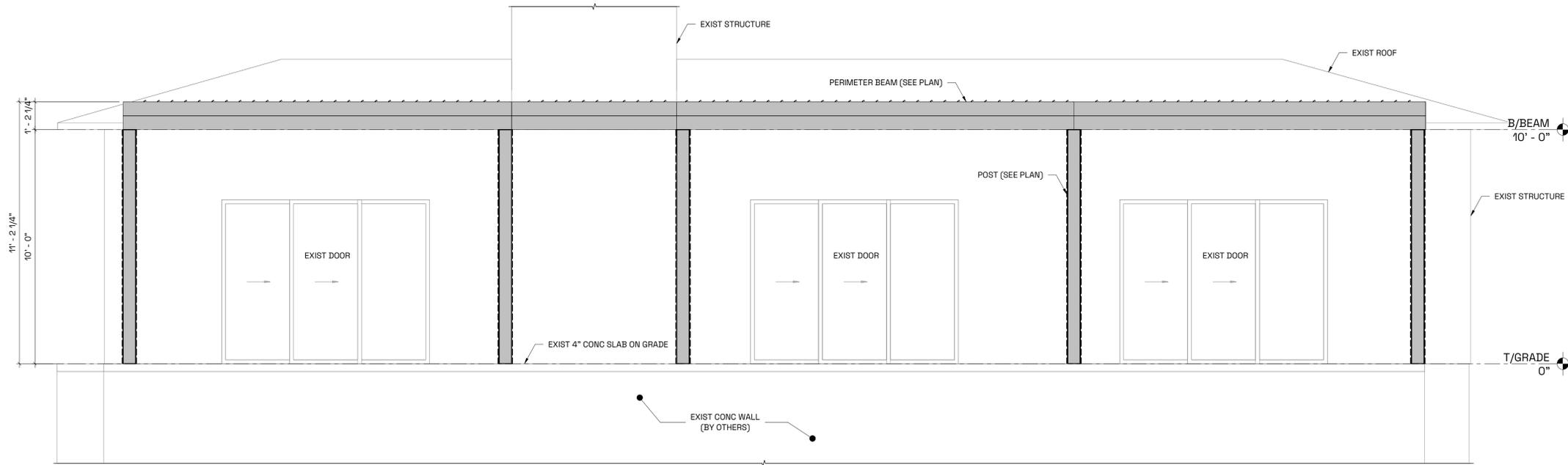


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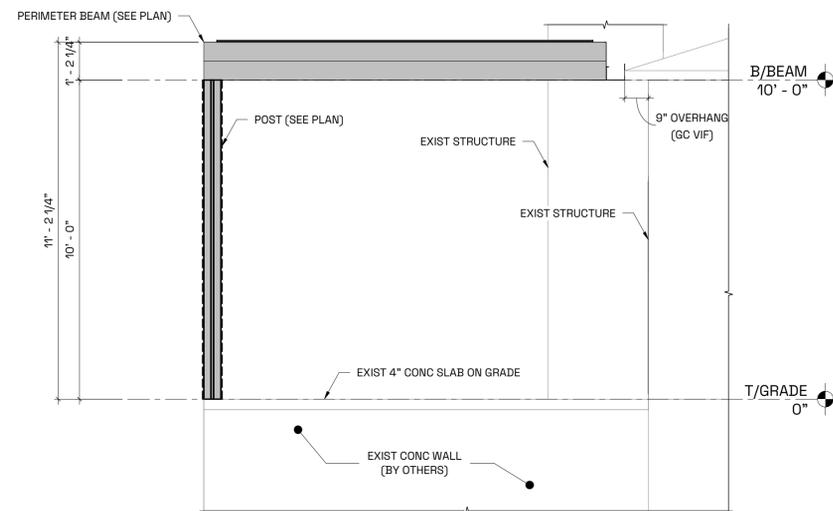
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1 FRONT ELEVATION
 3/8" = 1'-0"



2 SIDE ELEVATION
 3/8" = 1'-0"

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RAYMOND TATKO
 1751 E Hines Hill Rd, Hudson, OH 44236

ISSUANCE SCHEDULE

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ELEVATIONS

PROJECT #	24577
DATE	07/19/2024
DWN	BZ
CHK	AEM

S6

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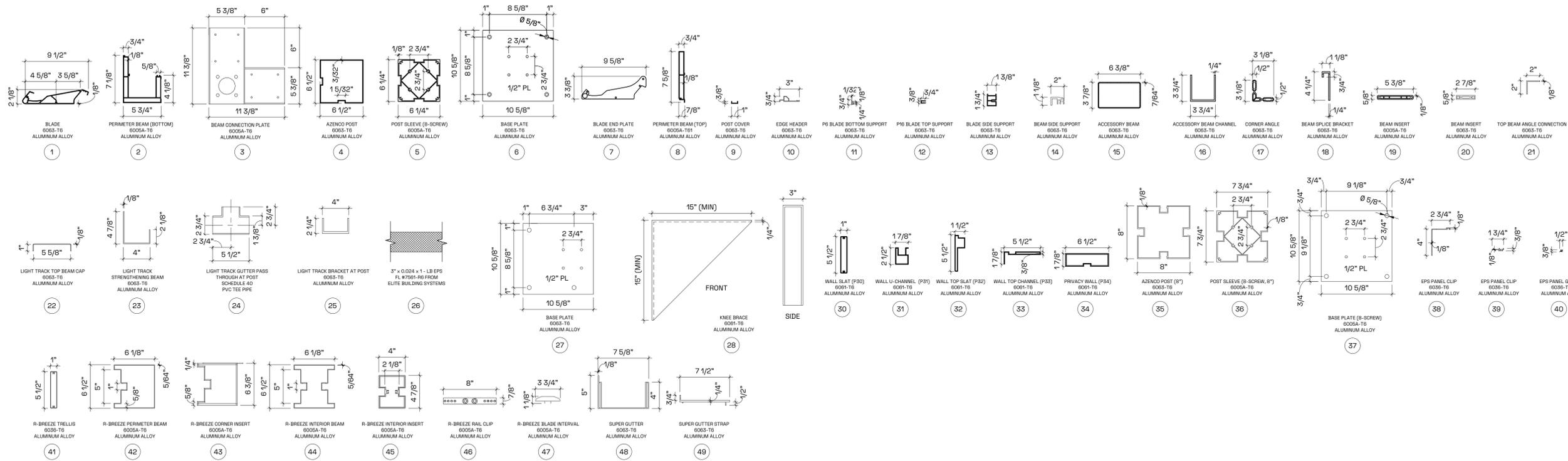


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A STRUCTURAL COMPONENTS

1/2" = 1'-0"

AA		5/8" Ø X 5" SIMPSON TITEN HD ANCHORS 316 SS
BB		5/8" Ø X 6" SIMPSON TITEN HD ANCHORS 316 SS
CC		3/8" Ø X 3" SIMPSON TITEN HD ANCHORS 316 SS
DD		1/2" Ø X 4" SIMPSON TITEN HD ANCHORS 316 SS
EE		3/4" Ø X 7" SIMPSON TITEN HD ANCHORS 316 SS
A		1/4" Ø X 1" HWH SDS 410 SS
B		1/4" Ø X 3" HWH SDS 410 SS
C		3/8" Ø X 5" HEX LAG SCREW SS
D		3/8" Ø X 6" HEX LAG SCREW SS

E		3/8" Ø X 6" HEX LAG SCREW SS
F		5/8" Ø X 6" HEX LAG SCREW SS
G		1/2" Ø X 6" HEX LAG SCREW SS
H		1/4" Ø X 3" HEX LAG SCREW SS
I		3/8" Ø X 5" HEX SCREW ANCHOR 316 SS
J		#8 X 1" HWH SDS 410 SS
K		1/2" Ø X 6" DEWALT SCREW BOLT 316 SS
L		RISER KIT: (1) 3/8"-16 Ø X 7" SS HEX CAP BOLT, (2) 3/8" X 7/8" SS FLAT WASHERS, (3) 3/8"-16 Ø SS HEX NUT
M		1/4" Ø X 4" HWH SDS 410 SS
N		1/2" Ø F1654 THREADED ROD W/ DEWALT PURE110+ EPOXY
TS		#14 X 1" HWH SDS 410 SS

Q1		M12 X 120mm PIN
Q2		M4.8 X 16mm PAN HEAD TORX BIT SHEET METAL SCREW
Q3		#14 X 3/4" HEX HEAD SELF-DRILL 316 SS SCREW
Q4		M8 X 40mm PAN HEAD TORX BIT SHEET METAL SCREW
Q5		HTK10-1.00 HEX SELF-TAPPING 410SS #10 X 1"
Q6		M8 X 40mm FLAT HEAD TORX BIT SHEET METAL SCREW
Q7		M4.8 X 25mm FLAT HEAD TORX BIT SHEET METAL SCREW
Q12		M8 X 80mm SOCKET CAP HEX HEAD PARTIALLY THREADED SCREW
Q13		M8 SS NUT
Q14		M5.5 X 25mm PAN HEAD SCREW
Q15		M3.5 X 80mm PIN
Q17		1/8" Ø X 3/4" PIN
Q18		#14 X 2" SMS SCREW SS410
Q21		1/2" Ø X 2" SS HEX HEAD SCREW
Q22		1/2" Ø SS NUT

B STRUCTURAL CONNECTORS

8" = 1'-0"

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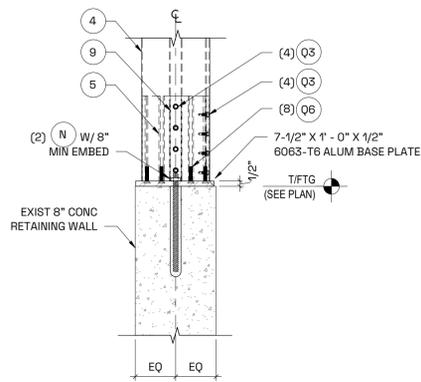
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0.	INITIAL ISSUANCE	07/19/2024

DETAILS CONT

PROJECT #	24577
DATE	07/19/2024
DWN	BZ
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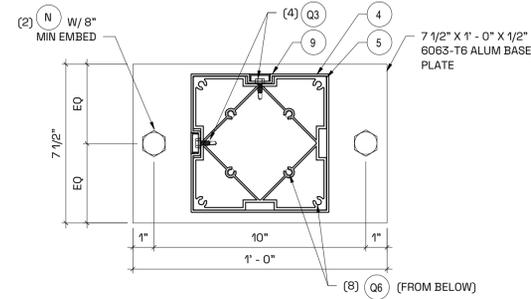
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TYPICAL POST BASE TO EXIST FOOTING

1 SECTION

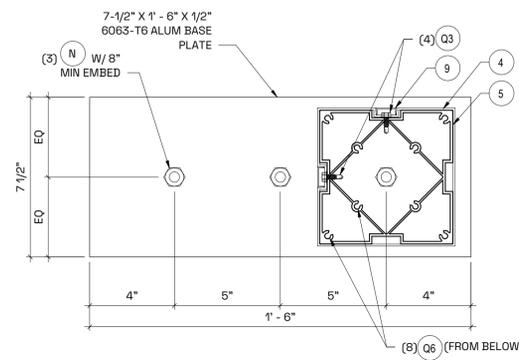
1 1/2" = 1'-0"



POST AND BASE PLATE

2 DETAIL

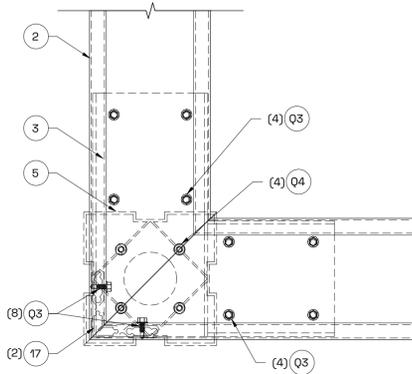
3" = 1'-0"



ECCENTRIC POST AND BASE PLATE (8-SCREW) (FLUSH EDGE)

3 DETAIL

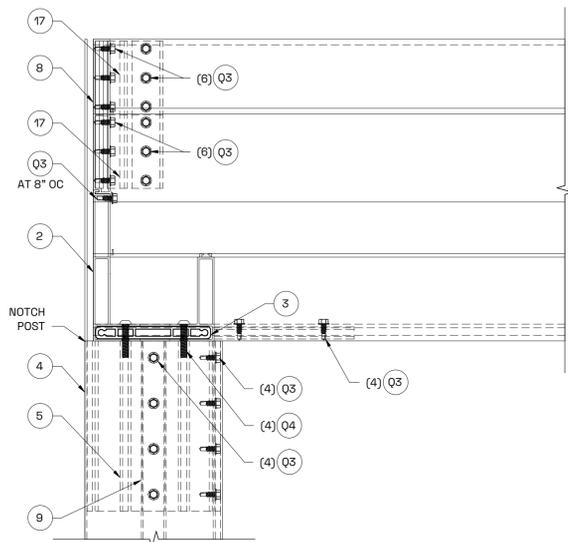
3" = 1'-0"



BEAM CORNER ASSEMBLY

4 DETAIL

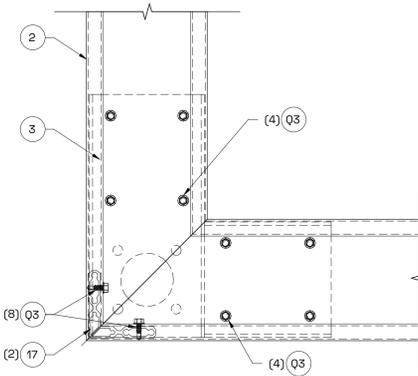
3" = 1'-0"



POST TO BEAM AT CORNER

5 SECTION

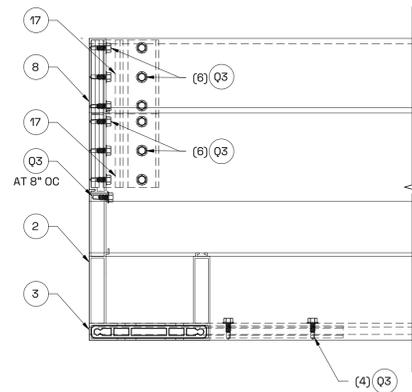
3" = 1'-0"



BEAM CORNER ASSEMBLY (NO POST)

6 DETAIL

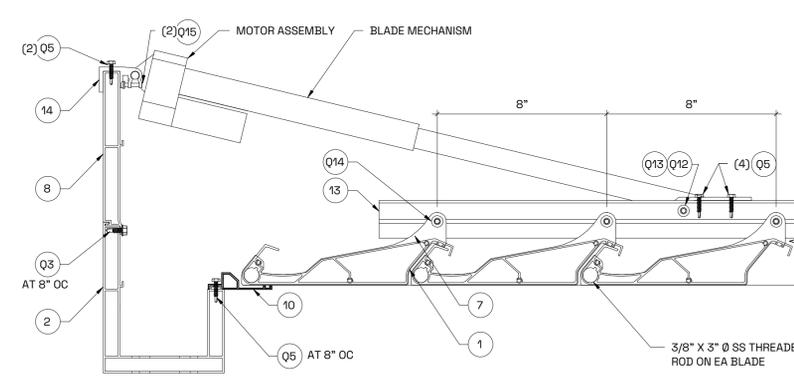
3" = 1'-0"



POST TO BEAM AT CORNER (NO POST)

7 SECTION

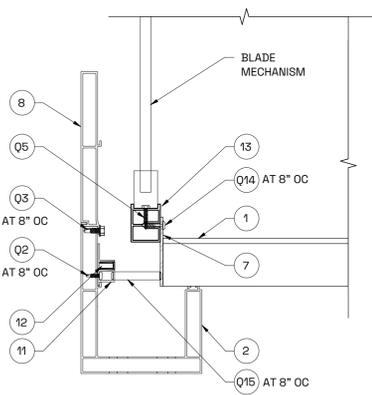
3" = 1'-0"



LOUVER ASSEMBLY AT BEAM - PERPENDICULAR

8 DETAIL

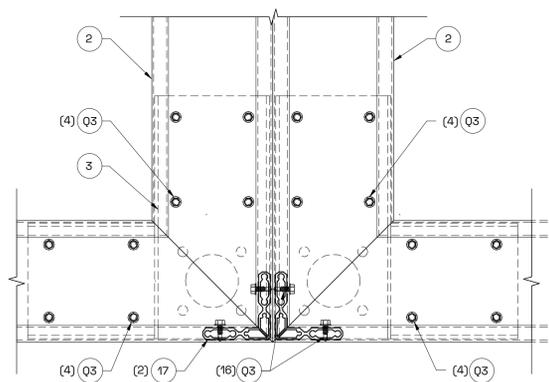
3" = 1'-0"



LOUVER ASSEMBLY AT BEAM - PARALLEL

9 SECTION

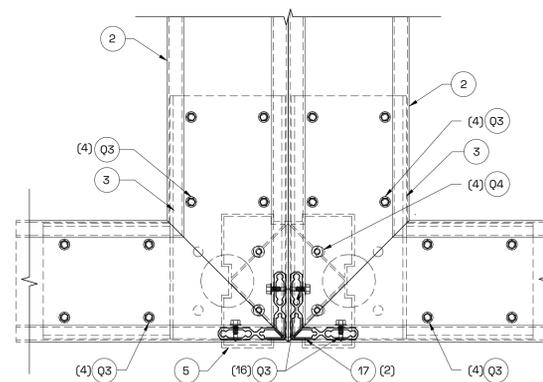
3" = 1'-0"



DOUBLE BEAM CORNER ASSEMBLY - NO POST

10 DETAIL

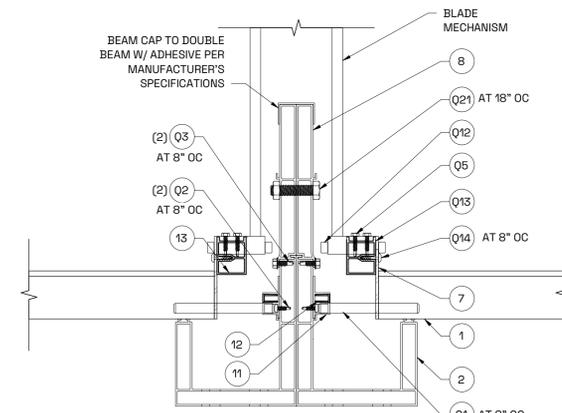
3" = 1'-0"



DOUBLE BEAM CORNER ASSEMBLY

11 DETAIL

3" = 1'-0"



DOUBLE BEAM ASSEMBLY

12 SECTION

3" = 1'-0"



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TIMAN WINDOW TREATMENTS, INC.
RAYMOND TATKO
 1751 E Hines Hill Rd, Hudson, OH 44236

ISSUANCE SCHEDULE

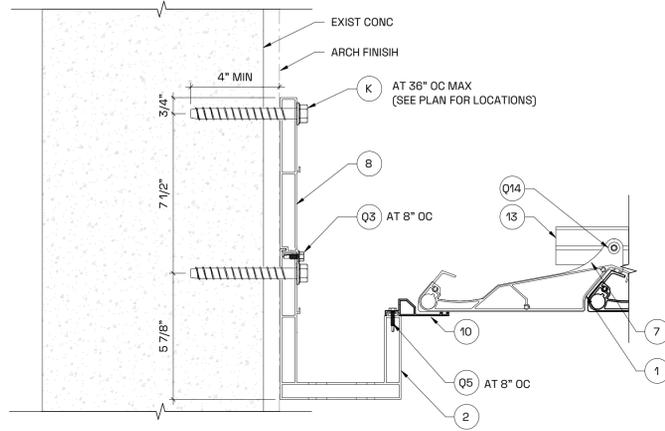
NO.	DESCRIPTION	DATE
0.	INITIAL ISSUANCE	07/19/2024

DETAILS CONT

PROJECT #	24577
DATE	07/19/2024
DWN	BZ
CHK	AEM

S7.2

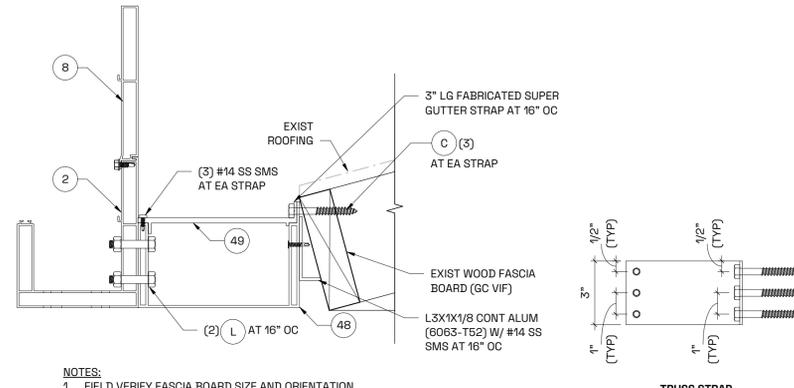
SCALE: AS INDICATED



MAIN BEAM CONNECTION TO HOST STRUCTURE

SECTION

1 3" = 1'-0"



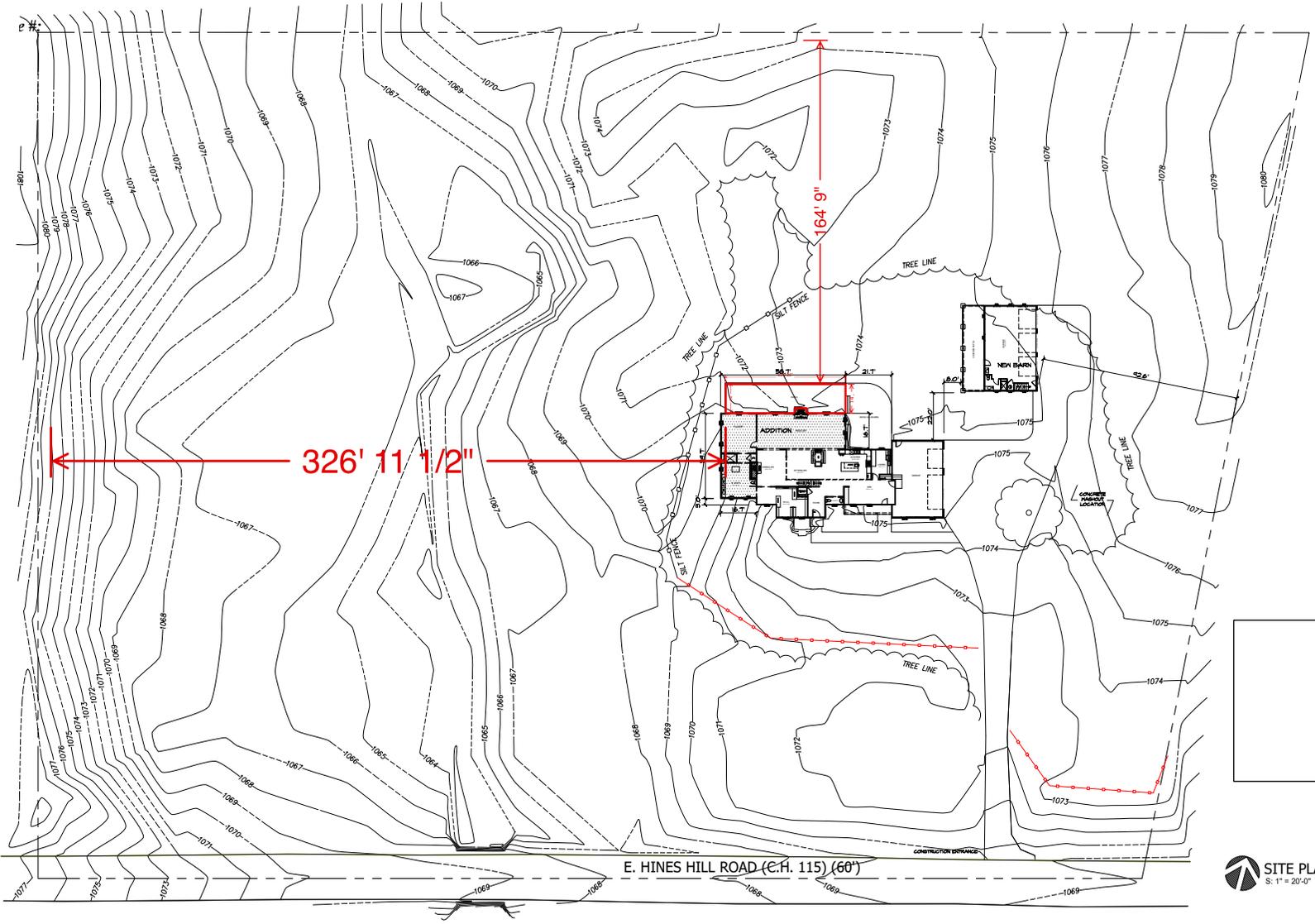
BEAM TO SUPER GUTTER CONNECTION

SECTION

2 3" = 1'-0"

NOTES:
 1. FIELD VERIFY FASCIA BOARD SIZE AND ORIENTATION.

TRUSS STRAP PLAN



SITE PLAN
S: 1" = 20'-0"

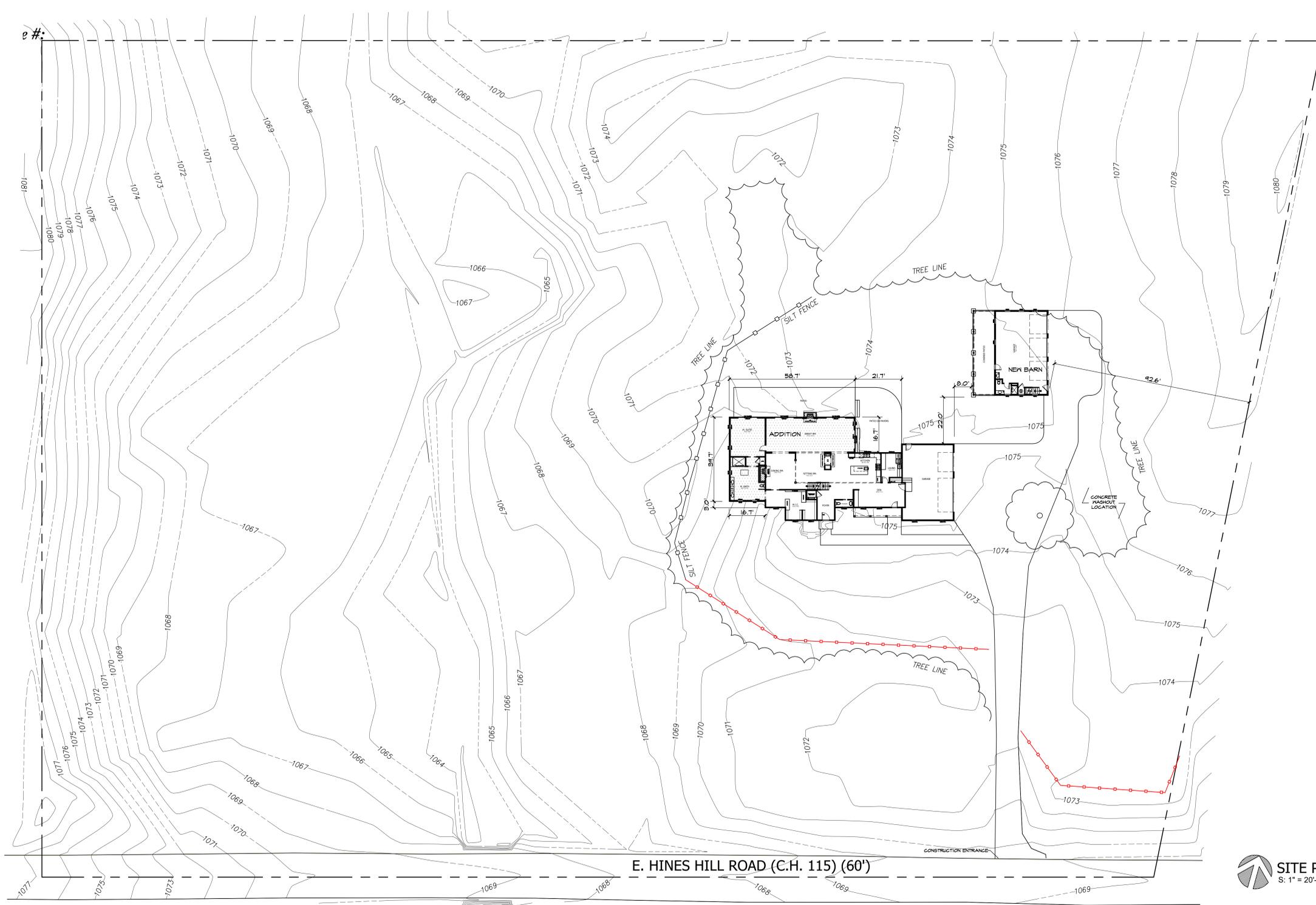
TATKO RESIDENCE
105 E. HINES HILL RD.
HEBRON, OH 43028
SITE PLAN

DATE: 11-12-21
2021-08-26 LAYOUT EXISTING
2021-09-02 ANHRI SUBMISSION
2021-11-09 ANHRI REVISION
2021-11-12 COVER SHEET

EXPERIENCED PROFESSIONAL ENGINEER
STATE OF OHIO LICENSE NO. 94127
J. KAPELA
105 E. HINES HILL RD.
HEBRON, OH 43028
PH: 614-885-5228
WWW.JKAPELA.COM

JOB # 2021-49

S101



SITE PLAN
S. 1" = 20'-0"

TATKO RESIDENCE
1751 E. HINES HILL RD.
HUDSON, OH 44236

SITE PLAN

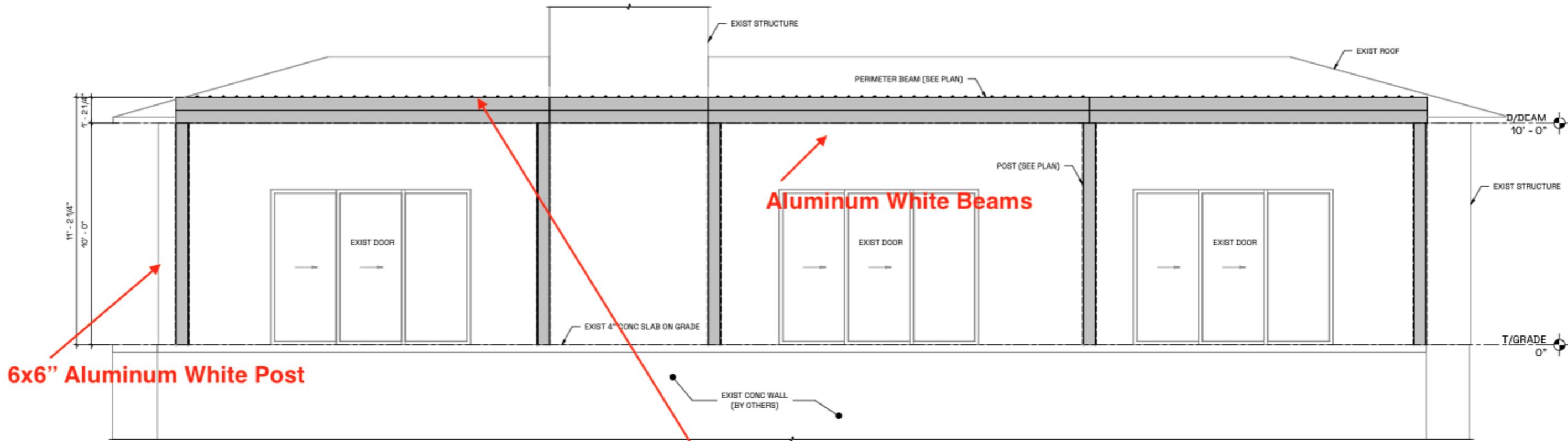
DATE:
2021-08-26 LAYOUT EXISTING
2021-11-02 AS-BUILT SUBMISSION
2021-11-09 AS-BUILT REVISION
2021-11-12 CONSTR. DOC.

11-12-21

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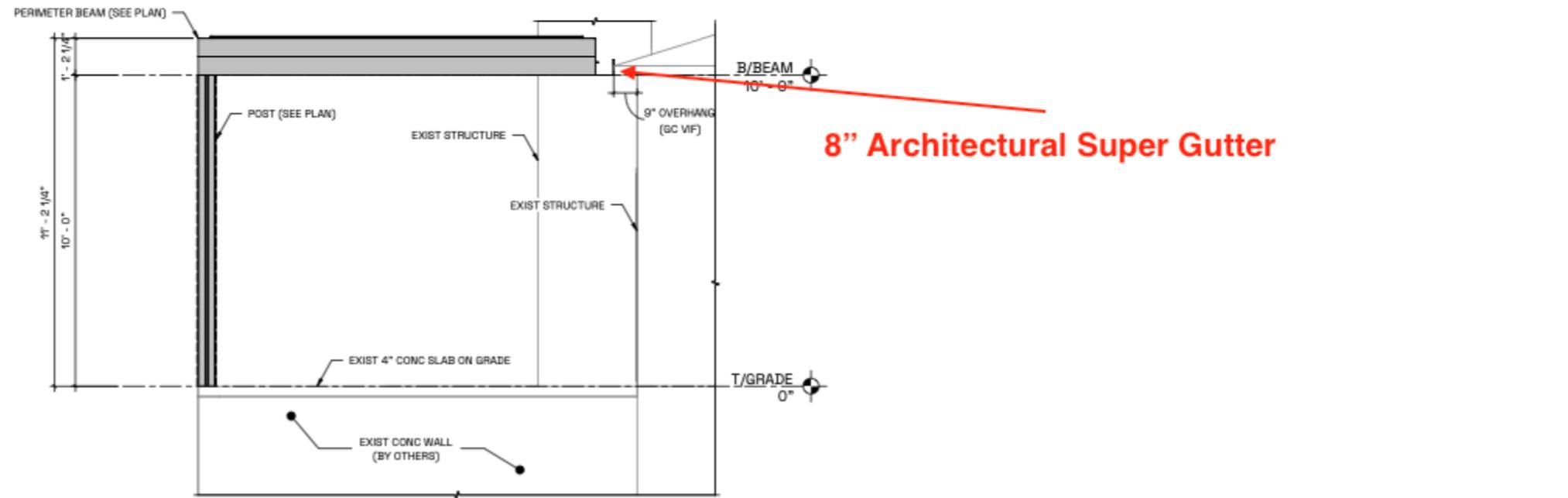
JOB # 2021-49

S101



White Aluminum Operable Roof Louver

1 FRONT ELEVATION
3/8" = 1'-0"













CALCULATION COVER SHEET

Calculations Prepared For:

TIMAN WINDOW TREATMENTS, INC
4533 WILLOW PARKWAY
CLEVELAND, OH

Project:

RAYMOND TATKO
1751 E HINES HILL RD
HUDSON, OH

Subject:

CANOPY CALCULATIONS

REFERENCE SEALED DRAWING BY BELOW-SIGNED ENGINEER FOR ALL NOTES AND DETAILS
INCORPORATED HEREIN

This item has been digitally signed and sealed by Andrew McCann P.E. On
07/19/24

Printed copies of this document are not considered signed and sealed and the
signature must be verified on any electronic copies.

Engineer's Seal Valid For Pages
1 Through 57

OH

07/19/24

ANDREW McCANN
PE 87255
Cert Auth 6511



Project # 24577

Timan Window Treatments, Inc Raymond Tatko

Wind Loading Criteria (ASCE 7-16)

Basic Wind Speed 110 MPH
 Wind Velocity (V_{asd}) 86 MPH
 Risk Category II
 Importance Factor 1.00
 Exposure Category C

ASCE 7-16
 ASD
 Residential

Snow Loading Criteria (ASCE 7-16)

Ground Snow Load 20 PSF
 Flat Roof Snow Load 25.00 PSF
 Snow Exposure Factor 1.00
 Snow Thermal Factor 1.20
 Snow Importance Factor 1.00

Live Loading Criteria (ASCE 7-16)

Roof Live Load 20 PSF

Dead Loading Criteria (ASCE 7-16)

Dead Load 5 PSF

Azenco

Seismic Load Criteria (ASCE 7-16)

Site Class D
 Occupancy Category II

Host Attached? Y ONE SIDE
 Host Supported? N

Mapped Spectral Response Accelerations:

S_s 0.136
 S₁ 0.049

Spectral Response Coefficients:

S_{DS} 0.145
 S_{D1} 0.078
 P 1.0
 SDC B
 TL 12

Load Combinations (ASCE 7-16)

Gravity D + (Lr or S or R)
 Uplift 0.6D + 0.7E

Project # 24577 - Raymond Tatko

**Timan Window Treatments, Inc
Raymond Tatko**

DESIGN CRITERIA:

Enter custom loads:

Vult = 110 mph
 Exposure: C
 Ground Snow Load: 20.00 psf
 Live Load: 20.00 psf
 Dead Load: 5.0 psf
 Wind Porosity: 50%
 Roof Type: Louvered

Type of project: Residential

These are the loads that this calculator will utilize:

Vult = 110 mph
 Exposure: C
 Ground Snow Load: 20.00 psf
 Design Live Load: 20.00 psf
 Design Dead Load: 5.00 psf
 Wind Porosity: 50%

Deflection criteria: L / 180

Critical positive grav comb. (+): 30.00 psf
 Critical negative uplift comb. (-): - 3.00 psf
 Critical lateral pressure (+): 16.76 psf

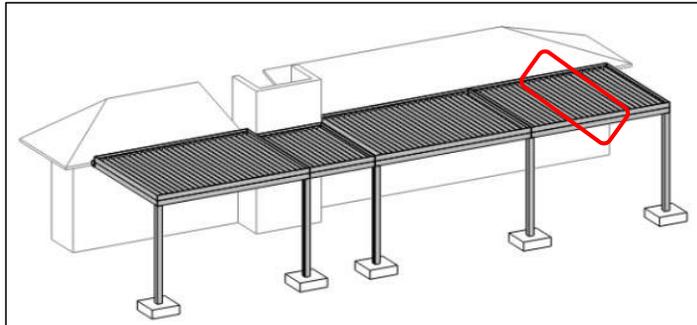
SYSTEM CONFIGURATION:

Louvers:

Overall Canopy Length: 55.9 ft
 Overall Canopy Width: 12.7 ft
 Roof Slope: 0.0 °

LOUVER BLADES CLOSED CHECK 6063-T6

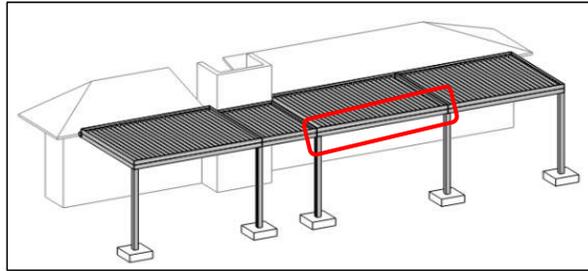
Sx	2.9166	in^3	Mmax	499.53	lb-ft		
Sy	1.26438	in^3	Stress	7.71	ksi	Stress Check:	85.0%
Length of Longest Louver Blade:	12 ft	8 in				Louver Length:	12.7 ft



Louver Support Beam

Check intermediate or edge? (Intermediate uses full louver blade tributary)

Edge Louver Beam Configuration



Louver Support Beam - Edge Condition Analysis

Support Spacing (Louver Beam Length):

Reinf:

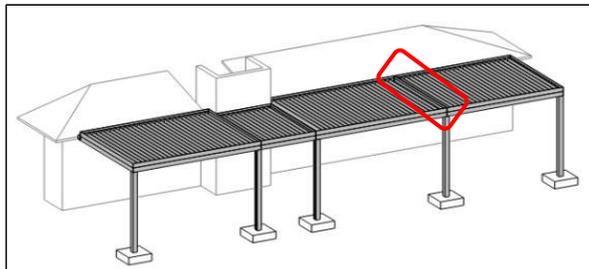
Single/Double/Triple/Quad:

Louver Beam Span:	17.0 ft
Louver Beam Trib:	6.3 ft
Shear at Ends:	1619 lb
Moment Check	59%
Deflection Check	39%

Main Beam

Check intermediate, edge, or none? (Intermediate doubles point loads from louver beams)

Intermediate Main Beam Configuration



Main Beam - Intermediate Condition Analysis

Post Spacing (Main Beam Length):

Reinf:

Single/Double/Triple/Quad:

Quantity of louver between a set of posts: (0 indicates Louver Beam line up directly over posts)

Assumed offset distance "a" of Louver Beam, measured from post (see diagram):

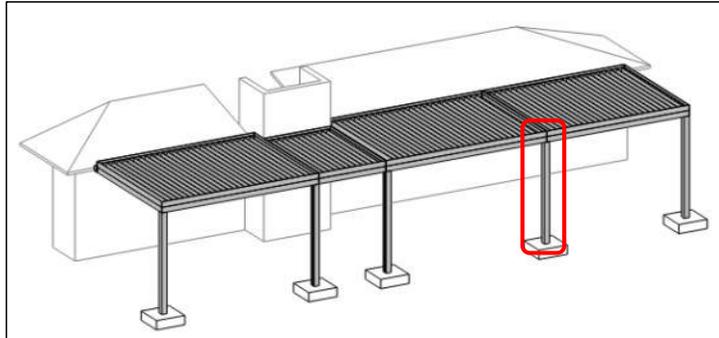
Louver beams line up directly on posts. No vertical load acting on main beam. Self-weight negligible.

Main Beam Span:	12.7 ft
Load from Louver Beam:	0 lb
Shear at Ends:	0 lb
Moment Check	0%
Deflection Check	0%

Support Posts

Check intermediate, edge, or none?

Edge Support Post Configuration



Mounting Height Above Grade: (Enter 0 for installations at ground level)
 Height of Posts:
 Attached to host? Total Mean Roof Height:
 Roof Eave Height=

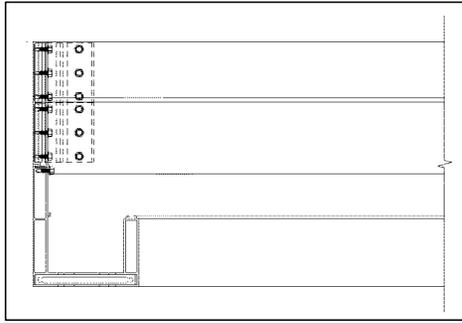
Support Posts - Edge Condition Analysis

Post Size: x x

X - AXIS Post Trib = ft
 Y - AXIS Post Trib = ft Host Attached
 FH1 = in, (side fascia height at HT1, normal to lateral windload)

ASD Method	Max. Moment/Axial/Shear	34%
	Moment/Axial Check:	34%
	Shear Check	4%
	Required Tension:	306 lb
	Required Compression:	3056 lb
	Required Shear:	462 lb
	Required Moment	1.7 kip-ft

Beam to Beam (Clip in Shear)



Qty **6** Anchor Qty at Connection per Beam

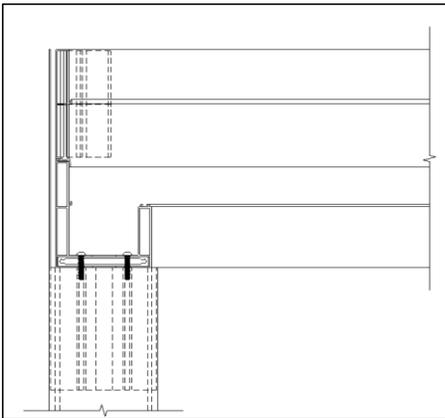
Required Tension:	0 lb
Required Shear:	1619 lb

(Analyzing 1/4-14 SMS, 316 SS, Steel Screw to 0.125" x 0.125" connecting parts thicknesses)

52%

OK, (6) anchors sufficient

Beam to Post Insert Connection



Qty **4** Anchor Qty at Connection

F 5148.59 lb $\Omega = 4$
 Fsu 60000.00
 Ats 0.34 in²
 n 20.00 threads/in
 LE 1.57 in
 Dsmin 0.31 in
 Enmax 0.28 in

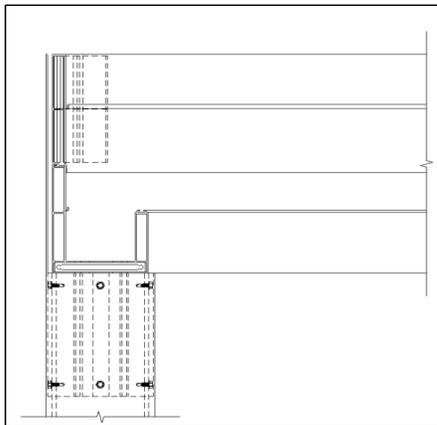
Required Tension:	306 lb
Capacity Shear:	20594 lb

Analyzing Beam to Post Insert with (4) M8x40 Machine screws

1%

OK

Post Insert to Post Connection



Qty **6** Anchor Qty at Connection

Required Tension:	0 lb
Required Shear:	462 lb

(Analyzing 1/4-14 SMS, 316 SS, Steel Screw to 0.125" x 0.125" connecting parts thicknesses)

15%

OK, (6) anchors sufficient

Foundation

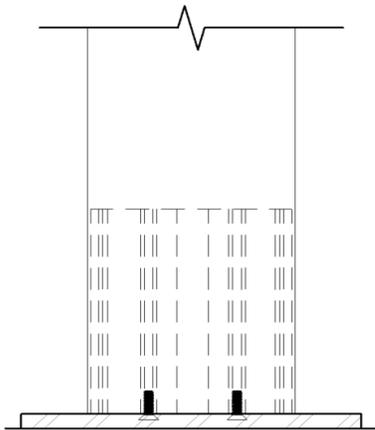
SEE SKYCIV			
Foundation Reactions (K,K-FT)			
	P	V	M
D	0.509	0.000	0.000
W	0.683	0.680	2.839
E	0.015	0.042	0.296
Lr	1.222	0.000	0.000
S	2.547	0.000	0.000

Baseplate 4-Bolt
 Plate Thickness: 0.5 in

Spacing: 8.625 in
 Width: 10.625 in

	SEE ANCHOR CALC
31%	Okay

Post Insert to Base Plate Connection



Qty 4 Anchor Qty at Connection

F	5148.59 lb	$\Omega = 4$
Fsu	60000.00	
Ats	0.34 in ²	
n	20.00 threads/in	
LE	1.57 in	
Dsmin	0.31 in	
Enmax	0.28 in	

Required Tension:	8608 lb
Capacity Shear:	20594 lb

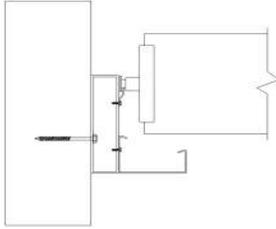
Analyzing Beam to Post Insert with (4)
 M8x40 Machine screws

42%

OK

Ledger Connection

Ledger Connection?



S=	16 inches	Anchor spacing along ledger
# of Screws=	3 pcs	Number of anchors per location
Required Tension:	9 lb	
Required Shear:	84 lb	

These are the ledger tributaries that this calculator will utilize:

$W =$	6.33 ft	Tributary Width
$RL =$	30.00 psf	Vert. Load on Tributary (LL, WL, DL, etc)
$FL =$	16.76 psf	Lateral Load on Tributary (Fascia Load)
$DL =$	0.00 lb/ft	Additional Beam Load (Weight or Service Loads)
$P_v =$	190.00 lb/ft	Total Vert. Loading on Beam
$P_f =$	19.73 lb/ft	Total Lat. Loading on Beam

3/8" diameter stainless steel lag screw with minimum 3" thread penetration, 3/4" edge distance, into southern yellow pine #2 wood (G = 0.55)

$V_{allow} =$	147 lb	Shear Capacity per Anchor
$T_{allow} =$	405 lb	Tension Capacity per Anchor

60%
Ledger Connection OK

Work Prepared For: Timan Window Treatments, Inc
 Project: 24577 - Raymond Tatko

DESIGN CRITERIA:

H = 10.00	ft, Mean Roof Height	ASCE: 7-16
Ø = 0.0 °	Roof Slope	Exposure: C
Vult = 110	mph, Wind Velocity (3-Second Gust)	Building Category: II
Kd = 0.85	Directionality Factor	
G = 0.85	Gust Effect Factor	Snow: Y
Kz = 0.85	Velocity Pressure Coefficient	Ground Snow Load: 20.00 psf
Kzt = 1	Topographic Factor	Design Snow Load: 25.00 psf
		Design Live Load: 20.00 psf
		Design Dead Load: 5.0 psf
Wind Flow: Clear		Wind Porosity: 50%
L = 55.88	ft, Overall Canopy Length	Method: ASD
W = 12.67	ft, Overall Canopy Width	Live Load Lr: 12.00 psf
a = 3.00 ft		Reduction Per Lo: 20.00 psf
		IBC R1: 0.6
		1607.13.2.1 R2: 1

LOADS ON COMPONENTS & CLADDING:

(Roof Decking and Decking Fasteners)

L1 = 12.67	ft, Effective Deck Panel Length	
W1 = 4.22 ft	Effective Deck Panel Width	
A = 53.48 ft ²	Effective Wind Area, L1*W1	<u>A > 4.0*a²</u>
CNp = 0.6	Positive Pressure Coefficient	
CNn = -0.5	Negative Pressure Coefficient	
qz = 11.18 psf	Velocity Pressure w/ Porosity	
Wlp = 6.71 psf	Positive Wind Load, = qz*G*CNp	
Wln = -5.36 psf	Negative Wind Load, = qz*G*CNn	
Grav = 30.00 psf	D + (Lr or S or R)	Critical positive DP
Uplift = -3.00 psf	0.6D + 0.7E	Critical negative DP

LOADS ON MAIN WIND FORCE RESISTING SYSTEM:

(Beams, Columns, Foundations)

<u>Wind Direction, γ = 0°</u>		<u>Wind Direction, γ = 180°</u>	
CNWa = 1.2	Cnw value, load case A	CNWa = 1.2	Cnw value, load case A
CNWb = -1.1	Cnw value, load case B	CNWb = -1.1	Cnw value, load case B
CNLa = 0.3	Cnl value, load case A	CNLa = 0.3	Cnl value, load case A
CNLb = -0.1	Cnl value, load case B	CNLb = -0.1	Cnl value, load case B
<u>Wind Direction, γ = 90°</u>			
CNa = -0.8	Cn value, load case A	CNb = 0.8	Cn value, load case B
CNp = 0.6	Critical Positive Pressure Coefficient		
CNn = -0.5	Critical Negative Pressure Coefficient		
Wlp = 6.71 psf	Critical Positive Wind Load, = qz*G*CNp		
Wln = -5.36 psf	Critical Negative Wind Load, = qz*G*CNn		
Grav = 30.00 psf	D + (Lr or S or R)		Critical positive DP
Uplift = -3.00 psf	0.6D + 0.7E		Critical negative DP

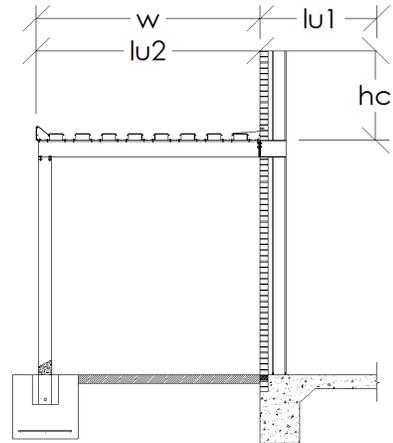
LOADS ON CANOPY FASCIA:

GCpn1 = 1.5	Combined Net Pressure Coefficient on windward fascia
GCpn1 = -1	Combined Net Pressure Coefficient on leeward fascia
WL = 16.76 psf	Average Wind Load on Fascia, qz*GCpn*.06

Work Prepared For: Timan Window Treatments, Inc
 Project: 24577 - Raymond Tatko

Snow Loads

$P_g = 20$ psf, Ground snow load
 $C_e = 1.0$ Exposure factor (Table 7-2)
 $C_t = 1.2$ Thermal factor (Table 7-3)
 $I_s = 1.0$ Importance factor (Table 7-4)
 $E_{vs} = 1.00^\circ$ Eave slope
 $S = 57.29$ Roof slope run for a rise of one
 $W = 15.00$ ft, Horizontal distance from eave to ridge
 $\gamma = 16.60$ pcf Snow density Eq. 7-3: $0.13(P_g) + 14 < 30$ psf
 $C_s = 1.00$ Slope factor at 1° (Figure 7-2)



Balanced Snow Loads

$P_f = 20.00$ psf Snow load on flat roofs (slope $< 5^\circ$): $P_f = \max[(I)(P_g), (0.7)(C_e)(C_t)(I)(P_g)]$
 $P_s = 20.00$ psf Sloped roof snow loads (slope $> 5^\circ$): $P_s = (C_s)(P_f)$

Drifts on Lower Roofs (Aerodynamic Shade)

$lu_1 = 20.00$ ft, Length of upper roof
 $lu_2 = 15.00$ ft Length of lower roof projection
 $hc = 10.00$ ft, Height from top of lower roof to top of eave

Drift snow required, $hc/h_b > 0.2$

$h_b = 1.20$ ft Height of balanced snow: $P_s / (\gamma)$
 $hd_1 = 1.23$ ft Height of snow drift (Fig 7-9): $0.43(lu)^{1/3}(P_g + 10)^{1/4} - 1.5$ (Leeward)
 $hd_2 = 0.74$ ft Height of snow drift (Fig 7-9): $0.43(lu)^{1/3}(P_g + 10)^{1/4} - 1.5$ (Windward)

ASCE 7-10/7-16 - Rain-On-Snow Surcharge (7.10)

Is P_g 20 PSF or less? YES
 5.00 psf Rain-On-Snow Surcharge
 $hd = 1.23$ ft Governing drift height
 $w = 4.93$ ft Governing drift width
 $hend = 0.00$ ft Drift height at edge of lower roof
 $pd = 10.22$ psf Surcharge load Uniform Distribution Over Drift Width
 3.36 psf Surcharge Load Distributed over Tributary Area

Unreducible Snow Load	Yes
Include Uniform Dist. Ice Load?	Yes
Include surcharge load?	Yes

SL = 25.00 psf Unreducible Roof Snow Load

Work Prepared For: Timan Window Treatments, Inc
 Project: 24577 - Raymond Tatko

Ice Load Due to Freezing Rain (per ASCE 7-16 - Chapter 10)

Accounting for Accumulating Ice on Louver Blades

$t_i =$	0.00 Nominal Ice Thickness (in.)	Louver (6" O.C.)
$K_{zt} =$	1.0 Topographic Factor	Depth (d) 6.000 in.
$Z =$	10.00 ft System Height	Width (bf) 1.866 in.
$I_i =$	1.00 Importance Factor	Thickness .065 in.
$I_d =$	56.00 Ice Density (56 pcf default)	Length 12.67 ft
	II Occupancy Category	

Per Table 10-1

$t_d =$	0.00 in, Design Ice Thickness	$t_d = t_i * I_i * f_z * (K_{zt})^{0.35}$
$W_i =$	0.00 psf Weight of Ice (for td)	$W_i = (td/12) * I_d$
$F_z =$	0.8875	$F_z = (Z/33)^{0.1}$

Ice Loading Ch 10.4

Louver Ice Loading

$D_c =$	6.32 in Circumscribing Diameter of Louver	$D_c = \sqrt{d^2 + bf^2}$
$A_i =$.00 in ² Area of Ice = $\pi t_d * (D_c + t_d)$	

$W_{i(Louver)} =$	0.00 plf Uniform Distributed Ice Load (Single Louver Blade)
	$W_i = (A_i/144) * I_d$

Louver Beam Ice Loading from Louver Blades

$W_{i(Louver)} =$	0.00 plf Distributed Ice Load on Louver Blade
$L =$	12.67 ft Length of Longest Louver Blade
$W_{i(Beam)} =$	0.0 plf Calculated Ice Load on Louver Beam
	$W_{i(Beam)} = W_{i(Louver)} * \text{Louver Length} * (1.866"/6")$
	(6" O.C. Louvers In Open Position)

$W_{i(Louver)} =$	0.00 plf Uniform Linear Ice Load (Louver Blade)
$W_{i(Beam)} =$	0.00 plf Uniform Linear Ice Load (Louver Beam)
	($W_{i(Beam)}$ doubled for intermediate Louver Beams)

Work Prepared For: Timan Window Treatments, Inc
 Project: 24577 - Raymond Tatko

Seismic Loads Criteria

$S_s = 0.136$ Max considered response acceleration for a period of 0.2 s
 $S_1 = 0.049$ Max response acceleration at period of 1 s

Height of Structure = 10.00 ft Attached to host structure? Y

Site Class D

$F_a = 1.6$ short period amplification factor
 $F_v = 2.4$ long period amplification factor
 $S_{MS} = 0.218$ modified spectral response acceleration at a period of 0.2 s $F_a * S_s$
 $S_{M1} = 0.118$ modified spectral response acceleration at a period of 1.0 s $F_v * S_1$

Spectral Response Acceleration Parameters

$S_{DS} = 0.145$ Design spectral response acceleration at a period of 0.2 s $(2/3) * S_{ms}$
 $S_{D1} = 0.078$ design spectral response acceleration at a period of 1.0 s $(2/3) * S_{M1}$

Structural Design Requirements

$T_a = 0.112$ Approximate fundamental period (s) $C_t * h_n^x$
 $T_L = 12.0$ Long Transition Period (s)
 $E_v = 0.102$ Vertical Seismic Loads (PSF)
 $R_p = 2.50$
 $a_p = 2.500$
 $l_p = 1.000$

$W_p = 509.31$ lbs Tributary Weight
 $F_p = 29.55$ lbs Seismic Design Force $0.4a_p * SDS * W_p / (R_p / l_p) * (1 + 2(z/h))$
 $F_{pMAX} = 118.21$ lbs
 $F_{pMIN} = 22.16$ lbs
 $P = 1$
 $\Omega = 2.00$
SERVICE = 0.7
 206.87 lb-ft Effective Seismic Moment $(H * F_p)$
 SDF OK? **OK**

Work Prepared For: Timan Window Treatments, Inc
 Project: 24577 - Raymond Tatko

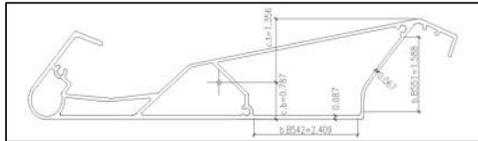
ALUMINUM DESIGN MANUAL (2020 EDITION)
Specifications for Aluminum Structures (Buildings)
Allowable Stress Design

LOUVER BLADE CHECK

Design Check of Azenco Louver Blade 0.087"x0.087"x2.1472"/1.0406" 6063-T6 Aluminum Tube
 Per 2020 Aluminum Design Manual

Alloy: 6063 Temper: T6 Critically Welded: N

MEMBER PROPERTIES



Flange width	9.963"
Flange thickness	0.087"
Web height	2.147"
Web thickness	0.087"
Moment of inertia about axis parallel to flange	1.04 in ⁴
Moment of inertia about axis parallel to web	17.38 in ⁴
Section modulus compression x-axis	0.78 in ³
Section modulus tension x-axis	1.31 in ³
Section modulus tension y-axis	3.16 in ³
Section modulus compression y-axis	3.95 in ³
Radius of gyration about centroidal axis parallel to flange	0.71 in
Radius of gyration about centroidal axis parallel to web	2.92 in
Torsion constant	1.26 in ⁴
Cross sectional area of member	2.04 in ²
Plastic section modulus	1.30 in ³
Warping constant	5.73 in ⁶

MEMBER SPANS

Unsupported member length (between supports)	$L =$	12.67 ft
Unbraced length for bending (between bracing against side-sway)	$L_b =$	12.67 ft
Effective length factor	$k =$	1.0
Tensile ultimate strength	$F_{tu} =$	30 ksi
Tensile yield strength	$F_{ty} =$	25 ksi
Compressive yield strength	$F_{cy} =$	25 ksi
Shear ultimate strength	$F_{su} =$	18 ksi
Shear yield strength	$F_{sy} =$	15 ksi
Compressive modulus of elasticity	$E =$	10,100 ksi

BUCKLING CONSTANTS

Compression in columns & beam flanges (Intercept)	$B_c =$	27.64 ksi
Compression in columns & beam flanges (Slope)	$D_c =$	0.14 ksi
Compression in columns & beam flanges (Intersection)	$C_c =$	78.38 ksi
Compression in flat plates (Intercept)	$B_p =$	31.39 ksi
Compression in flat plates (Slope)	$D_p =$	0.17 ksi
Compression in flat plates (Intersection)	$C_p =$	73.55 ksi
Compressive bending stress in solid rectangular bars (Intercept)	$B_{br} =$	46.12 ksi
Compressive bending stress in solid rectangular bars (Slope)	$D_{br} =$	0.38 ksi
Shear stress in flat plates (Intercept)	$B_s =$	18.98 ksi
Shear stress in flat plates (Slope)	$D_s =$	0.08 ksi
Shear stress in flat plates (Intersection)	$C_s =$	94.57 ksi
Ultimate strength coefficient of flat plates in compression (slenderness limit λ_2)	$k1c =$	0.35
Ultimate strength coefficient of flat plates in compression (stress for slenderness $> \lambda_2$)	$k2c =$	2.27
Ultimate strength of flat plates in bending (slenderness limit λ_2)	$k1b =$	0.50
Ultimate strength of flat plates in bending (stress for slenderness $> \lambda_2$)	$k2b =$	2.04
Tension coefficient	$kt =$	1.0

D.2 Axial Tension

Tensile Yielding - Unwelded Members	$[F_{ty}]$	$F_{ty_n} =$	25.00 ksi
		$\Omega =$	1.65
		$F_{ty_n}/\Omega =$	15.15 ksi
Tensile Rupture - Unwelded Members	$[F_{tu}]$	$F_{tu_n} =$	30.00 ksi
		$\Omega =$	1.95
		$F_{tu_n}/\Omega =$	15.38 ksi

AXIAL COMPRESSION MEMBERS

E.2 Compression Member Buckling

Lower slenderness limit	$\lambda_1 =$	18.23	
Upper slenderness limit	$\lambda_2 =$	78.38	
Slenderness	$\lambda(max) =$	212.97	$\geq \lambda_2$
$[0.85\pi^2E/\lambda^2]$	$F_{c_n} =$	1.87 ksi	
	$\Omega =$	1.65	
	$F_{c_n}/\Omega =$	1.13 ksi	

E.3 Local Buckling

For column elements in uniform compression subject to local buckling, the uniform compressive strength is addressed in Section B.5.4 calculated below.

B.5.4.2 - Flat elements supported on both edges (Flange)

B.5.4.2 - Flat elements supported on both edges (Web)

E.4 Buckling Interaction

Per Table B.5.1

$[\pi^2 E / (1.6 \cdot b / t b)^2]$	$F_e(\text{flange}) =$	3.08 ksi
$[F_{c_n}]$	$F_{c_n} =$	1.87 ksi
$F_e(\text{flange}) > F_{c_n}$ (E.2 Member Buckling)	$\Omega =$	1.65
	$F_{c_n} / \Omega =$	1.13 ksi
$[\pi^2 E / (1.6 \cdot h / t h)^2]$	$F_e(\text{web}) =$	75.69 ksi
$[F_{c_n}]$	$F_{c_n} =$	1.87 ksi
$F_e(\text{web}) > F_{c_n}$ (E.2 Member Buckling)	$\Omega =$	1.65
	$F_{c_n} / \Omega =$	1.13 ksi

FLEXURAL MEMBERS

F.2 Yielding and Rupture

Nominal flexural strength for yielding and rupture

Limit State of Yielding

$[S_c \cdot F_{cy}]$	$M_{np} =$	19.43 k-in
$[M_{np} / S_x]$	$F_{b_n} =$	14.97 ksi
	$\Omega =$	1.65
	$F_{b_n} / \Omega =$	9.07 ksi

Limit State of Rupture

$[Z \cdot F_{tu} / k_t]$	$M_{nu} =$	38.93 k-in
$[M_{nu} / Z]$	$F_{b_n} =$	30.00 ksi
	$\Omega =$	1.95
	$F_{b_n} / \Omega =$	15.38 ksi

F.4 Lateral-Torsional Buckling

Unsymmetric shape subject to lateral-torsional buckling

Slenderness for closed shape about the bending axis $\lambda_{F.4.2.5} =$ 11.55

Maximum slenderness $\lambda(\text{max}) =$ 11.55 < C_c

Nominal flexural strength - lateral-torsional buckling

$[M_{np}(1 - (\lambda / C_c)) + (\pi^2 \cdot E \cdot \lambda^2 \cdot S_x / C_c^3)]$	$M_{nmb} =$	18.42 k-in
$[M_{nmb} / S_x]$	$F_{b_n} =$	23.71 ksi
	$\Omega =$	1.65
	$F_{b_n} / \Omega =$	14.37 ksi

UNIFORM COMPRESSION ELEMENTS

B.5.4.2 Flat Elements Supported on Both Edges - Web & Flange

Uniform compression strength, flat elements supported on both edges

Lower slenderness limit	$\lambda_1 =$	22.8	
Upper slenderness limit	$\lambda_2 =$	39.2	
Flange Slenderness	$b / t b =$	112.52	$\geq \lambda_2$
Web Slenderness	$h / t h =$	22.68	$\leq \lambda_1$
$[k_2 c \cdot \sqrt{(B_p \cdot E) / (1.6 \cdot b / t b)}]$	$F_{c_n1} =$	7.10 ksi	
	$\Omega =$	1.65	
	$F_{c_n1} / \Omega =$	4.30 ksi	
$[F_{cy}]$	$F_{c_n2} =$	25.00 ksi	
	$\Omega =$	1.65	
	$F_{c_n2} / \Omega =$	15.15 ksi	

FLEXURAL COMPRESSION ELEMENTS

B.5.5.1 Flat Elements Supported on Both Edges - Web

Flexural compression strength, flat elements supported on both edges

Lower slenderness limit	$\lambda 1 =$	34.73	
Upper slenderness limit	$\lambda 2 =$	92.95	
Slenderness	$h/th =$	22.68	$\leq \lambda 1$
$[1.5*Fcy]$	$Fb_n =$	37.50 ksi	
	$\Omega =$	1.65	
	$Fb_n/\Omega =$	22.73 ksi	

SHEAR

G.2 Shear Supported on Both Edges - Web

Members with flat elements supported on both edges

Lower slenderness limit	$\lambda 1 =$	38.73	
Upper slenderness limit	$\lambda 2 =$	75.65	
Slenderness	$h/th =$	22.68	$\leq \lambda 1$
$[Fsy]$	$Fv_n =$	15.00 ksi	
	$\Omega =$	1.65	
	$Fv_n/\Omega =$	9.09 ksi	

ALLOWABLE STRESSES

Allowable bending stress	$Fb =$	9.07 ksi
Allowable axial stress, compression	$Fac =$	1.13 ksi
Allowable shear stress; webs	$Fv =$	9.09 ksi

Elastic buckling stress	$Fe =$	1.13 ksi
Weighted average allowable compressive stress (per Section E.3.1)	$Fao =$	6.12 ksi

Mmax	499.53	lb-ft
Stress	7.71	ksi
Fb	9.07 ksi	

85.032%

Work Prepared For: Timan Window Treatments, Inc
 Project: 24577 - Raymond Tatko
 Detail/Member: Louver Beam

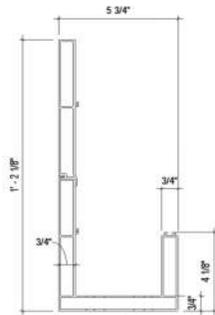
ALUMINUM DESIGN MANUAL (2020 EDITION)
Specifications for Aluminum Structures (Buildings)
Allowable Stress Design

Design Check of 5.75"x14.125"x0.125" 6005A-T6 Aluminum Tube

Per 2020 Aluminum Design Manual

Alloy: 6005A Temper: T6 Critically Welded: N

MEMBER PROPERTIES



Flange width	$b =$	5.750"
Flange thickness	$tb =$	0.125"
Web height	$h =$	14.125"
Web thickness	$th =$	0.125"
Moment of inertia about axis parallel to flange	$I_x =$	80.77 in ⁴
Moment of inertia about axis parallel to web	$I_y =$	17.42 in ⁴
Section modulus compression x-axis	$S_{xc} =$	8.94 in ³
Section modulus tension x-axis	$S_{xt} =$	15.88 in ³
Section modulus tension y-axis	$S_{yt} =$	4.34 in ³
Section modulus compression y-axis	$S_{yc} =$	10.03 in ³
Radius of gyration about centroidal axis parallel to flange	$r_x =$	4.36 in
Radius of gyration about centroidal axis parallel to web	$r_y =$	2.02 in
Torsion constant	$J =$	1.90 in ⁴
Cross sectional area of member	$A =$	4.25 in ²
Plastic section modulus	$Z =$	16.60 in ³
Warping constant	$C_w =$	5.75 in ⁶

MEMBER SPANS

Unsupported member length (between supports)	$L =$	17.04 ft
Unbraced length for bending (between bracing against side-sway)	$L_b =$	17.04 ft
Effective length factor	$k =$	1.0

MATERIAL PROPERTIES

Tensile ultimate strength	$F_{tu} =$	38 ksi
Tensile yield strength	$F_{ty} =$	35 ksi
Compressive yield strength	$F_{cy} =$	35 ksi
Shear ultimate strength	$F_{su} =$	23 ksi
Shear yield strength	$F_{sy} =$	21 ksi
Compressive modulus of elasticity	$E =$	10,100 ksi

BUCKLING CONSTANTS

Compression in columns & beam flanges (Intercept)	$B_c =$	39.37 ksi
Compression in columns & beam flanges (Slope)	$D_c =$	0.25 ksi
Compression in columns & beam flanges (Intersection)	$C_c =$	65.67 ksi
Compression in flat plates (Intercept)	$B_p =$	45.00 ksi
Compression in flat plates (Slope)	$D_p =$	0.30 ksi
Compression in flat plates (Intersection)	$C_p =$	61.42 ksi
Compressive bending stress in solid rectangular bars (Intercept)	$B_{br} =$	66.82 ksi
Compressive bending stress in solid rectangular bars (Slope)	$D_{br} =$	0.67 ksi
Shear stress in flat plates (Intercept)	$B_s =$	27.24 ksi
Shear stress in flat plates (Slope)	$D_s =$	0.14 ksi
Shear stress in flat plates (Intersection)	$C_s =$	78.95 ksi
Ultimate strength coefficient of flat plates in compression (slenderness limit λ_2)	$k_{1c} =$	0.35
Ultimate strength coefficient of flat plates in compression (stress for slenderness $> \lambda_2$)	$k_{2c} =$	2.27
Ultimate strength of flat plates in bending (slenderness limit λ_2)	$k_{1b} =$	0.50
Ultimate strength of flat plates in bending (stress for slenderness $> \lambda_2$)	$k_{2b} =$	2.04
Tension coefficient	$kt =$	1.0

D.2 Axial Tension

Tensile Yielding - Unwelded Members	$[F_{ty}]$	$F_{ty_n} =$	35.00 ksi
		$\Omega =$	1.65
		$F_{ty_n}/\Omega =$	21.21 ksi
Tensile Rupture - Unwelded Members	$[F_{tu}/kt]$	$F_{tu_n} =$	38.00 ksi
		$\Omega =$	1.95
		$F_{tu_n}/\Omega t =$	19.49 ksi

AXIAL COMPRESSION MEMBERS

E.2 Compression Member Buckling

Axial, gross section subject to buckling

Lower slenderness limit	$\lambda 1 =$	17.76	
Upper slenderness limit	$\lambda 2 =$	65.67	
Slenderness	$\lambda(max) =$	101.03	$\geq \lambda 2$
$[0.85\pi^2 E/\lambda^2]$	$F_{c_n} =$	8.30 ksi	
	$\Omega =$	1.65	
	$F_{c_n}/\Omega =$	5.03 ksi	

E.3 Local Buckling

For column elements in uniform compression subject to local buckling, the uniform compressive strength is addressed in Section B.5.4 calculated below.

B.5.4.2 - Flat elements supported on both edges (Flange)

B.5.4.2 - Flat elements supported on both edges (Web)

E.4 Buckling Interaction

Per Table B.5.1

$[\pi^2 E / (1.6^2 b/tb)^2]$	$F_e(flange) =$	20.11 ksi
$[F_{c_n}]$	$F_{c_n} =$	8.30 ksi
$F_e(flange) > F_{c_n}$ (E.2 Member Buckling)	$\Omega =$	1.65
	$F_{c_n}/\Omega =$	5.03 ksi
$[\pi^2 E / (1.6^2 h/th)^2]$	$F_e(web) =$	3.16 ksi
$[0.85\pi^2 E/\lambda(max)^2]^{1/3} * [F_e^{2/3}]$	$F_{c_n} =$	4.36 ksi
$F_e(web) < F_{c_n}$ (E.2 Member Buckling)	$\Omega =$	1.65
	$F_{c_n}/\Omega =$	2.64 ksi

FLEXURAL MEMBERS

F.2 Yielding and Rupture

Nominal flexural strength for yielding and rupture

Limit State of Yielding	$M_{np} =$	312.73 k-in
$[1.5 S_c F_{cy}]$	$F_{b_n} =$	35.00 ksi
$[M_{np}/S_x]$	$\Omega =$	1.65
	$F_{b_n}/\Omega =$	21.21 ksi
Limit State of Rupture	$M_{nu} =$	839.60 k-in
$[Z^* F_{tu}/kt]$	$F_{b_n} =$	38.00 ksi
$[M_{nu}/Z]$	$\Omega =$	1.95
	$F_{b_n}/\Omega =$	19.49 ksi

F.4 Lateral-Torsional Buckling

Unsymmetric shape subject to lateral-torsional buckling

Slenderness for shapes any shape about the bending axis

Maximum slenderness $\lambda F.4.2.5 = 44.83$

Nominal flexural strength - lateral-torsional buckling

$[M_{np}(1-(\lambda/C_c)) + (\pi^2 E \lambda^2 S_x / C_c^3)]$	$M_{nmb} =$	240.22 k-in
$[M_{nmb}/S_x]$	$F_{b_n} =$	26.88 ksi
	$\Omega =$	1.65
	$F_{b_n}/\Omega =$	16.29 ksi

UNIFORM COMPRESSION ELEMENTS

B.5.4.2 Flat Elements Supported on Both Edges - Web & Flange

Uniform compression strength, flat elements supported on both edges

Lower slenderness limit	$\lambda 1 =$	20.8	
Upper slenderness limit	$\lambda 2 =$	32.8	
Flange Slenderness	$b/tb =$	44.0	$\geq \lambda 2$
Web Slenderness	$h/th =$	111.0	$\geq \lambda 2$
$[k 2 c^2 (B_p E) / (1.6^2 b/tb)]$	$F_{c_n1} =$	21.74 ksi	
	$\Omega =$	1.65	
	$F_{c_n1}/\Omega =$	13.17 ksi	
$[k 2 c^2 (B_p E) / (1.6^2 h/th)]$	$F_{c_n2} =$	8.62 ksi	
	$\Omega =$	1.65	
	$F_{c_n2}/\Omega =$	5.22 ksi	

FLEXURAL COMPRESSION ELEMENTS

B.5.5.1 Flat Elements Supported on Both Edges - Web

Flexural compression strength, flat elements supported on both edges

Lower slenderness limit	$\lambda_1 =$	33.10	
Upper slenderness limit	$\lambda_2 =$	77.22	
Slenderness	$h/th =$	111.00	$\geq \lambda_2$
$[k2b*SQRT(Bbr*E)/(m*h/th)]$	$Fb_n =$	23.23 ksi	
	$\Omega =$	1.65	
	$Fb_n/\Omega =$	14.08 ksi	

SHEAR

G.2 Shear Supported on Both Edges - Web

Members with flat elements supported on both edges

Lower slenderness limit	$\lambda_1 =$	35.29	
Upper slenderness limit	$\lambda_2 =$	63.16	
Slenderness	$h/th =$	111.00	$\geq \lambda_2$
$[\pi^2E/(1.25*h/th)^2]$	$Fv_n =$	5.18 ksi	
	$\Omega =$	1.65	
	$Fv_n/\Omega =$	3.14 ksi	

ALLOWABLE STRESSES

Allowable bending stress	$Fb =$	16.29 ksi
Allowable axial stress, compression	$Fac =$	3.32 ksi
Allowable shear stress; webs	$Fv =$	3.14 ksi

Elastic buckling stress	$Fe =$	5.01 ksi
Weighted average allowable compressive stress (per Section E.3.1)	$Fao =$	7.48 ksi

MEMBER LOADING

Bending Moments

Bending moment developed in member	$Mz =$	6.9 kip-ft	
Bending stress developed in member	$fb =$	9.26 ksi	
Allowable bending stress of member	$Fb =$	16.29 ksi	< 1.0

Axial Loads

Axial load developed in member	$Fx =$	0 lb	
Axial stress developed in member	$fa =$	0.00 ksi	
Allowable compressive axial stress of member	$Fac =$	3.32 ksi	< 1.0

Shear Loads

Shear load developed in member	$Vz =$	1,619 lb	
Shear stress developed in member	$fv =$	0.47 ksi	
Allowable shear stress of member webs	$Fv =$	3.14 ksi	< 1.0

Interaction Equations

	$\sqrt{[(fb/Fb)^2 + (fv/Fv)^2]} =$	0.59	< 1.0
Eq H.1-1	$fa/Fa + fb/Fb =$	0.00	< 1.0
Eq H.3-2	$fa/Fa + (fb/Fb)^2 + (fv/Fv)^2 =$	0.00	< 1.0

CONFIGURATION AND MOMENT TABULATION TOOLS

# of beam=	1	Support Type	Beam =	Simple
		Beam Length	L =	17.04 ft
		Tributary Width	W =	6.33 ft
		Load on Tributary (LL, WL, DL, etc)	RL =	30.00 psf
		Additional Beam Load (Weight or Service Loads)	DL =	0.00 lb/ft
		Total Loading on Beam	w =	190.00 lb/ft
		Shear Loading at End of Beam	Vy =	1619 lbs
		CALCULATED MOMENT	Mmax =	6.9 kip-ft

Deflection Check

	Support =	Simple
	Deflection Limit =	L / 180
	w =	190.00 lb/ft
ALLOWABLE DEFLECTION	$\Delta_{Allow} =$	1.14 in
MAXIMUM DEFLECTION	$\Delta_{Max} =$	0.44 in

Simple Max Deflection = $5wl^4/384EI$
OK, Allowable Deflection Sufficient

Work Prepared For: Timan Window Treatments, Inc
 Project: 24577 - Raymond Tatko
 Detail/Member: Main Beam

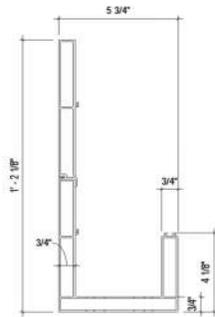
ALUMINUM DESIGN MANUAL (2020 EDITION)
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Allowable Stress Design

Design Check of 5.75"x14.125"x0.125"/0.125" 6005A-T6 Aluminum Tube

Per 2020 Aluminum Design Manual

Alloy: 6005A Temper: T6 Critically Welded: N

MEMBER PROPERTIES



Flange width	$b =$	5.750"
Flange thickness	$tb =$	0.125"
Web height	$h =$	14.125"
Web thickness	$th =$	0.125"
Moment of inertia about axis parallel to flange	$I_x =$	80.77 in ⁴
Moment of inertia about axis parallel to web	$I_y =$	17.42 in ⁴
Section modulus compression x-axis	$S_{xc} =$	8.94 in ³
Section modulus tension x-axis	$S_{xt} =$	15.88 in ³
Section modulus tension y-axis	$S_{yt} =$	4.34 in ³
Section modulus compression y-axis	$S_{yc} =$	10.03 in ³
Radius of gyration about centroidal axis parallel to flange	$r_x =$	4.36 in
Radius of gyration about centroidal axis parallel to web	$r_y =$	2.02 in
Torsion constant	$J =$	1.90 in ⁴
Cross sectional area of member	$A =$	4.25 in ²
Plastic section modulus	$Z =$	16.60 in ³
Warping constant	$C_w =$	5.75 in ⁶

MEMBER SPANS

Unsupported member length (between supports)	$L =$	12.67 ft
Unbraced length for bending (between bracing against side-sway)	$L_b =$	12.67 ft
Effective length factor	$k =$	1.0

MATERIAL PROPERTIES

Tensile ultimate strength	$F_{tu} =$	38 ksi
Tensile yield strength	$F_{ty} =$	35 ksi
Compressive yield strength	$F_{cy} =$	35 ksi
Shear ultimate strength	$F_{su} =$	23 ksi
Shear yield strength	$F_{sy} =$	21 ksi
Compressive modulus of elasticity	$E =$	10,100 ksi

BUCKLING CONSTANTS

Compression in columns & beam flanges (Intercept)	$B_c =$	39.37 ksi
Compression in columns & beam flanges (Slope)	$D_c =$	0.25 ksi
Compression in columns & beam flanges (Intersection)	$C_c =$	65.67 ksi
Compression in flat plates (Intercept)	$B_p =$	45.00 ksi
Compression in flat plates (Slope)	$D_p =$	0.30 ksi
Compression in flat plates (Intersection)	$C_p =$	61.42 ksi
Compressive bending stress in solid rectangular bars (Intercept)	$B_{br} =$	66.82 ksi
Compressive bending stress in solid rectangular bars (Slope)	$D_{br} =$	0.67 ksi
Shear stress in flat plates (Intercept)	$B_s =$	27.24 ksi
Shear stress in flat plates (Slope)	$D_s =$	0.14 ksi
Shear stress in flat plates (Intersection)	$C_s =$	78.95 ksi
Ultimate strength coefficient of flat plates in compression (slenderness limit λ_2)	$k_{1c} =$	0.35
Ultimate strength coefficient of flat plates in compression (stress for slenderness $> \lambda_2$)	$k_{2c} =$	2.27
Ultimate strength of flat plates in bending (slenderness limit λ_2)	$k_{1b} =$	0.50
Ultimate strength of flat plates in bending (stress for slenderness $> \lambda_2$)	$k_{2b} =$	2.04
Tension coefficient	$kt =$	1.0

D.2 Axial Tension

Tensile Yielding - Unwelded Members	$[F_{ty}]$	$F_{ty_n} =$	35.00 ksi
		$\Omega =$	1.65
		$F_{ty_n}/\Omega =$	21.21 ksi
Tensile Rupture - Unwelded Members	$[F_{tu}/kt]$	$F_{tu_n} =$	38.00 ksi
		$\Omega =$	1.95
		$F_{tu_n}/\Omega t =$	19.49 ksi

AXIAL COMPRESSION MEMBERS

E.2 Compression Member Buckling

Axial, gross section subject to buckling

Lower slenderness limit	$\lambda 1 =$	17.76	
Upper slenderness limit	$\lambda 2 =$	65.67	
Slenderness	$\lambda(max) =$	75.09	$\geq \lambda 2$
$[0.85\pi^2 E/\lambda^2]$	$F_{c_n} =$	15.03 ksi	
	$\Omega =$	1.65	
	$F_{c_n}/\Omega =$	9.11 ksi	

E.3 Local Buckling

For column elements in uniform compression subject to local buckling, the uniform compressive strength is addressed in Section B.5.4 calculated below.

B.5.4.2 - Flat elements supported on both edges (Flange)

B.5.4.2 - Flat elements supported on both edges (Web)

E.4 Buckling Interaction

Per Table B.5.1

$[\pi^2 E / (1.6^2 b/tb)^2]$	$F_e(flange) =$	20.11 ksi
$[F_{c_n}]$	$F_{c_n} =$	15.03 ksi
$F_e(flange) > F_{c_n}$ (E.2 Member Buckling)	$\Omega =$	1.65
	$F_{c_n}/\Omega =$	9.11 ksi
$[\pi^2 E / (1.6^2 h/th)^2]$	$F_e(web) =$	3.16 ksi
$[0.85\pi^2 E/\lambda(max)^2]^{1/3} * [F_e^{2/3}]$	$F_{c_n} =$	5.31 ksi
$F_e(web) < F_{c_n}$ (E.2 Member Buckling)	$\Omega =$	1.65
	$F_{c_n}/\Omega =$	3.22 ksi

FLEXURAL MEMBERS

F.2 Yielding and Rupture

Nominal flexural strength for yielding and rupture

Limit State of Yielding	$M_{np} =$	312.73 k-in
$[1.5^*S_c^*F_{cy}]$	$F_{b_n} =$	35.00 ksi
$[M_{np}/S_x]$	$\Omega =$	1.65
	$F_{b_n}/\Omega =$	21.21 ksi
Limit State of Rupture	$M_{nu} =$	839.60 k-in
$[Z^*F_{tu}/kt]$	$F_{b_n} =$	38.00 ksi
$[M_{nu}/Z]$	$\Omega =$	1.95
	$F_{b_n}/\Omega =$	19.49 ksi

F.4 Lateral-Torsional Buckling

Unsymmetric shape subject to lateral-torsional buckling

Slenderness for shapes any shape about the bending axis	$\lambda F.4.2.5 =$	40.04	
Maximum slenderness	$\lambda(max) =$	40.04	$< C_c$

Nominal flexural strength - lateral-torsional buckling

$[M_{np}(1-(\lambda/C_c)) + (\pi^2 E^* \lambda^2 S_x / C_c^3)]$	$M_{nmb} =$	247.98 k-in
$[M_{nmb}/S_x]$	$F_{b_n} =$	27.75 ksi
	$\Omega =$	1.65
	$F_{b_n}/\Omega =$	16.82 ksi

UNIFORM COMPRESSION ELEMENTS

B.5.4.2 Flat Elements Supported on Both Edges - Web & Flange

Uniform compression strength, flat elements supported on both edges

Lower slenderness limit	$\lambda 1 =$	20.8	
Upper slenderness limit	$\lambda 2 =$	32.8	
Flange Slenderness	$b/tb =$	44.0	$\geq \lambda 2$
Web Slenderness	$h/th =$	111.0	$\geq \lambda 2$
$[k2c^*\sqrt{(Bp^*E)/(1.6^*b/tb)}]$	$F_{c_n1} =$	21.74 ksi	
	$\Omega =$	1.65	
	$F_{c_n1}/\Omega =$	13.17 ksi	
$[k2c^*\sqrt{(Bp^*E)/(1.6^*h/th)}]$	$F_{c_n2} =$	8.62 ksi	
	$\Omega =$	1.65	
	$F_{c_n2}/\Omega =$	5.22 ksi	

FLEXURAL COMPRESSION ELEMENTS

B.5.5.1 Flat Elements Supported on Both Edges - Web

Flexural compression strength, flat elements supported on both edges

Lower slenderness limit	$\lambda 1 =$	33.10	
Upper slenderness limit	$\lambda 2 =$	77.22	
Slenderness	$h/th =$	111.00	$\geq \lambda 2$
$[k2b*SQRT(Bbr*E)/(m*h/th)]$	$Fb_n =$	23.23 ksi	
	$\Omega =$	1.65	
	$Fb_n/\Omega =$	14.08 ksi	

SHEAR

G.2 Shear Supported on Both Edges - Web

Members with flat elements supported on both edges

Lower slenderness limit	$\lambda 1 =$	35.29	
Upper slenderness limit	$\lambda 2 =$	63.16	
Slenderness	$h/th =$	111.00	$\geq \lambda 2$
$[\pi^2E/(1.25*h/th)^2]$	$Fv_n =$	5.18 ksi	
	$\Omega =$	1.65	
	$Fv_n/\Omega =$	3.14 ksi	

ALLOWABLE STRESSES

Allowable bending stress	$Fb =$	16.82 ksi
Allowable axial stress, compression	$Fac =$	4.89 ksi
Allowable shear stress; webs	$Fv =$	3.14 ksi

Elastic buckling stress	$Fe =$	9.07 ksi
Weighted average allowable compressive stress (per Section E.3.1)	$Fao =$	7.48 ksi

MEMBER LOADING

Bending Moments

Bending moment developed in member	$Mz =$	0.0 kip-ft	
Bending stress developed in member	$fb =$	0.00 ksi	
Allowable bending stress of member	$Fb =$	16.82 ksi	< 1.0

Axial Loads

Axial load developed in member	$Fx =$	0 lb	
Axial stress developed in member	$fa =$	0.00 ksi	
Allowable compressive axial stress of member	$Fac =$	4.89 ksi	< 1.0

Shear Loads

Shear load developed in member	$Vz =$	0 lb	
Shear stress developed in member	$fv =$	0.00 ksi	
Allowable shear stress of member webs	$Fv =$	3.14 ksi	< 1.0

Interaction Equations

	$\sqrt{[(fb/Fb)^2 + (fv/Fv)^2]} =$	0.00	< 1.0
Eq H.1-1	$fa/Fa + fb/Fb =$	0.00	< 1.0
Eq H.3-2	$fa/Fa + (fb/Fb)^2 + (fv/Fv)^2 =$	0.00	< 1.0

CONFIGURATION AND MOMENT TABULATION TOOLS

# of beam=	1
# P load=	0
a=	0.00 ft

Support Type	Beam =	Simple
Beam Length	L =	12.67 ft
Tributary Width	W =	0.00 ft
	P Load=	3237.9 lb
Load on Tributary (LL, WL, DL, etc)	RL =	0.00 psf
Additional Beam Load (Weight or Service Loads)	DL =	0.00 lb/ft
Total Loading on Beam	w =	0.00 lb/ft
Shear Loading at End of Beam	Vy =	0 lbs
CALCULATED MOMENT	Mmax =	0.00 kip-ft

Deflection Check

	Support =	Simple
	Deflection Limit =	L / 180
	w =	0.00 lb/ft
ALLOWABLE DEFLECTION	Δ Allow =	0.84 in
MAXIMUM DEFLECTION	Δ Max =	0.00 in
		0%

OK, Allowable Deflection Sufficient

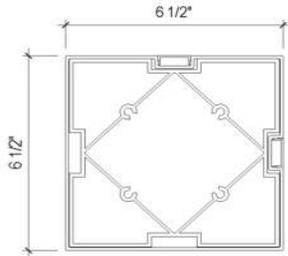
Work Prepared For: Timan Window Treatments, Inc
 Project: 24577 - Raymond Tatko
 Detail/Member: Post Design

ALUMINUM DESIGN MANUAL (2020 EDITION)
Specifications for Aluminum Structures (Buildings)
Allowable Stress Design

Design Check of 6.5"x6.5"x0.125"/0.125" 6063-T6 Aluminum Tube
 Per 2020 Aluminum Design Manual

Alloy: 6063 Temper: T6 Critically Welded: N

MEMBER PROPERTIES



Flange width	b =	6.500"
Flange thickness	tb =	0.125"
Web height	h =	6.500"
Web thickness	th =	0.125"
Moment of inertia about axis parallel to flange	Ix =	40.50 in ⁴
Moment of inertia about axis parallel to web	Iy =	40.50 in ⁴
Section modulus about the x-axis	Sx =	12.46 in ³
Radius of gyration about centroidal axis parallel to flange	rx =	2.33 in
Radius of gyration about centroidal axis parallel to web	ry =	2.33 in
Torsion constant	J =	32.39 in ⁴
Cross sectional area of member	A =	7.44 in ²
Plastic section modulus	Z =	6.05 in ³

MEMBER SPANS

Unsupported member length (between supports)	L =	10.0 ft
Unbraced length for bending (between bracing against side-sway X-Axis)	Lbx =	10.0 ft
Unbraced length for bending (between bracing against side-sway Y-Axis)	Lby =	10.0 ft
Effective length factor	kx =	2.0
	ky =	1.0

MATERIAL PROPERTIES

Tensile ultimate strength	Ftu =	30 ksi
Tensile yield strength	Fty =	25 ksi
Compressive yield strength	Fcy =	25 ksi
Shear ultimate strength	Fsu =	18 ksi
Shear yield strength	Fsy =	15 ksi
Compressive modulus of elasticity	E =	10,100 ksi

BUCKLING CONSTANTS

Compression in columns & beam flanges (Intercept)	Bc =	27.64 ksi
Compression in columns & beam flanges (Slope)	Dc =	0.14 ksi
Compression in columns & beam flanges (Intersection)	Cc =	78.38 ksi
Compression in flat plates (Intercept)	Bp =	31.39 ksi
Compression in flat plates (Slope)	Dp =	0.17 ksi
Compression in flat plates (Intersection)	Cp =	73.55 ksi
Compressive bending stress in solid rectangular bars (Intercept)	Bbr =	46.12 ksi
Compressive bending stress in solid rectangular bars (Slope)	Dbr =	0.38 ksi
Shear stress in flat plates (Intercept)	Bs =	18.98 ksi
Shear stress in flat plates (Slope)	Ds =	0.08 ksi
Shear stress in flat plates (Intersection)	Cs =	94.57 ksi
Ultimate strength coefficient of flat plates in compression (slenderness limit λ2)	k1c =	0.35
Ultimate strength coefficient of flat plates in compression (stress for slenderness > λ2)	k2c =	2.27
Ultimate strength of flat plates in bending (slenderness limit λ2)	k1b =	0.50
Ultimate strength of flat plates in bending (stress for slenderness > λ2)	k2b =	2.04
Tension coefficient	kt =	1.0

D.2 Axial Tension

Tensile Yielding - Unwelded Members	[Fty]	Fty_n =	25.00 ksi
		Ω =	1.65
		Fty_n/Ω =	15.15 ksi
Tensile Rupture - Unwelded Members	[Ftu/kt]	Ftu_n =	30.00 ksi
		Ω =	1.95
		Ftu_n/Ωt =	15.38 ksi

AXIAL COMPRESSION MEMBERS

E.2 Compression Member Buckling

Axial, gross section subject to buckling

Lower slenderness limit	$\lambda 1 =$	18.23	
Upper slenderness limit	$\lambda 2 =$	78.38	
Slenderness	$\lambda(max) =$	102.89	$\geq \lambda 2$
$[0.85\pi^2 E/\lambda^2]$	$F_c_n =$	8.00 ksi	
	$\Omega =$	1.65	
	$F_c_n/\Omega =$	4.85 ksi	

E.3 Local Buckling

For column elements in uniform compression subject to local buckling, the uniform compressive strength is addressed in Section B.5.4 calculated below.

B.5.4.2 - Flat elements supported on both edges (Flange)

B.5.4.2 - Flat elements supported on both edges (Web)

E.4 Buckling Interaction

Per Table B.5.1

$[\pi^2 E / (1.6^2 b/tb)^2]$	$F_e(flange) =$	15.58 ksi
$[F_c_n]$	$F_c_n =$	8.00 ksi
$F_e(flange) > F_c_n$ (E.2 Member Buckling)	$\Omega =$	1.65
	$F_c_n/\Omega =$	4.85 ksi
$[\pi^2 E / (1.6^2 h/th)^2]$	$F_e(web) =$	15.58 ksi
$[F_c_n]$	$F_c_n =$	8.00 ksi
$F_e(web) > F_c_n$ (E.2 Member Buckling)	$\Omega =$	1.65
	$F_c_n/\Omega =$	4.85 ksi

FLEXURAL MEMBERS

F.2 Yielding and Rupture

Nominal flexural strength for yielding and rupture

Limit State of Yielding

$[Z^*F_{cy}]$	$M_{np} =$	151.29 k-in
$[M_{np}/Z]$	$F_b_n =$	25.00 ksi
	$\Omega =$	1.65
	$F_b_n/\Omega =$	15.15 ksi

Limit State of Rupture

$[Z^*F_{tu}/kt]$	$M_{nu} =$	181.55 k-in
$[M_{nu}/Z]$	$F_b_n =$	30.00 ksi
	$\Omega =$	1.95
	$F_b_n/\Omega =$	15.38 ksi

F.4 Lateral-Torsional Buckling

Square or rectangular tubes subject to lateral-torsional buckling

Slenderness for shapes symmetric about the bending axis

Slenderness for closed shapes

Slenderness for any shape

Maximum slenderness

$\lambda F.4.2.1 =$	15.13
$\lambda F.4.2.3 =$	14.78
$\lambda F.4.2.5 =$	15.13
$\lambda(max) =$	15.13

$< C_c$

Nominal flexural strength - lateral-torsional buckling

$[M_{np}(1-(\lambda/C_c)) + (\pi^2 E^* \lambda^2 S_x / C_c^3)]$	$M_{nmb} =$	161.12 k-in
$[M_{nmb}/S_x]$	$F_b_n =$	12.93 ksi
	$\Omega =$	1.65
	$F_b_n/\Omega =$	7.84 ksi

UNIFORM COMPRESSION ELEMENTS

B.5.4.2 Flat Elements Supported on Both Edges - Web & Flange

Uniform compression strength, flat elements supported on both edges

Lower slenderness limit	$\lambda 1 =$	22.8	
Upper slenderness limit	$\lambda 2 =$	39.2	
Flange Slenderness	$b/tb =$	50.0	$\geq \lambda 2$
Web Slenderness	$h/th =$	50.0	$\geq \lambda 2$
$[k2c^* \sqrt{(Bp^*E)/(1.6^2 b/tb)}]$	$F_c_{n1} =$	15.98 ksi	
	$\Omega =$	1.65	
	$F_c_{n1}/\Omega =$	9.68 ksi	
$[k2c^* \sqrt{(Bp^*E)/(1.6^2 h/th)}]$	$F_c_{n2} =$	15.98 ksi	
	$\Omega =$	1.65	
	$F_c_{n2}/\Omega =$	9.68 ksi	

FLEXURAL COMPRESSION ELEMENTS

B.5.5.1 Flat Elements Supported on Both Edges - Web

Flexural compression strength, flat elements supported on both edges

Lower slenderness limit	$\lambda 1 =$	34.73	
Upper slenderness limit	$\lambda 2 =$	92.95	
Slenderness	$h/th =$	50.00	$\lambda 1 - \lambda 2$
$[Bbr-m*Dbr*h/th]$	$Fb_n =$	33.71 ksi	
	$\Omega =$	1.65	
	$Fb_n/\Omega =$	20.43 ksi	

SHEAR

G.2 Shear Supported on Both Edges - Web

Members with flat elements supported on both edges

Lower slenderness limit	$\lambda 1 =$	38.73	
Upper slenderness limit	$\lambda 2 =$	75.65	
Slenderness	$h/th =$	50.00	$\lambda 1 - \lambda 2$
$[Bs-1.25Ds*h/th]$	$Fv_n =$	13.84 ksi	
	$\Omega =$	1.65	
	$Fv_n/\Omega =$	8.39 ksi	

ALLOWABLE STRESSES

Allowable bending stress	$Fb =$	6.54 ksi
Allowable axial stress, compression	$Fac =$	4.85 ksi
Allowable shear stress; webs	$Fv =$	8.39 ksi
Allowable axial stress, Tension	$Fat =$	15.15 ksi

Elastic buckling stress	$Fe =$	4.83 ksi
Weighted average allowable compressive stress (per Section E.3.1)	$Fao =$	9.68 ksi

MEMBER LOADING

Bending Moments

Bending moment developed in member	$Mz =$	1.7 kip-ft	
Bending stress developed in member	$fb =$	1.64 ksi	
Allowable bending stress of member	$Fb =$	6.54 ksi	< 1.0

Compression Loads

Compression load developed in member	$P =$	3,056 lb	
Compression stress developed in member	$fc =$	0.41 ksi	
Allowable compressive axial stress of member	$Fac =$	4.85 ksi	< 1.0

Tension Loads

Tension load developed in member	$T =$	306 lb	
Tension stress developed in member	$ft =$	0.03 ksi	
Allowable Tension axial stress of member	$Fat =$	15.15 ksi	< 1.0

Shear Loads

Shear load developed in member	$Vz =$	462 lb	
Shear stress developed in member	$fv =$	0.30 ksi	
Allowable shear stress of member webs	$Fv =$	8.39 ksi	< 1.0

Interaction Equations

$\sqrt{[(fb/Fb)^2 + (fv/Fv)^2]} =$	0.25	< 1.0
$fa/Fa + fb/Fb =$	0.34	< 1.0
$fa/Fa + (fb/Fb)^2 + (fv/Fv)^2 =$	0.34	< 1.0

CONFIGURATION AND MOMENT TABULATION TOOLS

Member Loads:

$Mx =$ 20.44 kip-in	Applied Moment Per Member	30 PSF	Total Gravity Load
$My =$ 1.36 kip-in	Applied moment Per Member	6.3 FT	Post Trib Area in X-Axis
$Tn =$ 0.33 kip-in	Applied Torsion Per Member	16.1 FT	Post Trib Area in Y-Axis
$Vx =$ 216 lbs	Applied Shear Load Per Member	3 PSF	Uplift
$Vy =$ 408 lbs	Applied Shear Load Per Member	17 PSF	Lateral Load
$V =$ 462 lbs	Applied Resultant Shear Load Per Member		
$P =$ 3,056 lbs	Applied axial compression load		
$T =$ 306 lbs	Applied axial tension load	2.48 kip-in	Seismic Moment

Work Prepared For: Timan Window Treatments, Inc
 Project: 24577 - Raymond Tatko
 Member/Detail: Beam To Beam

Steel Spaced Thread Tapping Screw to Aluminum Connections

†2020 Aluminum Design Manual, *AMMA TIR-A9-2014

Anchor: 1/4-14 SMS, 316 SS, Steel Screw

Size:	1/4-14 SMS	Nominal Anchor Size Designation
Alloy:	316 SS	Screw Material
Ftu=	100 ksi	Anchor Ultimate Tensile Strength
Fy =	65 ksi	Anchor Yield Strength
D =	0.250"	Nominal Screw Diameter (*Table 20.1,20.2)
Dmin =	0.185"	Basic Minor Diameter (*Table 20.1,20.2)
As =	0.027 in ²	Tensile Stress Area (*Table 20.1,20.2)
Ar =	0.027 in ²	Thread Root Area (*Table 20.1,20.2)
n =	14	Thread Per Inch
Dw=	0.625"	Washer Diameter <input type="checkbox"/> Consider Washer?
Dws =	0.500"	Anchor Head Diameter
Dh =	0.250"	Nominal Hole Diameter
Screw Boss?	No	Is anchor placed in a screw boss/chase/slot?
Countersunk?	No	Yes or No?
CS Depth =		Countersink depth
de =	0.500"	Aluminum Edge Distance

Member in Contact with Screw Head:

Alloy 1:	6063-T6	
t1 =	0.125"	Thickness of Member 1
Ftu1 =	30 ksi	Tensile Ultimate Strength of Member 1
Fty1 =	25 ksi	Tensile Yield Strength of Member 1

Member not in Contact with Screw Head:

Alloy 2:	6063-T6	
t2 =	0.125"	Thickness of Member 2
Le =	0.125"	Depth of Full Thread Engagement Into t2 (Not Including Tapping/Drilling Point)
Ftu2 =	30 ksi	Tensile Ultimate Strength of Member 2
Fty2 =	25 ksi	Tensile Yield Strength of Member 2
t3 =	0.125"	Screw Boss Wall Thickness
Le1 =	0.500"	Minimum Depth of Full Thread Engagement Into Screw Boss <i>If Applicable</i> (Not Including Tapping/Drilling Point)

Allowable Tension

C=	1.0	Coeff. Dependent On Screw Location (+Sect. J.5.4.2)
Ks=	1.2	Coeff. Dependent On Member 2 Thickness (+Sect. J.5.4.1.1b)
Rn_t1 =	937.5 lb	Nominal Pull-Out Strength Of Screw (+Sect. J.5.4.1.1b)
Rn_t2 =	937.5 lb	Nominal Pull-Over Strength Of Screw (+Sect. J.5.4.2)
Rn_t3 =	N/A	Nominal Pull-Out Strength From Screw Boss (if applicable) (+Sect. J.5.4.1.2)
Pnt =	896.0 lb	Allowable Tensile Capacity Of Screw (*Eqn. 10.4-10.7)
Ω =	3.0	Safety Factor For Connections; Building Type Structures
Ω =	3.0	Safety Factor For Anchor

Allowable Tension = 313 lb

Allowable Shear:

Rn_v1 =	1875.0 lb	Bearing On Member 1 (+Sect. J.5.5.1)
Rn_v2 =	1875.0 lb	Bearing On Member 2 (+Sect. J.5.5.1)
Rn_v3 =	2784.2 lb	Screw Tilting (+Sect. J.5.5.2)
Rn_v4 =	N/A	Shear Capacity Of Screw Boss Wall
Pnv =	517.3 lb	Allowable Shear Capacity Of Screw (*Eqn. 7.5)
Ω =	3.0	Safety Factor For Connections; Building Type Structures
Ω =	3.0	Safety Factor For Anchor

Allowable Shear = 517 lb

Alternate Options:

- Disregard the limiting allowable capacities from Member 1 (member in contact with screw head)
- Disregard the limiting allowable capacities from Member 2 (member in NOT in contact with screw head)

Concentrated Shear & Tensile Reactions

(Select this connection type)

Qty	6	Anchor Qty at Connection
Treq	0 lb	Required Tensile Loading on Connection
Vreq	1619 lb	Required Shear Loading on Connection
n	1.00	Exponent factor
Tcap	1875 lb	Tensile capacity of connection (Qty * Rz)
Vcap	3104 lb	Shear capacity of connection (Qty * Rx)

$$\frac{R_z}{T_{CAP}} + \frac{R_x}{V_{CAP}} = 0.52$$

OK, (6) anchors sufficient

Work Prepared For: Timan Window Treatments, Inc
 Project: 24577 - Raymond Tatko
 Member/Detail: CLIP TO POST CONNECTION

Steel Spaced Thread Tapping Screw to Aluminum Connections

†2020 Aluminum Design Manual, *AMMA TIR-A9-2014

Anchor: 1/4-14 SMS, 316 SS, Steel Screw

Size:	1/4-14 SMS	Nominal Anchor Size Designation
Alloy:	316 SS	Screw Material
Ftu=	100 ksi	Anchor Ultimate Tensile Strength
Fy =	65 ksi	Anchor Yield Strength
D =	0.250"	Nominal Screw Diameter (*Table 20.1,20.2)
Dmin =	0.185"	Basic Minor Diameter (*Table 20.1,20.2)
As =	0.027 in ²	Tensile Stress Area (*Table 20.1,20.2)
Ar =	0.027 in ²	Thread Root Area (*Table 20.1,20.2)
n =	14	Thread Per Inch
Dw=	0.625"	Washer Diameter <input type="checkbox"/> Consider Washer?
Dws =	0.500"	Anchor Head Diameter
Dh =	0.250"	Nominal Hole Diameter
Screw Boss?	No	Is anchor placed in a screw boss/chase/slot?
Countersunk?	No	Yes or No?
CS Depth =		Countersink depth
de =	0.500"	Aluminum Edge Distance

Member in Contact with Screw Head:

Alloy 1:	6063-T6	
t1 =	0.125"	Thickness of Member 1
Ftu1 =	30 ksi	Tensile Ultimate Strength of Member 1
Fty1 =	25 ksi	Tensile Yield Strength of Member 1

Member not in Contact with Screw Head:

Alloy 2:	6063-T6	
t2 =	0.125"	Thickness of Member 2
Le =	0.125"	Depth of Full Thread Engagement Into t2 (Not Including Tapping/Drilling Point)
Ftu2 =	30 ksi	Tensile Ultimate Strength of Member 2
Fty2 =	25 ksi	Tensile Yield Strength of Member 2
t3 =	0.125"	Screw Boss Wall Thickness
Le1 =	0.500"	Minimum Depth of Full Thread Engagement Into Screw Boss <i>If Applicable</i> (Not Including Tapping/Drilling Point)

Allowable Tension

C=	1.0	Coeff. Dependent On Screw Location (+Sect. J.5.4.2)
Ks=	1.2	Coeff. Dependent On Member 2 Thickness (+Sect. J.5.4.1.1b)
Rn_t1 =	937.5 lb	Nominal Pull-Out Strength Of Screw (+Sect. J.5.4.1.1b)
Rn_t2 =	937.5 lb	Nominal Pull-Over Strength Of Screw (+Sect. J.5.4.2)
Rn_t3 =	N/A	Nominal Pull-Out Strength From Screw Boss (if applicable) (+Sect. J.5.4.1.2)
Pnt =	896.0 lb	Allowable Tensile Capacity Of Screw (*Eqn. 10.4-10.7)
Ω =	3.0	Safety Factor For Connections; Building Type Structures
Ω =	3.0	Safety Factor For Anchor

Allowable Tension = 313 lb

Allowable Shear:

Rn_v1 =	1875.0 lb	Bearing On Member 1 (+Sect. J.5.5.1)
Rn_v2 =	1875.0 lb	Bearing On Member 2 (+Sect. J.5.5.1)
Rn_v3 =	2784.2 lb	Screw Tilting (+Sect. J.5.5.2)
Rn_v4 =	N/A	Shear Capacity Of Screw Boss Wall
Pnv =	517.3 lb	Allowable Shear Capacity Of Screw (*Eqn. 7.5)
Ω =	3.0	Safety Factor For Connections; Building Type Structures
Ω =	3.0	Safety Factor For Anchor

Allowable Shear = 517 lb

Alternate Options:

- Disregard the limiting allowable capacities from Member 1 (member in contact with screw head)
- Disregard the limiting allowable capacities from Member 2 (member in NOT in contact with screw head)

Concentrated Shear & Tensile Reactions

(Select this connection type)

Qty	6	Anchor Qty at Connection
Treq	0 lb	Required Tensile Loading on Connection
Vreq	462 lb	Required Shear Loading on Connection
n	1.00	Exponent factor
Tcap	1875 lb	Tensile capacity of connection (Qty * Rz)
Vcap	3104 lb	Shear capacity of connection (Qty * Rx)

$$\frac{R_z}{T_{CAP}} + \frac{R_x}{V_{CAP}} = 0.15$$

OK, (6) anchors sufficient

Work Prepared For: Timan Window Treatments, Inc
 Project: 24577 - Raymond Tatko
 Detail: Ledger connection to Host

Loading

Design Uplift (WL) = 3.00 psf

Design Gravity Load ("Grav") = 24.00 psf

Governing Load



Ledger Configuration

W =	6.33 ft	Tributary Width
RL =	30.00 psf	Load on Tributary (LL, WL, DL, etc)
FL =	30.00 psf	Load on Tributary (LL, WL, DL, etc)
DL =	0.00 lb/ft	Additional Beam Load (Weight or Service Loads)
Pv =	190.00 lb/ft	Total Vertical Loading on Beam
Pf =	19.73 lb/ft	Total Lateral Loading on Beam
S =	1.33 ft	Anchor spacing
# of Screws =	3 pcs	Number of anchors per location
Shear required =	84.44 lb	Per anchor
Tension required =	9.38 lb	Per anchor
Shear Connection Capacity =	147.00 lb	
		Tension Connection Capacity = 405.00 lb
Connection Check =	0.60	Ledger Connection OK

Bracket Spacing =	16	in
Shear required =	253.33 lb	Bracket
Tension required =	26.31 lb	Bracket
Gutter Size =	7.625	in
Moment Arm =	1931.6667	lb-in
Beam Depth =	5.3125	in
T/C Required =	390	lb

Work Prepared For: Timan Window Treatments, Inc
 Project: 24577 - Raymond Tatko
 Detail/Member: Top Bent Plate

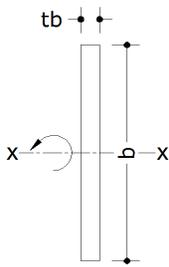
ALUMINUM DESIGN MANUAL (2020 EDITION)
Specifications for Aluminum Structures (Buildings)
Allowable Stress Design

Design Check of 0.25"x3" 6063-T6 Aluminum Flat Plate

Per 2020 Aluminum Design Manual

Alloy: 6063 Temper: T6 Critically Welded: N

MEMBER PROPERTIES



Flat Plate Height	$b =$	0.250"
Flat Plate Thickness	$tb =$	3.000"
Moment of inertia about axis parallel to flange	$I_x =$	0.00 in ⁴
Moment of inertia about axis parallel to web	$I_y =$	0.56 in ⁴
Section modulus about the x-axis	$S_x =$	0.03 in ³
Radius of gyration about centroidal axis parallel to flange	$r_x =$	0.07 in
Radius of gyration about centroidal axis parallel to web	$r_y =$	0.87 in
Torsion constant	$J =$	2.25 in ⁴
Cross sectional area of member	$A =$	0.75 in ²
Plastic section modulus	$Z =$	0.05 in ³
Warping constant	$C_w =$	0.00 in ⁶

MEMBER SPANS

Unsupported member length (between supports)	$L =$	0.64 ft
Unbraced length for bending (between bracing against side-sway)	$L_b =$	0.64 ft
Effective length factor	$k =$	1.0

MATERIAL PROPERTIES

Tensile ultimate strength	$F_{tu} =$	30 ksi
Tensile yield strength	$F_{ty} =$	25 ksi
Compressive yield strength	$F_{cy} =$	25 ksi
Shear ultimate strength	$F_{su} =$	18 ksi
Shear yield strength	$F_{sy} =$	15 ksi
Compressive modulus of elasticity	$E =$	10,100 ksi

BUCKLING CONSTANTS

Compression in columns & beam flanges (Intercept)	$B_c =$	27.64 ksi
Compression in columns & beam flanges (Slope)	$D_c =$	0.14 ksi
Compression in columns & beam flanges (Intersection)	$C_c =$	78.38 ksi
Compression in flat plates (Intercept)	$B_p =$	31.39 ksi
Compression in flat plates (Slope)	$D_p =$	0.17 ksi
Compression in flat plates (Intersection)	$C_p =$	73.55 ksi
Compressive bending stress in solid rectangular bars (Intercept)	$B_{br} =$	46.12 ksi
Compressive bending stress in solid rectangular bars (Slope)	$D_{br} =$	0.38 ksi
Compressive bending stress in solid rectangular bars (Intersection)	$C_{br} =$	80.56 ksi
Shear stress in flat plates (Intercept)	$B_s =$	18.98 ksi
Shear stress in flat plates (Slope)	$D_s =$	0.08 ksi
Shear stress in flat plates (Intersection)	$C_s =$	94.57 ksi
Ultimate strength coefficient of flat plates in compression (slenderness limit λ_2)	$k_{1c} =$	0.35
Ultimate strength coefficient of flat plates in compression (stress for slenderness $> \lambda_2$)	$k_{2c} =$	2.27
Ultimate strength of flat plates in bending (slenderness limit λ_2)	$k_{1b} =$	0.50
Ultimate strength of flat plates in bending (stress for slenderness $> \lambda_2$)	$k_{2b} =$	2.04
Tension coefficient	$kt =$	1.0

D.2 Axial Tension

Tensile Yielding - Unwelded Members	$[F_{ty}]$	$F_{ty_n} =$	25.00 ksi
		$\Omega =$	1.65
		$F_{ty_n}/\Omega =$	15.15 ksi
Tensile Rupture - Unwelded Members	$[F_{tu}/kt]$	$F_{tu_n} =$	30.00 ksi
		$\Omega =$	1.95
		$F_{tu_n}/\Omega =$	15.38 ksi

AXIAL COMPRESSION MEMBERS

E.2 Compression Member Buckling

Axial, gross section subject to buckling	Lower slenderness limit	$\lambda 1 =$	18.23	
	Upper slenderness limit	$\lambda 2 =$	78.38	
	Slenderness	$\lambda(max) =$	105.66	$\geq \lambda 2$
	$[0.85\pi^2 E/\lambda^2]$	$F_{c_n} =$	7.59 ksi	
		$\Omega =$	1.65	
		$F_{c_n}/\Omega =$	4.60 ksi	

FLEXURAL MEMBERS

F.2 Yielding and Rupture

Nominal flexural strength for yielding and rupture	Limit State of Yielding			
	$[Z*F_{cy}]$	$M_{np} =$	1.17 k-in	
	$[M_{np}/Z]$	$F_b =$	25.00 ksi	
		$\Omega =$	1.65	
		$F_{b_n}/\Omega =$	15.15 ksi	
		Limit State of Rupture		
	$[Z*F_{tu}/kt]$	$M_{nu} =$	1.41 k-in	
	$[M_{nu}/Z]$	$F_b =$	30.00 ksi	
		$\Omega =$	1.95	
		$F_{b_n}/\Omega =$	15.38 ksi	

F.4 Lateral-Torsional Buckling

Rectangular bars subject to lateral-torsional buckling	Slenderness for shapes symmetric about the bending axis	$\lambda F.4.2.1 =$	1.05	
	Slenderness for rectangular bars	$\lambda F.4.2.4 =$	1.06	
	Slenderness for any shape	$\lambda F.4.2.5 =$	1.05	
	Maximum slenderness	$\lambda(max) =$	1.06	$< C_c$
Nominal flexural strength - lateral-torsional buckling	$[M_{np}(1-(\lambda/C_c))+(\pi^2*E*\lambda*S_x/C_c^3)]$	$M_{nmb} =$	1.16 k-in	
	$[M_{nmb}/S_x]$	$F_{b_n} =$	37.21 ksi	
		$\Omega =$	1.65	
		$F_{b_n}/\Omega =$	22.55 ksi	

SHEAR

G.2 Shear Supported on Both Edges

Members with flat elements supported on both edges	Lower slenderness limit	$\lambda 1 =$	38.73	
	Upper slenderness limit	$\lambda 2 =$	75.65	
	Slenderness	$b/t_b =$	0.08	$\leq \lambda 1$
	$[F_{sy}]$	$F_{v_n} =$	15.00 ksi	
		$\Omega =$	1.65	
		$F_{v_n}/\Omega =$	9.09 ksi	

ALLOWABLE STRESSES

Allowable bending stress	$F_b =$	15.15 ksi
Allowable axial stress, compression	$F_{ac} =$	4.60 ksi
Allowable shear stress	$F_v =$	9.09 ksi

Elastic buckling stress	$F_e =$	4.58 ksi
Weighted average allowable compressive stress (per Section E.3.1)	$F_{ao} =$	4.60 ksi

MEMBER LOADING

Bending Moments

Bending moment developed in member	$M_z =$	0.0 kip-ft	
Bending stress developed in member	$f_b =$	0.00 ksi	
Allowable bending stress of member	$F_b =$	15.15 ksi	< 1.0

Axial Loads

Axial load developed in member	$F_x =$	390 lb	
Axial stress developed in member	$f_a =$	0.52 ksi	
Allowable compressive axial stress of member	$F_{ac} =$	4.60 ksi	< 1.0

Shear Loads

Shear load developed in member	$V_z =$	253 lb	
Shear stress developed in member	$f_v =$	0.34 ksi	
Allowable shear stress of member	$F_v =$	9.09 ksi	< 1.0

Interaction Equations

	$\sqrt{[(f_b/F_b)^2 + (f_v/F_v)^2]} =$	0.00	< 1.0
Eq H.1-1	$f_a/F_a + f_b/F_b =$	0.00	< 1.0
Eq H.3-2	$f_a/F_a + (f_b/F_b)^2 + (f_v/F_v)^2 =$	0.00	< 1.0

Work Prepared For: Timan Window Treatments, Inc
 Project: 24577 - Raymond Tatko
 Member/Detail: Gutter Connection Check

Steel Spaced Thread Tapping Screw to Aluminum Connections

†2020 Aluminum Design Manual, *AMMA TIR-A9-2014

Anchor: 1/4-14 SMS, 316 SS, Steel Screw

Size:	1/4-14 SMS	Nominal Anchor Size Designation
Alloy:	316 SS	Screw Material
Ftu=	100 ksi	Anchor Ultimate Tensile Strength
Fy =	65 ksi	Anchor Yield Strength
D =	0.250"	Nominal Screw Diameter (*Table 20.1,20.2)
Dmin =	0.185"	Basic Minor Diameter (*Table 20.1,20.2)
As =	0.027 in ²	Tensile Stress Area (*Table 20.1,20.2)
Ar =	0.027 in ²	Thread Root Area (*Table 20.1,20.2)
n =	14	Thread Per Inch
Dw=	0.625"	Washer Diameter <input type="checkbox"/> Consider Washer?
Dws =	0.500"	Anchor Head Diameter
Dh =	0.250"	Nominal Hole Diameter
Screw Boss?	No	Is anchor placed in a screw boss/chase/slot?
Countersunk?	No	Yes or No?
CS Depth =		Countersink depth
de =	0.500"	Aluminum Edge Distance

Member in Contact with Screw Head:

Alloy 1:	6063-T6	
t1 =	0.141"	Thickness of Member 1
Ftu1 =	30 ksi	Tensile Ultimate Strength of Member 1
Fty1 =	25 ksi	Tensile Yield Strength of Member 1

Member not in Contact with Screw Head:

Alloy 2:	6063-T6	
t2 =	0.125"	Thickness of Member 2
Le =	0.125"	Depth of Full Thread Engagement Into t2 (Not Including Tapping/Drilling Point)
Ftu2 =	30 ksi	Tensile Ultimate Strength of Member 2
Fty2 =	25 ksi	Tensile Yield Strength of Member 2
t3 =	0.125"	Screw Boss Wall Thickness
Le1 =	0.500"	Minimum Depth of Full Thread Engagement Into Screw Boss <i>If Applicable</i> (Not Including Tapping/Drilling Point)

Allowable Tension		
C=	1.0	Coeff. Dependent On Screw Location (+Sect. J.5.4.2)
Ks=	1.2	Coeff. Dependent On Member 2 Thickness (+Sect. J.5.4.1.1b)
Rn_t1 =	937.5 lb	Nominal Pull-Out Strength Of Screw (+Sect. J.5.4.1.1b)
Rn_t2 =	1054.7 lb	Nominal Pull-Over Strength Of Screw (+Sect. J.5.4.2)
Rn_t3 =	N/A	Nominal Pull-Out Strength From Screw Boss (if applicable) (+Sect. J.5.4.1.2)
Pnt =	896.0 lb	Allowable Tensile Capacity Of Screw (*Eqn. 10.4-10.7)
Ω =	3.0	Safety Factor For Connections; Building Type Structures
Ω =	3.0	Safety Factor For Anchor
Allowable Tension = 313 lb		
Allowable Shear:		
Rn_v1 =	2109.4 lb	Bearing On Member 1 (+Sect. J.5.5.1)
Rn_v2 =	1875.0 lb	Bearing On Member 2 (+Sect. J.5.5.1)
Rn_v3 =	2784.2 lb	Screw Tilting (+Sect. J.5.5.2)
Rn_v4 =	N/A	Shear Capacity Of Screw Boss Wall
Pnv =	517.3 lb	Allowable Shear Capacity Of Screw (*Eqn. 7.5)
Ω =	3.0	Safety Factor For Connections; Building Type Structures
Ω =	3.0	Safety Factor For Anchor
Allowable Shear = 517 lb		
Alternate Options:		
<input type="checkbox"/> Disregard the limiting allowable capacities from Member 1 (member in contact with screw head)		
<input type="checkbox"/> Disregard the limiting allowable capacities from Member 2 (member in NOT in contact with screw head)		
Concentrated Shear & Tensile Reactions		<input checked="" type="checkbox"/> (Select this connection type)
Qty	4	Anchor Qty at Connection
Treq	390 lb	Required Tensile Loading on Connection
Vreq	0 lb	Required Shear Loading on Connection
n	1.00	Exponent factor
Tcap	1250 lb	Tensile capacity of connection (Qty * Rz)
Vcap	2069 lb	Shear capacity of connection (Qty * Rx)
$\frac{R_z}{T_{CAP}} + \frac{R_x}{V_{CAP}} =$		0.31
OK, (4) anchors sufficient		

Work Prepared For: Hudson
 Project: 24577 - Raymond Tatko
 Detail/Member: Base Plate Design

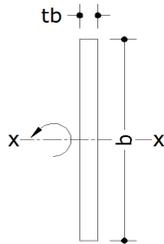
ALUMINUM DESIGN MANUAL (2020 EDITION)
Specifications for Aluminum Structures (Buildings)
Allowable Stress Design

Design Check of 10.625"x0.5" 6063-T6 Aluminum Flat Plate

Per 2020 Aluminum Design Manual

Alloy: **6063** Temper: **T6** Critically Welded: **N**

MEMBER PROPERTIES



Flat Plate Height	<i>b</i> =	10.625"
Flat Plate Thickness	<i>tb</i> =	0.500"
Moment of inertia about axis parallel to flange	<i>I_x</i> =	49.98 in ⁴
Moment of inertia about axis parallel to web	<i>I_y</i> =	0.11 in ⁴
Section modulus about the x-axis	<i>S_x</i> =	9.41 in ³
Radius of gyration about centroidal axis parallel to flange	<i>r_x</i> =	3.07 in
Radius of gyration about centroidal axis parallel to web	<i>r_y</i> =	0.14 in
Torsion constant	<i>J</i> =	0.44 in ⁴
Cross sectional area of member	<i>A</i> =	5.31 in ²
Plastic section modulus	<i>Z</i> =	14.11 in ³
Warping constant	<i>C_w</i> =	0.00 in ⁶

MEMBER SPANS

Unsupported member length (between supports)	<i>L</i> =	0.72 ft
Unbraced length for bending (between bracing against side-sway)	<i>L_b</i> =	0.72 ft
Effective length factor	<i>k</i> =	1.0

MATERIAL PROPERTIES

Tensile ultimate strength	<i>F_{tu}</i> =	30 ksi
Tensile yield strength	<i>F_{ty}</i> =	25 ksi
Compressive yield strength	<i>F_{cy}</i> =	25 ksi
Shear ultimate strength	<i>F_{su}</i> =	18 ksi
Shear yield strength	<i>F_{sy}</i> =	15 ksi
Compressive modulus of elasticity	<i>E</i> =	10,100 ksi

BUCKLING CONSTANTS

Compression in columns & beam flanges (Intercept)	<i>B_c</i> =	27.64 ksi
Compression in columns & beam flanges (Slope)	<i>D_c</i> =	0.14 ksi
Compression in columns & beam flanges (Intersection)	<i>C_c</i> =	78.38 ksi
Compression in flat plates (Intercept)	<i>B_p</i> =	31.39 ksi
Compression in flat plates (Slope)	<i>D_p</i> =	0.17 ksi
Compression in flat plates (Intersection)	<i>C_p</i> =	73.55 ksi
Compressive bending stress in solid rectangular bars (Intercept)	<i>B_{br}</i> =	46.12 ksi
Compressive bending stress in solid rectangular bars (Slope)	<i>D_{br}</i> =	0.38 ksi
Compressive bending stress in solid rectangular bars (Intersection)	<i>C_{br}</i> =	80.56 ksi
Shear stress in flat plates (Intercept)	<i>B_s</i> =	18.98 ksi
Shear stress in flat plates (Slope)	<i>D_s</i> =	0.08 ksi
Shear stress in flat plates (Intersection)	<i>C_s</i> =	94.57 ksi
Ultimate strength coefficient of flat plates in compression (slenderness limit λ ₂)	<i>k_{1c}</i> =	0.35
Ultimate strength coefficient of flat plates in compression (stress for slenderness > λ ₂)	<i>k_{2c}</i> =	2.27
Ultimate strength of flat plates in bending (slenderness limit λ ₂)	<i>k_{1b}</i> =	0.50
Ultimate strength of flat plates in bending (stress for slenderness > λ ₂)	<i>k_{2b}</i> =	2.04
Tension coefficient	<i>kt</i> =	1.0

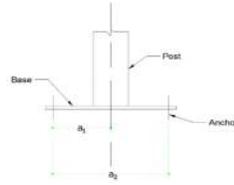
D.2 Axial Tension

Tensile Yielding - Unwelded Members	<i>[F_{ty}]</i>	<i>F_{ty_n}</i> =	25.00 ksi
		<i>Ω</i> =	1.65
		<i>F_{ty_n/Ω}</i> =	15.15 ksi
Tensile Rupture - Unwelded Members	<i>[F_{tu}/kt]</i>	<i>F_{tu_n}</i> =	30.00 ksi
		<i>Ω</i> =	1.95
		<i>F_{tu_n/Ω}</i> =	15.38 ksi

AXIAL COMPRESSION MEMBERS			
E.2 Compression Member Buckling			
Axial, gross section subject to buckling	Lower slenderness limit	$\lambda_1 =$	18.23
	Upper slenderness limit	$\lambda_2 =$	78.38
	Slenderness	$\lambda(max) =$	59.76 < λ_2
	$[(Bc - Cc * \lambda)(0.85 + 0.15 * ((Cc - \lambda) / (Cc - \lambda_1)))]$	$F_{c_n} =$	17.03 ksi
		$\Omega =$	1.65
		$F_{c_n} / \Omega =$	10.32 ksi
FLEXURAL MEMBERS			
F.2 Yielding and Rupture			
Nominal flexural strength for yielding and rupture	Limit State of Yielding	$M_{np} =$	352.78 k-in
	$[Z * F_{cy}]$	$F_b =$	25.00 ksi
	$[M_{np} / Z]$	$\Omega =$	1.65
		$F_{b_n} / \Omega =$	15.15 ksi
	Limit State of Rupture	$M_{nu} =$	423.34 k-in
	$[Z * F_{tu} / kt]$	$F_b =$	30.00 ksi
	$[M_{nu} / Z]$	$\Omega =$	1.95
		$F_{b_n} / \Omega =$	15.38 ksi
F.4 Lateral-Torsional Buckling			
Rectangular bars subject to lateral-torsional buckling	Slenderness for shapes symmetric about the bending axis	$\lambda_{F.4.2.1} =$	62.30
	Slenderness for rectangular bars	$\lambda_{F.4.2.4} =$	44.04
	Slenderness for any shape	$\lambda_{F.4.2.5} =$	62.30
	Maximum slenderness	$\lambda(max) =$	62.30 < C_c
Nominal flexural strength - lateral-torsional buckling	$[M_{np}(1 - (\lambda / C_c)) + (\pi^2 * E * \lambda^2 * S_x / C_c^3)]$	$M_{nmb} =$	193.69 k-in
	$[M_{nmb} / S_x]$	$F_{b_n} =$	20.59 ksi
		$\Omega =$	1.65
		$F_{b_n} / \Omega =$	12.48 ksi
G.2 Shear Supported on Both Edges			
Members with flat elements supported on both edges	Lower slenderness limit	$\lambda_1 =$	38.73
	Upper slenderness limit	$\lambda_2 =$	75.65
	Slenderness	$b / t_b =$	21.25 $\leq \lambda_1$
	$[F_{sy}]$	$F_{v_n} =$	15.00 ksi
		$\Omega =$	1.65
		$F_{v_n} / \Omega =$	9.09 ksi
ALLOWABLE STRESSES			
	Allowable bending stress	$F_b =$	12.48 ksi
	Allowable axial stress, compression	$F_{ac} =$	10.32 ksi
	Allowable shear stress	$F_v =$	9.09 ksi
	Elastic buckling stress	$F_e =$	14.32 ksi
Weighted average allowable compressive stress (per Section E.3.1)		$F_{ao} =$	10.32 ksi

MEMBER LOADING			
<u>Bending Moments</u>			
Bending moment developed in member		$M_z =$	2.9 kip-ft
Bending stress developed in member		$f_b =$	3.70 ksi
Allowable bending stress of member		$F_b =$	12.48 ksi < 1.0
<u>Axial Loads</u>			
Axial load developed in member		$F_x =$	462 lb
Axial stress developed in member		$f_a =$	0.09 ksi
Allowable compressive axial stress of member		$F_{ac} =$	10.32 ksi < 1.0
<u>Shear Loads</u>			
Shear load developed in member		$V_z =$	3,056 lb
Shear stress developed in member		$f_v =$	0.58 ksi
Allowable shear stress of member		$F_v =$	9.09 ksi < 1.0
<u>Interaction Equations</u>			
		$\sqrt{[(f_b/F_b)^2 + (f_v/F_v)^2]} =$	0.30 < 1.0
Eq H.1-1		$f_a/F_a + f_b/F_b =$	0.31 < 1.0
Eq H.3-2		$f_a/F_a + (f_b/F_b)^2 + (f_v/F_v)^2 =$	0.10 < 1.0

Work Prepared For: Timan Window Treatments, Inc
Project: 24577 - Raymond Tatko
Mark/Detail: Post anchors to concrete



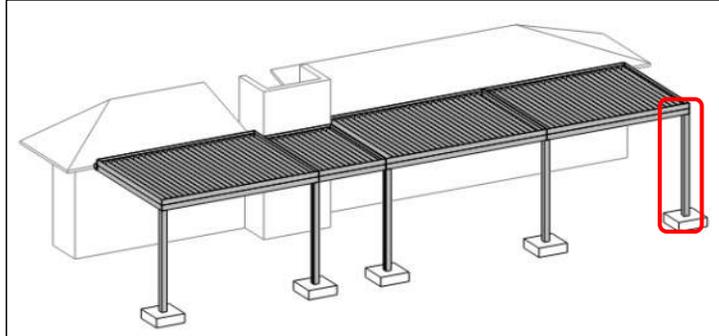
Design Loads (LRFD)

M	32710 lb-in
T	489 lb
C	4889 lb
V	739 lb

Support Posts

Check intermediate, edge, or none?

Edge Support Post Configuration



Mounting Height Above Grade: (Enter 0 for installations at ground level)

Height of Posts:

Attached to host?

Roof Eave Height=

Total Mean Roof Height:

Support Posts - Edge Condition Analysis

Post Size: x x

X - AXIS Post Trib = ft

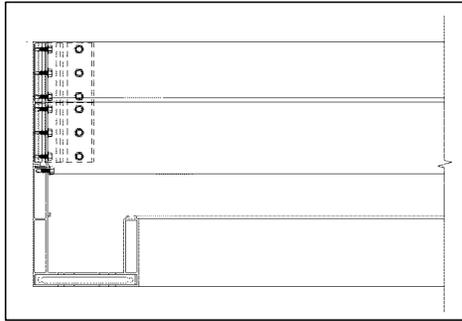
Y - AXIS Post Trib = ft

FH1 = in, (side fascia height at HT1, normal to lateral windload)

Host Attached

ASD Method	Max. Moment/Axial/Shear	29%
	Moment/Axial Check:	29%
	Shear Check	2%
	Required Tension:	140 lb
	Required Compression:	1401 lb
	Required Shear:	320 lb
	Required Moment	1.7 kip-ft

Beam to Beam (Clip in Shear)



Qty **6** Anchor Qty at Connection per Beam

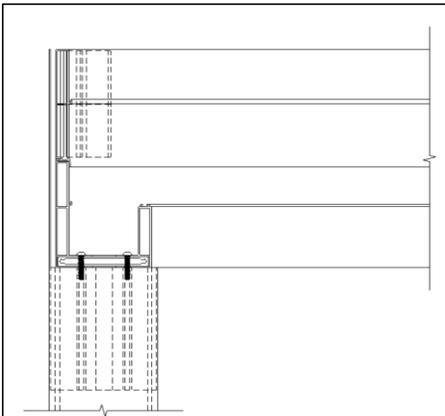
Required Tension:	0 lb
Required Shear:	1619 lb

(Analyzing 1/4-14 SMS, 316 SS, Steel Screw to 0.125" x 0.125" connecting parts thicknesses)

52%

OK, (6) anchors sufficient

Beam to Post Insert Connection



Qty **4** Anchor Qty at Connection

F 5148.59 lb $\Omega = 4$
 Fsu 60000.00
 Ats 0.34 in²
 n 20.00 threads/in
 LE 1.57 in
 Dsmin 0.31 in
 Enmax 0.28 in

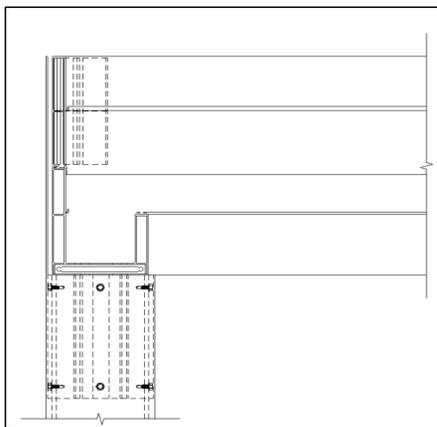
Required Tension:	140 lb
Capacity Shear:	20594 lb

Analyzing Beam to Post Insert with (4) M8x40 Machine screws

1%

OK

Post Insert to Post Connection



Qty **6** Anchor Qty at Connection

Required Tension:	0 lb
Required Shear:	320 lb

(Analyzing 1/4-14 SMS, 316 SS, Steel Screw to 0.125" x 0.125" connecting parts thicknesses)

10%

OK, (6) anchors sufficient

Foundation

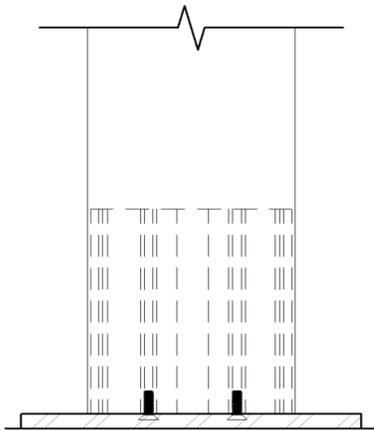
SEE SKYCIV			
Foundation Reactions (K,K-FT)			
	P	V	M
D	0.234	0.000	0.000
W	0.313	0.394	2.839
E	0.007	0.019	0.136
Lr	0.561	0.000	0.000
S	1.168	0.000	0.000

Baseplate 4-Bolt
 Plate Thickness: 0.5 in

Spacing: 8.625 in
 Width: 10.625 in

	SEE ANCHOR CALC
30%	Okay

Post Insert to Base Plate Connection



Qty 4 Anchor Qty at Connection

F	5148.59 lb	$\Omega = 4$
Fsu	60000.00	
Ats	0.34 in ²	
n	20.00 threads/in	
LE	1.57 in	
Dsmin	0.31 in	
Enmax	0.28 in	

Required Tension:	8608 lb
Capacity Shear:	20594 lb

Analyzing Beam to Post Insert with (4)
 M8x40 Machine screws

42%

OK

Work Prepared For: Timan Window Treatments, Inc
 Project: 24577 - Raymond Tatko
 Detail/Member: Post Design

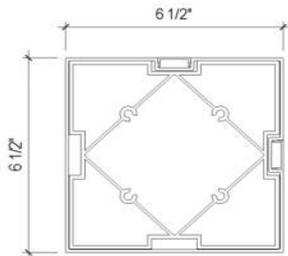
ALUMINUM DESIGN MANUAL (2020 EDITION)
Specifications for Aluminum Structures (Buildings)
Allowable Stress Design

Design Check of 6.5"x6.5"x0.125"/0.125" 6063-T6 Aluminum Tube

Per 2020 Aluminum Design Manual

Alloy: 6063 Temper: T6 Critically Welded: N

MEMBER PROPERTIES



Flange width	b =	6.500"
Flange thickness	tb =	0.125"
Web height	h =	6.500"
Web thickness	th =	0.125"
Moment of inertia about axis parallel to flange	Ix =	40.50 in ⁴
Moment of inertia about axis parallel to web	Iy =	40.50 in ⁴
Section modulus about the x-axis	Sx =	12.46 in ³
Radius of gyration about centroidal axis parallel to flange	rx =	2.33 in
Radius of gyration about centroidal axis parallel to web	ry =	2.33 in
Torsion constant	J =	32.39 in ⁴
Cross sectional area of member	A =	7.44 in ²
Plastic section modulus	Z =	6.05 in ³

MEMBER SPANS

Unsupported member length (between supports)	L =	10.0 ft
Unbraced length for bending (between bracing against side-sway X-Axis)	Lbx =	10.0 ft
Unbraced length for bending (between bracing against side-sway Y-Axis)	Lby =	10.0 ft
Effective length factor	kx =	2.0
	ky =	1.0

MATERIAL PROPERTIES

Tensile ultimate strength	Ftu =	30 ksi
Tensile yield strength	Fty =	25 ksi
Compressive yield strength	Fcy =	25 ksi
Shear ultimate strength	Fsu =	18 ksi
Shear yield strength	Fsy =	15 ksi
Compressive modulus of elasticity	E =	10,100 ksi

BUCKLING CONSTANTS

Compression in columns & beam flanges (Intercept)	Bc =	27.64 ksi
Compression in columns & beam flanges (Slope)	Dc =	0.14 ksi
Compression in columns & beam flanges (Intersection)	Cc =	78.38 ksi
Compression in flat plates (Intercept)	Bp =	31.39 ksi
Compression in flat plates (Slope)	Dp =	0.17 ksi
Compression in flat plates (Intersection)	Cp =	73.55 ksi
Compressive bending stress in solid rectangular bars (Intercept)	Bbr =	46.12 ksi
Compressive bending stress in solid rectangular bars (Slope)	Dbr =	0.38 ksi
Shear stress in flat plates (Intercept)	Bs =	18.98 ksi
Shear stress in flat plates (Slope)	Ds =	0.08 ksi
Shear stress in flat plates (Intersection)	Cs =	94.57 ksi
Ultimate strength coefficient of flat plates in compression (slenderness limit λ2)	k1c =	0.35
Ultimate strength coefficient of flat plates in compression (stress for slenderness > λ2)	k2c =	2.27
Ultimate strength of flat plates in bending (slenderness limit λ2)	k1b =	0.50
Ultimate strength of flat plates in bending (stress for slenderness > λ2)	k2b =	2.04
Tension coefficient	kt =	1.0

D.2 Axial Tension

Tensile Yielding - Unwelded Members	[Fty]	Fty_n =	25.00 ksi
		Ω =	1.65
		Fty_n/Ω =	15.15 ksi
Tensile Rupture - Unwelded Members	[Ftu/kt]	Ftu_n =	30.00 ksi
		Ω =	1.95
		Ftu_n/Ωt =	15.38 ksi

AXIAL COMPRESSION MEMBERS

E.2 Compression Member Buckling

Axial, gross section subject to buckling

Lower slenderness limit	$\lambda 1 =$	18.23	
Upper slenderness limit	$\lambda 2 =$	78.38	
Slenderness	$\lambda(max) =$	102.89	$\geq \lambda 2$
$[0.85\pi^2 E/\lambda^2]$	$F_c_n =$	8.00 ksi	
	$\Omega =$	1.65	
	$F_c_n/\Omega =$	4.85 ksi	

E.3 Local Buckling

For column elements in uniform compression subject to local buckling, the uniform compressive strength is addressed in Section B.5.4 calculated below.

B.5.4.2 - Flat elements supported on both edges (Flange)

B.5.4.2 - Flat elements supported on both edges (Web)

E.4 Buckling Interaction

Per Table B.5.1

$[\pi^2 E / (1.6 * b / t b)^2]$	$F_e(flange) =$	15.58 ksi
$[F_c_n]$	$F_c_n =$	8.00 ksi
$F_e(flange) > F_c_n$ (E.2 Member Buckling)	$\Omega =$	1.65
	$F_c_n/\Omega =$	4.85 ksi
$[\pi^2 E / (1.6 * h / t h)^2]$	$F_e(web) =$	15.58 ksi
$[F_c_n]$	$F_c_n =$	8.00 ksi
$F_e(web) > F_c_n$ (E.2 Member Buckling)	$\Omega =$	1.65
	$F_c_n/\Omega =$	4.85 ksi

FLEXURAL MEMBERS

F.2 Yielding and Rupture

Nominal flexural strength for yielding and rupture

Limit State of Yielding

$[Z * F_{cy}]$

$[M_{np}/Z]$

$M_{np} =$	151.29 k-in
$F_b_n =$	25.00 ksi
$\Omega =$	1.65
$F_b_n/\Omega =$	15.15 ksi

Limit State of Rupture

$[Z * F_{tu}/k_t]$

$[M_{nu}/Z]$

$M_{nu} =$	181.55 k-in
$F_b_n =$	30.00 ksi
$\Omega =$	1.95
$F_b_n/\Omega =$	15.38 ksi

F.4 Lateral-Torsional Buckling

Square or rectangular tubes subject to lateral-torsional buckling

Slenderness for shapes symmetric about the bending axis

Slenderness for closed shapes

Slenderness for any shape

Maximum slenderness

$\lambda F.4.2.1 =$	15.13
$\lambda F.4.2.3 =$	14.78
$\lambda F.4.2.5 =$	15.13
$\lambda(max) =$	15.13

$< C_c$

Nominal flexural strength - lateral-torsional buckling

$[M_{np}(1 - (\lambda/C_c)) + (\pi^2 * E * \lambda * S_x / C_c^3)]$

$[M_{nmb}/S_x]$

$M_{nmb} =$	161.12 k-in
$F_b_n =$	12.93 ksi
$\Omega =$	1.65
$F_b_n/\Omega =$	7.84 ksi

UNIFORM COMPRESSION ELEMENTS

B.5.4.2 Flat Elements Supported on Both Edges - Web & Flange

Uniform compression strength, flat elements supported on both edges

Lower slenderness limit	$\lambda 1 =$	22.8	
Upper slenderness limit	$\lambda 2 =$	39.2	
Flange Slenderness	$b/tb =$	50.0	$\geq \lambda 2$
Web Slenderness	$h/th =$	50.0	$\geq \lambda 2$
$[k2c * \sqrt{(Bp * E)/(1.6 * b / t b)}]$	$F_c_n1 =$	15.98 ksi	
	$\Omega =$	1.65	
	$F_c_n1/\Omega =$	9.68 ksi	
$[k2c * \sqrt{(Bp * E)/(1.6 * h / t h)}]$	$F_c_n2 =$	15.98 ksi	
	$\Omega =$	1.65	
	$F_c_n2/\Omega =$	9.68 ksi	

FLEXURAL COMPRESSION ELEMENTS

B.5.5.1 Flat Elements Supported on Both Edges - Web

Flexural compression strength, flat elements supported on both edges

Lower slenderness limit	$\lambda_1 =$	34.73	
Upper slenderness limit	$\lambda_2 =$	92.95	
Slenderness	$h/th =$	50.00	$\lambda_1 - \lambda_2$
$[Bbr-m*Dbr*h/th]$	$Fb_n =$	33.71 ksi	
	$\Omega =$	1.65	
	$Fb_n/\Omega =$	20.43 ksi	

SHEAR

G.2 Shear Supported on Both Edges - Web

Members with flat elements supported on both edges

Lower slenderness limit	$\lambda_1 =$	38.73	
Upper slenderness limit	$\lambda_2 =$	75.65	
Slenderness	$h/th =$	50.00	$\lambda_1 - \lambda_2$
$[Bs-1.25Ds*h/th]$	$Fv_n =$	13.84 ksi	
	$\Omega =$	1.65	
	$Fv_n/\Omega =$	8.39 ksi	

ALLOWABLE STRESSES

Allowable bending stress	$Fb =$	6.54 ksi
Allowable axial stress, compression	$Fac =$	4.85 ksi
Allowable shear stress; webs	$Fv =$	8.39 ksi
Allowable axial stress, Tension	$Fat =$	15.15 ksi

Elastic buckling stress	$Fe =$	4.83 ksi
Weighted average allowable compressive stress (per Section E.3.1)	$Fao =$	9.68 ksi

MEMBER LOADING

Bending Moments

Bending moment developed in member	$Mz =$	1.7 kip-ft	
Bending stress developed in member	$fb =$	1.64 ksi	
Allowable bending stress of member	$Fb =$	6.54 ksi	< 1.0

Compression Loads

Compression load developed in member	$P =$	1,401 lb	
Compression stress developed in member	$fc =$	0.19 ksi	
Allowable compressive axial stress of member	$Fac =$	4.85 ksi	< 1.0

Tension Loads

Tension load developed in member	$T =$	140 lb	
Tension stress developed in member	$ft =$	0.01 ksi	
Allowable Tension axial stress of member	$Fat =$	15.15 ksi	< 1.0

Shear Loads

Shear load developed in member	$Vz =$	320 lb	
Shear stress developed in member	$fv =$	0.20 ksi	
Allowable shear stress of member webs	$Fv =$	8.39 ksi	< 1.0

Interaction Equations

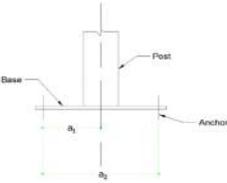
$\sqrt{[(fb/Fb)^2 + (fv/Fv)^2]} =$	0.25	< 1.0
$fa/Fa + fb/Fb =$	0.29	< 1.0
$fa/Fa + (fb/Fb)^2 + (fv/Fv)^2 =$	0.29	< 1.0

CONFIGURATION AND MOMENT TABULATION TOOLS

Member Loads:

$Mx =$ 20.44 kip-in	Applied Moment Per Member	30 PSF	Total Gravity Load
$My =$ 1.36 kip-in	Applied moment Per Member	6.3 FT	Post Trib Area in X-Axis
$Tn =$ 0.09 kip-in	Applied Torsion Per Member	7.4 FT	Post Trib Area in Y-Axis
$Vx =$ 216 lbs	Applied Shear Load Per Member	3 PSF	Uplift
$Vy =$ 236 lbs	Applied Shear Load Per Member	17 PSF	Lateral Load
$V =$ 320 lbs	Applied Resultant Shear Load Per Member		
$P =$ 1,401 lbs	Applied axial compression load		
$T =$ 140 lbs	Applied axial tension load	1.14 kip-in	Seismic Moment

Work Prepared For: Timan Window Treatments, Inc
Project: 24577 - Raymond Tatko
Mark/Detail: Post anchors to concrete



Design Loads (LRFD)	
M	32710 lb-in
T	224 lb
C	2242 lb
V	512 lb

Work Prepared For: Hudson
 Project: 24577 - Raymond Tatko
 Detail/Member: Base Plate Design

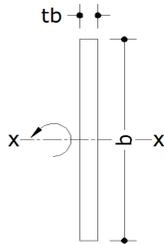
ALUMINUM DESIGN MANUAL (2020 EDITION)
Specifications for Aluminum Structures (Buildings)
Allowable Stress Design

Design Check of 10.625"x0.5" 6063-T6 Aluminum Flat Plate

Per 2020 Aluminum Design Manual

Alloy: **6063** Temper: **T6** Critically Welded: **N**

MEMBER PROPERTIES



Flat Plate Height	<i>b</i> =	10.625"
Flat Plate Thickness	<i>tb</i> =	0.500"
Moment of inertia about axis parallel to flange	<i>I_x</i> =	49.98 in ⁴
Moment of inertia about axis parallel to web	<i>I_y</i> =	0.11 in ⁴
Section modulus about the x-axis	<i>S_x</i> =	9.41 in ³
Radius of gyration about centroidal axis parallel to flange	<i>r_x</i> =	3.07 in
Radius of gyration about centroidal axis parallel to web	<i>r_y</i> =	0.14 in
Torsion constant	<i>J</i> =	0.44 in ⁴
Cross sectional area of member	<i>A</i> =	5.31 in ²
Plastic section modulus	<i>Z</i> =	14.11 in ³
Warping constant	<i>C_w</i> =	0.00 in ⁶

MEMBER SPANS

Unsupported member length (between supports)	<i>L</i> =	0.72 ft
Unbraced length for bending (between bracing against side-sway)	<i>L_b</i> =	0.72 ft
Effective length factor	<i>k</i> =	1.0

MATERIAL PROPERTIES

Tensile ultimate strength	<i>F_{tu}</i> =	30 ksi
Tensile yield strength	<i>F_{ty}</i> =	25 ksi
Compressive yield strength	<i>F_{cy}</i> =	25 ksi
Shear ultimate strength	<i>F_{su}</i> =	18 ksi
Shear yield strength	<i>F_{sy}</i> =	15 ksi
Compressive modulus of elasticity	<i>E</i> =	10,100 ksi

BUCKLING CONSTANTS

Compression in columns & beam flanges (Intercept)	<i>B_c</i> =	27.64 ksi
Compression in columns & beam flanges (Slope)	<i>D_c</i> =	0.14 ksi
Compression in columns & beam flanges (Intersection)	<i>C_c</i> =	78.38 ksi
Compression in flat plates (Intercept)	<i>B_p</i> =	31.39 ksi
Compression in flat plates (Slope)	<i>D_p</i> =	0.17 ksi
Compression in flat plates (Intersection)	<i>C_p</i> =	73.55 ksi
Compressive bending stress in solid rectangular bars (Intercept)	<i>B_{br}</i> =	46.12 ksi
Compressive bending stress in solid rectangular bars (Slope)	<i>D_{br}</i> =	0.38 ksi
Compressive bending stress in solid rectangular bars (Intersection)	<i>C_{br}</i> =	80.56 ksi
Shear stress in flat plates (Intercept)	<i>B_s</i> =	18.98 ksi
Shear stress in flat plates (Slope)	<i>D_s</i> =	0.08 ksi
Shear stress in flat plates (Intersection)	<i>C_s</i> =	94.57 ksi
Ultimate strength coefficient of flat plates in compression (slenderness limit λ ₂)	<i>k_{1c}</i> =	0.35
Ultimate strength coefficient of flat plates in compression (stress for slenderness > λ ₂)	<i>k_{2c}</i> =	2.27
Ultimate strength of flat plates in bending (slenderness limit λ ₂)	<i>k_{1b}</i> =	0.50
Ultimate strength of flat plates in bending (stress for slenderness > λ ₂)	<i>k_{2b}</i> =	2.04
Tension coefficient	<i>kt</i> =	1.0

D.2 Axial Tension

Tensile Yielding - Unwelded Members	<i>[F_{ty}]</i>	<i>F_{ty_n}</i> =	25.00 ksi
		<i>Ω</i> =	1.65
		<i>F_{ty_n/Ω}</i> =	15.15 ksi
Tensile Rupture - Unwelded Members	<i>[F_{tu}/kt]</i>	<i>F_{tu_n}</i> =	30.00 ksi
		<i>Ω</i> =	1.95
		<i>F_{tu_n/Ω}</i> =	15.38 ksi

AXIAL COMPRESSION MEMBERS			
E.2 Compression Member Buckling			
Axial, gross section subject to buckling	Lower slenderness limit	$\lambda_1 =$	18.23
	Upper slenderness limit	$\lambda_2 =$	78.38
	Slenderness	$\lambda(max) =$	59.76 < λ_2
	$[(Bc - Dc * \lambda)(0.85 + 0.15 * ((Cc - \lambda) / (Cc - \lambda_1)))]$	$F_{c_n} =$	17.03 ksi
		$\Omega =$	1.65
		$F_{c_n} / \Omega =$	10.32 ksi
FLEXURAL MEMBERS			
F.2 Yielding and Rupture			
Nominal flexural strength for yielding and rupture	Limit State of Yielding	$M_{np} =$	352.78 k-in
	$[Z * F_{cy}]$	$F_b =$	25.00 ksi
	$[M_{np} / Z]$	$\Omega =$	1.65
		$F_{b_n} / \Omega =$	15.15 ksi
	Limit State of Rupture	$M_{nu} =$	423.34 k-in
	$[Z * F_{tu} / kt]$	$F_b =$	30.00 ksi
	$[M_{nu} / Z]$	$\Omega =$	1.95
		$F_{b_n} / \Omega =$	15.38 ksi
F.4 Lateral-Torsional Buckling			
Rectangular bars subject to lateral-torsional buckling	Slenderness for shapes symmetric about the bending axis	$\lambda_{F.4.2.1} =$	62.30
	Slenderness for rectangular bars	$\lambda_{F.4.2.4} =$	44.04
	Slenderness for any shape	$\lambda_{F.4.2.5} =$	62.30
	Maximum slenderness	$\lambda(max) =$	62.30 < C_c
Nominal flexural strength - lateral-torsional buckling	$[M_{np}(1 - (\lambda / C_c)) + (\pi^2 * E * \lambda^2 * S_x / C_c^3)]$	$M_{nmb} =$	193.69 k-in
	$[M_{nmb} / S_x]$	$F_{b_n} =$	20.59 ksi
		$\Omega =$	1.65
		$F_{b_n} / \Omega =$	12.48 ksi
G.2 Shear Supported on Both Edges			
Members with flat elements supported on both edges	Lower slenderness limit	$\lambda_1 =$	38.73
	Upper slenderness limit	$\lambda_2 =$	75.65
	Slenderness	$b / t_b =$	21.25 $\leq \lambda_1$
	$[F_{sy}]$	$F_{v_n} =$	15.00 ksi
		$\Omega =$	1.65
		$F_{v_n} / \Omega =$	9.09 ksi
ALLOWABLE STRESSES			
	Allowable bending stress	$F_b =$	12.48 ksi
	Allowable axial stress, compression	$F_{ac} =$	10.32 ksi
	Allowable shear stress	$F_v =$	9.09 ksi
	Elastic buckling stress	$F_e =$	14.32 ksi
Weighted average allowable compressive stress (per Section E.3.1)		$F_{ao} =$	10.32 ksi

MEMBER LOADING			
<u>Bending Moments</u>			
Bending moment developed in member	$M_z =$	2.9 kip-ft	
Bending stress developed in member	$f_b =$	3.70 ksi	
Allowable bending stress of member	$F_b =$	12.48 ksi	< 1.0
<u>Axial Loads</u>			
Axial load developed in member	$F_x =$	320 lb	
Axial stress developed in member	$f_a =$	0.06 ksi	
Allowable compressive axial stress of member	$F_{ac} =$	10.32 ksi	< 1.0
<u>Shear Loads</u>			
Shear load developed in member	$V_z =$	1,401 lb	
Shear stress developed in member	$f_v =$	0.26 ksi	
Allowable shear stress of member	$F_v =$	9.09 ksi	< 1.0
<u>Interaction Equations</u>			
		$\sqrt{[(f_b/F_b)^2 + (f_v/F_v)^2]} =$	0.30 < 1.0
Eq H.1-1		$f_a/F_a + f_b/F_b =$	0.30 < 1.0
Eq H.3-2		$f_a/F_a + (f_b/F_b)^2 + (f_v/F_v)^2 =$	0.09 < 1.0

1. Project Information

Company:
Project Engineer: - -
Address: Jupiter, FL 33458
Phone: M: - P: -
Email: amccann@am-structures.com
Project Name: Raymond Tatko
Project Address: 1751 E Hines Hill Rd, Hudson, OH 44236
Notes:

2. Selected Anchor Information

Selected Anchor : Screw-Bolt+
Brand: DEWALT
Material: 5/8" Ø Low Carbon Steel
Embedment: h_{ef} 3.73 in h_{nom} 5 in
Approval: ICC-ES ESR-3889
Issued/Revision: Nov,2023 -
Drill method: Hammer Drilled



3. Design Principles

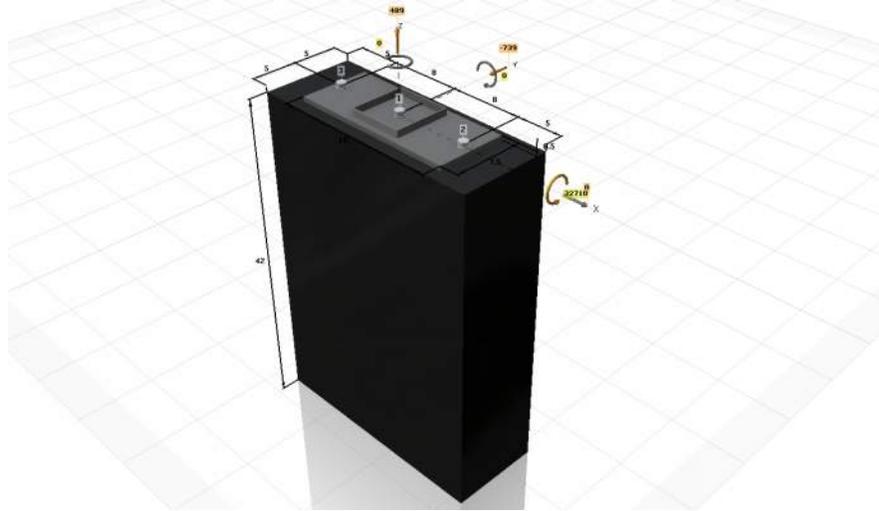
Design Method: ACI 318-19
Load Combinations: IBC 2018 User Defined Loads
Seismic Loading: Tension None Shear None $\Omega_0 = 2.5$

4. Base Material Information

Concrete:
 Type: Cracked Normal Weight Concrete
 Strength 3000 psi
Reinforcement:
 Edge Reinforcement None or < #4 Rebar
 Spacing Tension No (Condition B) Shear No (Condition B)
 Controls Breakout Tension False Shear False
Base Plate:
 Sizing Thickness 0.5 in Length 7.5 in Width 18 in
 Standoff None Height 0 in
 Strength 36000 psi
Profile: HSS Rectangular 6 X 6 X 0.125

Input data and results must be checked for agreement with the existing conditions, the standards and guidelines and must be checked for plausibility

5.Geometric Conditions



h_{slab}	42	in	h_{min}	7.000	in
Edge Cx-	5	in	c_{min}	1.750	in
Edge Cx+	5	in	c_{ac}	10.100	in
Edge Cy-	5	in	s_{min}	2.750	in
Edge Cy+	5	in			

6.Summary Results

Tension Loading

Design Proof	Demand(lb)	Capacity(lb)	Utilization	Status	Critical
Steel Strength:	3429.77	17069.00	0.201	OK	
Concrete Breakout Strength:	10289.00	11829.00	0.870	OK	Controls
Pullout Strength:	3430.00	4913.00	0.698	OK	

Shear Loading

Design Proof	Demand(lb)	Capacity(lb)	Utilization	Status	Critical
Steel Strength	246.00	9351.00	0.026	OK	
Concrete Breakout Strength:	739.00	5290.00	0.140	OK	Controls
Pryout Strength	739.00	25478.00	0.029	OK	

Input data and results must be checked for agreement with the existing conditions, the standards and guidelines and must be checked for plausibility

7. Warnings and Remarks

ANCHOR DESIGN CRITERIA IS SATISFIED

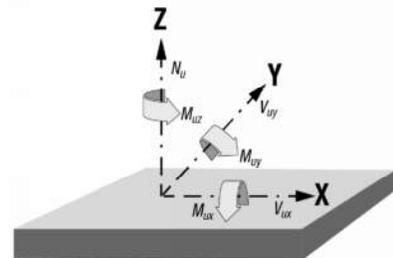


- The results of the calculations carried out by means of the DDA Software are based essentially on the data you put in. Therefore, you bear the sole responsibility for the absence of errors, the completeness and the relevance of the data to be put in by you. Moreover, you bear sole responsibility for having the results of the calculation checked and cleared by an design professional/engineer, particularly with regard to compliance with applicable standards, norms and permits, prior to using them for your specific project. The DDA Software serves only as an aid to interpret standards, norms and permits without any guarantee as to the absence of errors, the correctness and the relevance of the results or suitability for a specific application.

8. Load Condition

Design Loads / Actions

Nu	489	lb	Vux	0	lb	Vuy	-739	lb
Muz	0	in-lb	Mux	32710	in-lb	Muy	0	in-lb
Consider Load Reversal			X Direction	0%		Y Direction	0%	



Input data and results must be checked for agreement with the existing conditions, the standards and guidelines and must be checked for plausibility



9. Load Distribution

Max. concrete compressive strain:	0.202	%	<u>Anchor Eccentricity</u>					
Max. concrete compressive stress:	880.265	psi	ex	0	in	ey	0	in
Resulting tension force:	10289.297	lb	<u>Profile Eccentricity</u>					
Resulting compression force:	9800.299	lb	ex	0	in	ey	0	in

Resulting anchor forces / Load distribution

Anchor	Tension Load (lb)	Shear Load (lb)	Component Shear Load (lb)		Anchor Coordinates (in)	
			Shear X	Shear Y	X	Y
1	3429.77	246.3	0.0	-246.3	0.000	0.000
2	3429.77	246.3	0.0	-246.3	8.000	0.000
3	3429.77	246.3	0.0	-246.3	-8.000	0.000

Input data and results must be checked for agreement with the existing conditions, the standards and guidelines and must be checked for plausibility



10.Design Proof Tension Loading

Steel Strength

ACI 318-19 17.6.1

Variables

N_{sa} (lb)	ϕ
26260.000	0.65

Results

ϕN_{sa}	=	17069.0	lb
N_{ua}	=	3429.8	lb
Utilization	=	20.1%	



Table 17.5.2

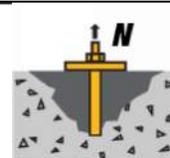
Concrete Breakout Strength

ACI 318-19 17.6.2

Equations

$$N_{cbg} = (A_{Nc} / A_{Nc0}) \cdot \Psi_{ec,N} \cdot \Psi_{ed,N} \cdot \Psi_{c,N} \cdot \Psi_{cp,N} \cdot N_b$$

$$N_b = k_c \cdot \lambda_a \cdot (f'_c)^{0.5} \cdot h_{ef}^{1.5}$$



Eqn. 17.6.2.1b

Eqn. 17.6.2.2.1

Variables

A_{Nc} (in ²)	A_{Nc0} (in ²)	$\Psi_{ec,N}$	$\Psi_{ed,N}$	$\Psi_{c,N}$	$\Psi_{cp,N}$
259.964	99.980	1.000	1.000	1.000	1.000
c_{ac} (in)	k_c	λ_a	f'_c (psi)	h'_{ef} (in)	C_{amin} (in)
10.100	21.000	1.000	3000	3.33	5
N_b (lb)	ϕ				
6998.950	0.65				

Results

ϕN_{cbg}	=	11829	lb
N_{ua}	=	10289.0	lb
Utilization	=	87.0%	

Table 17.5.2

Input data and results must be checked for agreement with the existing conditions, the standards and guidelines and must be checked for plausibility



Pullout Strength:

ACI 318-19 17.6.3

Equations

$$N_{pn} = \Psi_{c,P} \cdot N_p \cdot (f'_c / 2500)^n$$



Eqn. 17.6.3.1

Variables

$\Psi_{c,P}$	$N_{p,eq}$ (lb)	f'_c (psi)	n	ϕ
1.000	7559.000	3000	0.500	0.65

Results

ϕN_{pn}	=	4913	lb
N_{ua}	=	3430	lb
Utilization	=	69.8%	

Table 17.5.2

Input data and results must be checked for agreement with the existing conditions, the standards and guidelines and must be checked for plausibility



11.Design Proof Shear Loading

Reference

Steel Strength:

ACI 318-19 17.7.1

Variables

V_{sa} (lb)	ϕ
15585.00	0.60

Results

ϕV_{sa}	=	9351	lb
V_{ua}	=	246	lb
Utilization	=	2.6%	

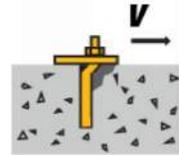


Table 17.5.2

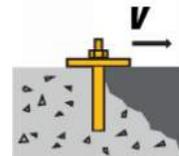
Concrete Breakout Strength

ACI 318-19 17.7.2

Equations

$$V_{cbg} = (A_{Vc} / A_{Vc0}) \cdot \Psi_{ec,V} \cdot \Psi_{ed,V} \cdot \Psi_{c,V} \cdot \Psi_{h,V} \cdot V_b$$

$$V_b = (7 \cdot (l_e / d_a)^{0.2} \cdot (d_a)^{0.5}) \cdot \lambda_a \cdot (f'_c)^{0.5} \cdot (c_{al})^{1.5}$$



Eqn. 17.7.2.1b

Eqn. 17.7.2.2.1a

Variables

A_{Vc} (in ²)	A_{Vc0} (in ²)	$\Psi_{ec,V}$	$\Psi_{ed,V}$	$\Psi_{c,V}$	$\Psi_{h,V}$
195.000	112.500	1.000	0.900	1.000	1.000
$\Psi_{\alpha,V}$	l_e (in)	d_a (in)	λ_a	f'_c (psi)	c_{al} (in)
1.000	3.730	0.625	1.000	3000	5.000
V_b (lb)	ϕ			V_{cbg} (lb)	
4844.171	0.70			7556.907	

Results

ϕV_{cbg}	=	5290	lb
Direction	=	Y-	
V_{ua}	=	739	lb
Utilization	=	14.0%	

Table 17.5.2

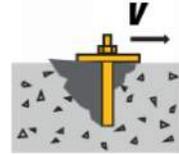
Shear Breakout Case 2 Controls

Input data and results must be checked for agreement with the existing conditions, the standards and guidelines and must be checked for plausibility



Pryout Strength

ACI 318-19 17.7.3



Equations

$$V_{cpg} = k_{cp} \cdot N_{cpg}$$

Eqn. 17.7.3.1b

$$N_{cbg} = (A_{Nc} / A_{Nc0}) \cdot \Psi_{ec,N} \cdot \Psi_{ed,N} \cdot \Psi_{c,N} \cdot \Psi_{cp,N} \cdot N_b$$

Eqn. 17.6.2.1b

$$N_b = k_c \cdot \lambda_a \cdot (f'_c)^{0.5} \cdot h_{ef}^{1.5}$$

Eqn. 17.6.2.2.1

Variables

A_{Nc} (in ²)	A_{Nc0} (in ²)	$\Psi_{ec,N}$	$\Psi_{ed,N}$	$\Psi_{c,N}$	$\Psi_{cp,N}$
259.964	99.980	1.000	1.000	1.000	1.000
c_{ac} (in)	k_c	λ_a	h'_{ef} (in)	f'_c (psi)	C_{amin} (in)
10.100	21.000	1.000	3.33	3000.000	5.000
N_b (lb)	k_{cp}	N_{cpg} (lb)	ϕ		
6998.950	2.000	18198.390	0.70		

Results

ϕV_{cpg}	=	25478.00 lb
V_{ua}	=	739 lb
Utilization	=	2.9%

Table 17.5.2

12. Interaction of Tension and Shear Loads

Reference

ACI 318-19 17.8**Equations**

$$\left(\left(\frac{N_{ua}}{\phi \cdot N_n} \right) + \left(\frac{V_{ua}}{\phi \cdot V_n} \right) \right) / 1.2 \leq 1.0 \quad \text{Eqn. 17.8.3}$$

Variables

$\frac{N_{ua}}{\phi \cdot N_n}$	$\frac{V_{ua}}{\phi \cdot V_n}$
0.870	0.140

Results

0.870	≤	1.0
Status	:	OK

ANCHOR DESIGN CRITERIA IS SATISFIED



Input data and results must be checked for agreement with the existing conditions, the standards and guidelines and must be checked for plausibility